

ABSTRACT

JENSEN, CHRISTINA LOUISE. Effect of atmospheric pressure plasma treatment on interfacial bonding strength of Ultrahigh Modulus Polyethylene fibers to epoxy resin (Under the direction of Yiping Qiu.)

The surface modification of UHMPE fibers by atmospheric pressure plasma treatments was examined. In one study the aging effects of atmospheric plasma treatments were studied. UHMPE fibers were treated for 0.5 and 1 min with He/O₂/air gas and for 2 and 4 min with He/air gas by atmospheric pressure plasma on a capacitively coupled device. The samples were tested for fiber/epoxy interfacial shear strength using the microbond technique at time intervals of 0, 3, 15 and 30 days after initial plasma treatment. Interfacial shear strengths (IFSS) for plasma treated fibers were 2 - 3 times as high as that of the control. The IFSS for the plasma treated fibers remained constant up to 15 days and then decreased afterwards. XPS Analysis and SEM photographs characterized the fiber surface modification. In a second study, delamination phenomena was studied by the transverse compression of seven-fiber bundle UHMPE microcomposites, a peeling test of laminated plain weave UHMPE fabric and tensile shear strength testing of a ten-layer plain weave UHMPE / Epoxy flat panel composite. Results showed a 49% increase in yield modulus of the plasma treated sample compared to the control in the transverse compression test. There was an 82.5% increase in bonding strength for the plasma treated sample during the peel test and a 25.7% increase in interlaminar shear strength of the ten-ply UHMPE composite proving that atmospheric plasma treatments are very effective in surface modification on a microscopic fiber level and a full-scale composite production level.

**EFFECT OF ATMOSPHERIC PRESSURE PLASMA TREATMENT ON
INTERFACIAL BONDING STRENGTH OF ULTRAHIGH MODULUS
POLYETHYLENE FIBERS TO EPOXY RESIN**

by

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A thesis submitted to the Graduate Faculty of
North Carolina State University
in partial fulfillment of the
requirements for the Degree of
Master of Science

TEXTILE ENGINEERING AND SCIENCE

Raleigh

2002

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DEDICATION

I would like to dedicate this thesis to my family. To my mother Diane, for her constant encouragement to “do anything you want to” and to my late father Lee, who was a strong role model, both academically and in life. To my sister Melinda, who keeps things in perspective and to my niece Miya whom, is just beginning down the long road of education. Without their support and encouragement and their willingness to allow me to move halfway across the world, I would not be where I am today.

BIOGRAPHY

Christina Jensen was born in Newcastle, Australia, on the 11th May 1979 to Lee and Diane Jensen. She has one sister, Melinda and a niece, Miya. She attended Newcastle Grammar School for much of her early education and then Abbotsleigh for High School in Sydney, NSW. She was involved in many school plays, the orchestra and woodwind quintet where she played the bassoon, and was on the social Cricket and Netball teams. It was also here that she was first exposed to Textile studies.

After winning the Year 12 prize for Textiles and Design, she decided to undertake a Bachelor of Science degree in Textile Engineering at Philadelphia University (formerly Philadelphia College of Textiles and Science) where she was awarded an academic scholarship. Her undergraduate studies at Philadelphia were coupled with numerous organization memberships, including Phi Psi professional textile fraternity, the Text newspaper, and Delta Phi Epsilon sorority, where she held multiple positions, including president. She was awarded the prize for Outstanding Undergraduate in Textile Engineering from 1997-99, was the recipient of the 1999-2000 DyStar scholarship and the 2000 SAMPE Undergraduate Engineering Scholarship and held the honor of Dean's List for all eight semesters. In 2001, Christina was the recipient of the AATCC Outstanding student in Textiles award, finished first in her major and was given the honor of being named in Who's Who in American Colleges and Universities. In 2001, she was invited to the SAMPE Student Symposium competition in Long Beach, CA; to present the paper her and another textile student had worked on for their senior research project. She was awarded first place for "Knitted Fiberglass, Kevlar and Spectra Composite Baseball Helmets". Christina also held

numerous jobs along the way, including teaching freshman textile classes, tutoring textile core courses, assisting in the Grundy testing lab, a summer internship at DyStar in Charlotte, and a six-month stint at the United States Department of Agriculture in Wyndmoor, PA.

In August 2001, she moved to Raleigh, NC to pursue a Masters of Science degree in Textile Engineering. For three semesters, she worked on surface modification of Ultrahigh Modulus Polyethylene fibers using atmospheric plasma for use in composites. As she prepares to graduate in December 2002, she will also be beginning a new career in the Advanced Materials Research division at Nike World Headquarters in Beaverton, OR.

ACKNOWLEDGEMENTS

I would like to thank Dr Yiping Qiu, my advisor, for his encouragement, direction and weekly meetings that constantly held me accountable and allowed me to aim for an early graduation. Thank you also to each of my committee members, Dr. Marian McCord and Dr Mohamed Bourham for their plasma know-how and experimental suggestions. A thank you to the members of my research group, to Chuyang Zhang, for explaining the transverse compression machine, for SEM photographs and for answering the many questions I had about compliance calculations. To Brice Langston, for patiently showing me how to manufacture composites, use the vacuum system and for assistance in shear tensile sample preparation and testing. To Dr. Zasheng Cai, Brian Bures and Yoon Hwang, for assisting me in atmospheric plasma treatments. Thanks to Dr Jan Pegram, for setting up the Sintech machine many times and to Brian Nablo at UNC for XPS data and an explanation of the analysis. And lastly, a very special thank you to Bill Hayes of Industrial Machine Solutions in Raleigh, and Hai Bui of the COT machine shop, for last minute composite cutting and machining.

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CHAPTER ONE

INTRODUCTION

1.1 Fiber Reinforced Composites

Composites are a material structure made up of two or more materials combined together so that on a macroscopic scale, the components can be distinguished, yet are firmly bonded together [1]. Fiber Reinforced Composites (FRC) consist of highly aligned fibers in a matrix. These fibers are usually high performance fibers such as aramid, carbon, fiberglass, boron or ultrahigh modulus polyethylene (UHMPE). Matrices usually consist of thermoset resins such as epoxy, pholic and polyimide or thermoplastic resins such as nylon, polyethylene and polysulfone [1].

When well designed, composites exhibit the best qualities of their constituents such as:

- Light weight
- High strength to weight ratio
- High stiffness to weight ratio
- Corrosion resistance
- Wear resistance
- Attractiveness
- Long fatigue life
- Thermal insulation
- Thermal conductivity
- Acoustical insulation [2]

1.2 Composite Applications

The industrial application of composites is vast and varied, from sporting goods to Formula One racing cars. The advantages listed above have showcased composites as an excellent alternative to traditional structural materials such as metals and concrete. Fiber Reinforced Composites have found their application in aerospace technologies, such as helicopter rotor blades and aircraft brakes, and in leisure activities such as golf clubs and bicycle frames [1]. Composites are used in the construction of yachts and ship hulls and can also be found as casings for gas turbine combustors and microelectronic devices [1].

Composites have also allowed engineers to dramatically cut down on the time taken to construct or assemble structures. For example, traditional bridge building consists of layers of steel bars and concrete, often taking two to three weeks to complete. New technologies, some studied at NCSU, have cut this time down to three days using lighter weight composite materials that can be manufactured off-site and fitted together at the scene eliminating the need for heavy cranes and large concrete pourers.

As the use of composites increases in construction and in specialized product design, the ability to predict the failure strength of that composite becomes important. Fiber Reinforced Composites in particular, require an understanding of the constituent fiber properties, and it is here that the Textile Engineer can play an important role in the design process.

1.3 Ultra High Modulus Polyethylene Fibers

According to Honeywell, UHMPE fiber is one of the world's strongest and lightest fibers [3]. It is, pound-for-pound, ten times stronger than steel, more durable than polyester and has a specific strength that is forty percent greater than aramid fiber [3].

UHMPE has a specific gravity less than 1g/cc and has excellent toughness and visco-elastic properties [3]. It can absorb energy and dampen vibrations, can withstand flexural fatigue and has a low dielectric constant (which makes it undetectable by radar) [3]. UHMPE is also resistant to chemicals, water and UV light [3]. While for most applications this is advantageous, in composite applications, relatively inert chemical structure of the fiber that renders the resistance to chemicals proves problematic.

Honeywell states that UHMPE fiber is used in numerous high-performance applications, such as police and military ballistic-resistant vests, helmets and armored vehicles, as well as sailcloth, fishing lines, marine cordage, lifting slings, and cut-resistant gloves and apparel [3].

1.4 Adhesion problems between fiber and matrix

The major problem associated with the failure of all FRC is insufficient adhesion between layers, and subsequent delamination [5]. UHMPE fibers are particularly susceptible to this failure mechanism simply due to their molecular structure. While some textile companies, such as 3Tex have decided to combat this issue through a 3D woven structure with vertical Z-yarns that hold the multiple layers of the composite together [5], others have looked to more traditional methods such as surface modification. Another

problem is that the wettability of the UHMPE fibers is also low, leading to poor resin infiltration into the preforms.

1.5 Modification of UHMPE fiber surface with plasma treatments

One solution to the interfacial bonding problem between UHMPE and matrix that has been studied extensively is surface modification by plasma treatments. While the traditional plasma process occurs under vacuum, this proves problematic for textile fabrics that are packaged on large beams, in hundreds of yards. The vacuum plasma process is a batch process, and has limitations on the size of the sample to be treated. By using an atmospheric plasma treatment device, the machine can operate in a pre-existing continuous processing line,. This can allow fabric to be thread through the device at a steady rate, treated and wound up again at the other end. An advantage of plasma treatments over other chemical finishing techniques for surface



Figure 1.1 Atmospheric Plasma Chamber at North Carolina State University

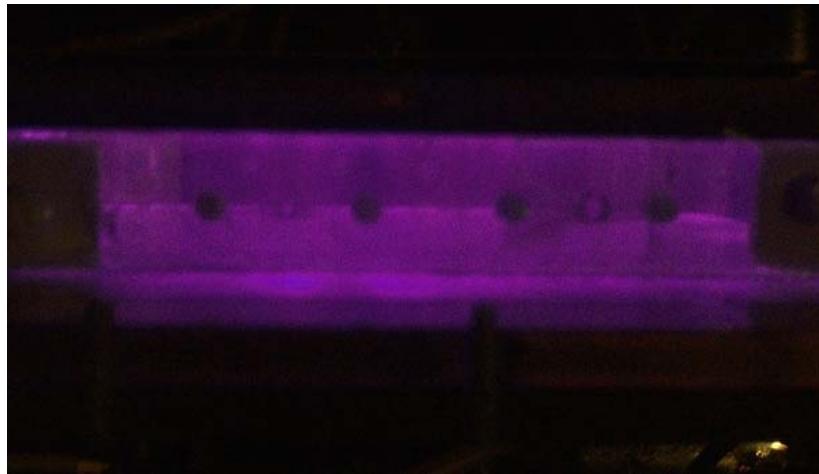


Figure 1.2 Plasma Glow – Helium / air.

modification is that there is virtually no waste generated. Plasma uses a series of inert gases that can be safely removed through an exhaust system.

1.6 Atmospheric pressure plasma treatment process

The atmospheric pressure plasma treatment system set up in the College of Textiles at North Carolina State University is a capacitively coupled chamber. It contains two horizontal parallel dielectric plates to which an electric charge is applied. A neutral gas such as Helium or Helium/Oxygen is pumped into the chamber where the electric field causes a breakdown of the gas into electrons and ions and the formation of charge carriers [6]. These charge carriers accelerated in the oscillating electric field couple their energy into the plasma via collisions with other particles [6].

There is very little research in how long atmospheric plasma treatment will last on a textile substrate. It has been suggested that traditional vacuum plasma treatments eventually “wear off” after a certain length of time, as chemical functional groups grafted

to the surface of the polymer begin to rearrange. While the etching that occurs on some substrates does not revert, there is some suggestion that prolonged environmental exposure time after treatment will cause the surface to attract debris, filling in some of these micro-pits and therefore becoming less effective in resin bonding. The main focus of Chapter Two is to examine the aging of the atmospheric plasma treatment on UHMPE fibers. A one month study into levels of interfacial shear strength on the fiber level will reveal the length of time the plasma treatment stays effective for improved composite manufacturing.

1.7 Interfacial bonding tests

To quantify the effectiveness of the atmospheric plasma treatments on UHMPE fibers and fabrics, a series of interfacial bonding tests have been performed throughout this thesis. These include A) Microbond test that studies the interfacial shear strength (IFSS) of an epoxy bead on the individual fiber surface (predominantly Mode II failure). B) Transverse compression of seven-fiber microcomposites that will study the interfacial bonding strength in a predominantly Mode I failure. C) Peel test, that will study the bonding strength of two UHMPE fabric layers bonded together with epoxy (predominantly Mode I failure), and D) Tensile shear test, that studied the force required to delaminate composite layers in a pure shear loading mechanism (predominantly Mode II failure). By studying the interfacial bonding strength of UHMPE to epoxy before and after the plasma treatments through several micromechanical tests and macro-composite level tests, we are able to determine the effectiveness of the plasma treatment on UHMPE fiber surface modification.

1.8 References

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CHAPTER TWO

THE AGING OF ATMOSPHERIC PLASMA TREATED ULTRAHIGH-MODULUS POLYETHYLENE FIBERS

2.1 INTRODUCTION

Ultra high modulus polyethylene (UHMPE) fibers have found increasing use in fiber-reinforced composites and especially in ballistic applications where high-energy absorption is required due to their high tensile strength and modulus [1] and its low specific density. The molecular structure of the polymer, however, allows low or no chemical bonding to other substances [2]. This is particularly detrimental in fiber-reinforced composites where good adhesion between the fiber and the matrix is critical. In order to enhance the adhesion between UHMPE fibers and matrix, both low pressure (vacuum) and atmospheric pressure plasmas have been effectively used [3-6]. There are two major contributors to the improved adhesion between the fiber and the matrices. Through surface chemical analysis, radicals have been shown to graft to the surface of the polymer allowing chemical bonding to occur and through Scanning Electron Microscopy (SEM), mechanical etchings have been observed that increase surface roughness allowing greater adhesion to resins [3-11]

However, it is well known that the effect of the plasma treatments cannot be permanent. In other words, there is an aging process of the plasma treated surface. X-ray Photoelectron Spectroscopy (XPS) studies have found that over the time there is diffusion and reorientation of polar functional groups, adsorption of contaminants, and reaction of

free radicals on the surface of plasma treated polymers [12-16]. Therefore there is a decrease in surface hydrophilicity that causes the polymer to lose some of its effective adhesion properties. In these studies, only low pressure plasma was used and little has been reported on aging of atmospheric pressure plasma treated fibers or polymers.

In our previous studies, UHMPE fibers treated by He/Air and He/O₂ atmospheric pressure plasmas showed a significant increase of interfacial shear strength (IFSS) with epoxy. It was found that the surface chemical composition of the Spectra® fibers was altered by the He/Air and He/O₂ plasma treatments significantly [7,8]. In addition, it was found that the 2 min He/air plasma treatment tended to increase the tensile strength of the fiber [7], while He-O₂/air tended to decrease its tensile strength due to a more aggressive surface etching effect of the treatment [8].

This paper presents a comprehensive aging study of atmospheric plasma treated UHMPE fibers over a period of thirty days using the treatment conditions proved to be effective in our previous studies. Interfacial shear strength will be measured using the microbond technique. XPS and SEM will analyze surface chemical and morphological changes.

2.2 EXPERIMENTAL PROCEDURE

2.2.1 Plasma Treatment

UHMPE fibers (Ningbo Dacheng Chemical Fibre Group Co.) were wound onto glass frames to allow effective overall treatment and then rinsed in acetone to remove any finishes. The frames were then placed sequentially in the atmospheric plasma chamber for treatment. The plasma chamber is a capacitively coupled device developed at the

College of Textiles, North Carolina State University. It had a frequency set to 5 kHz with a 100-150 watts maximum power input. A constant rate of flow for each gas was used in the treatment. A 5 mm thick Plexiglas dielectric material was used and the plate separation distance was 5.5 cm. It has been found from our previous studies [8,17] that a He/air only plasma treatment becomes effective for increasing UHMPE/epoxy IFSS at treatment times of 2 minutes and greater while He/O₂/air plasma treatments become effective at 30 seconds and longer. Therefore, two sets of fibers were treated with He/air at 2 minute and 4-minute treatment times respectively. Two sets of fibers were treated in a mixture of He/O₂/air at 30 second and 1-minute treatment times respectively. After the treatment, the fibers were taken off the glass frames and wound onto smaller bobbins, then placed in a desiccator.

2.2.2 Interfacial bonding sample preparation and test

The application of resin on the single fibers was time-critical, so the first set of samples was prepared exactly one hour after plasma treatment. The second set was prepared exactly three days from the initial plasma treatment and the third and fourth sets of samples were prepared 15 days and 30 days respectively after the initial plasma treatment. The process involved stringing individual fibers (8 denier) onto a frame then placing beads of epoxy resin along the length of each hung fiber, as described in the article by Qiu et. al. [18]. The sequence of applying resin beads to the fiber samples was determined using a randomization process to eliminate any systematic error due to the viscosity change in the resin over time. The total working time to distribute resin beads

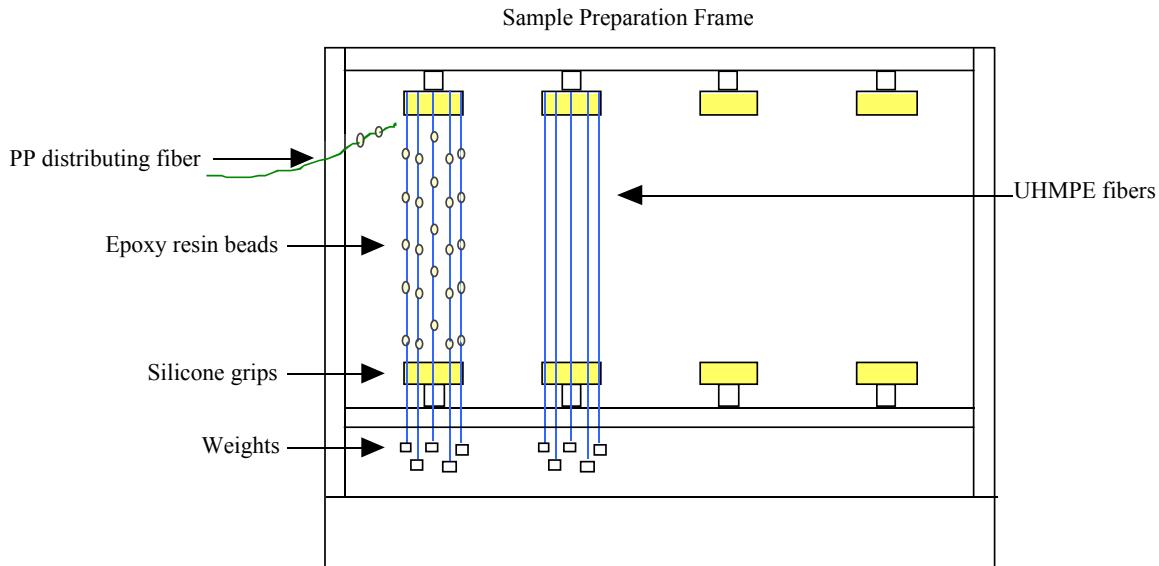


Figure 2.1 Microbond frame used to prepare epoxy resin beads on UHMPE fibers

was half an hour and exact placement and time was recorded. The epoxy resin used was 70% Dow Epoxy Resin (DER) 331 (polymer of epichlorohydrin and bisphenol-A), 30% DER 732 (polymer of epichlorohydrin and polyglycol) with 12 phr (parts per hundred) of Dow Epoxy Hardener (DEH) 26 (tetraethylenepentamine, triethylenetetramine and polyethylenepolyamine mix) from Dow Chemical. The fibers were then placed in the oven and cured for 3 hours at 80°C. The microbond test [18] was carried out at an upper clamp displacement rate of 1mm/min on a Sintech universal testing machine with a load cell of 25 N.

2.2.3 Microscopy examination

An Anti-Mould Nikon Labophot2-POL polarized light microscope equipped with a Sony CCD camera was used to take images of the beads. The length and width of the

beads was then determined using pixel measurements from Adobe Photoshop. A typical fiber/bead photograph is shown in Figure 2.2. Original data for the pixel measurements can be found in the Appendix.

2.2.4 Scanning electron microscopy

The Hitachi S-3200 Variable pressure SEM model was used to inspect the surface modification of the fibers directly after plasma treatment, then 3, 15 and 30 days after plasma treatment. The fibers were mounted on a sample platform and gold coated for three minutes in an Argon plasma sputtering device. SEM images were taken with a 5000x magnification.

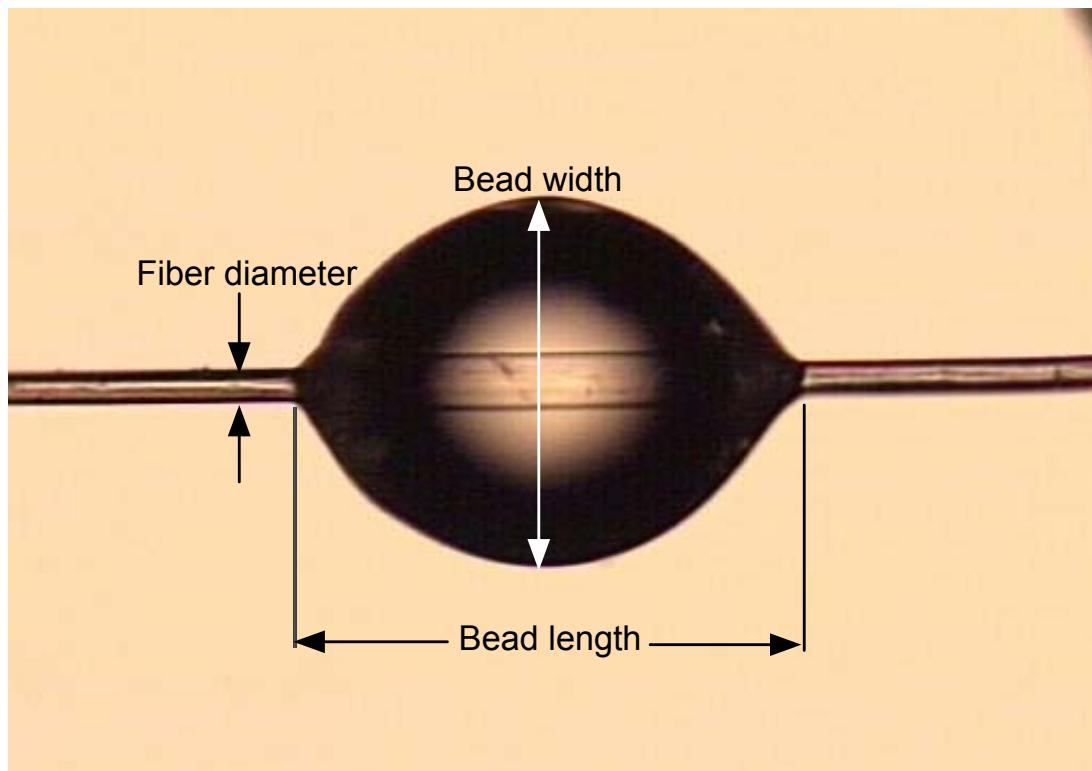


Figure 2.2 Typical fiber/bead photograph taken under the microscope for pixel measurements to calculate the fiber diameter and bead sizes. The values are used to plug into the Shear-lag model for calculation of Interfacial shear strength.

2.2.5 XPS Analysis

XPS analysis was performed on a Perkin Elmer PHI 5400 machine. The surface chemical composition of the UHMPE was obtained from measurement of the areas of the C 1s, N 1s and O 1s peaks, obtained at 35.75eV pass energy at a photoemission angle of 45°. Curve fitting was performed on the C 1s peaks, following subtraction of an “Integrated Shirley” baseline. A Gaussian band-type was used for the peak shape. Five component peaks were fitted to each sample at the following binding energies: C-C 285.0eV, C-OH 286.2eV, =C=O 288.1eV, -COOH 289.4eV and O=C-O 290.5eV [9,10,12]

2.2.6 Tensile Test

Single fiber tensile tests were performed at the beginning of aging study and after 30 days to determine any strength variability in the treated samples. Single fibers were prepared following ASTM standard D-3822 (Standard Test Method for Tensile Properties of Single Textile Fibers) at a gage length of 100 mm. The Sintech universal testing machine with a load cell of 25 N was again utilized for this test. The displacement rate was 50 mm/min. Data was collected from at least 20 fiber samples for each treatment.

2.2.7 Statistical Analysis

One way analysis of variance (ANOVA) and Fisher’s pair-wise multiple comparisons were used to compare the tensile strengths of the fibers in different treatment groups. They were also used to evaluate the difference in IFSS among the treatment groups. A P-value less than 0.05 was considered significant.

2.3 RESULTS AND DISCUSSION

2.3.1 Microscopy examination

The surfaces of the UHMPE fibers with each plasma treatment are shown in Fig 1. The He/O₂/ air plasma treatment at 30 seconds and 1 minute appeared to etch the fiber surface where micro cracks were developed. Similar etching effect is evident again in the He/air plasma treatments at 2 minutes and 4 minutes respectively. The surface of the Control sample (no plasma treatment) remains free of etching or micro cracks at each time interval during the aging study (0, 3, 15, 30 days). The He/O₂/ air plasma treated fibers show similar levels of etching at each time interval with the 30 second treated samples slightly less etched than the 1-minute treatment. Similarly, the 2-minute He/air plasma treatment was etched and micro-cracked less severely than the 4-minute He/air plasma treated samples. The level of etching for each time interval during the aging process however remains fairly constant. Some surface debris appeared on SEM photographs taken at the 30-day interval. This may be due to a decrease in surface hydrophilicity, which leads to static charge build up and hence attracts more dust particles.

2.3.2 Micro-bond Test

The IFSS, τ_i , was calculated using the following equation [19], derived from the well-known shear-lag model:

$$\tau_i = \frac{n P_{\max} \coth(nL/r)}{2A} \quad (1)$$

where P_{\max} is the peak load, A is the cross-sectional area of the fiber, L is the imbedded length, r is the equivalent fiber radius calculated from the fiber cross-sectional area and n is defined as:

$$n = \left[\frac{E_m}{E_f(1 + \nu_m) \ln(R/r)} \right]^{1/2} \quad (2)$$

Where E_m (=1.4 GPa) is Young's modulus of the matrix reported in Ref. [18], ν_m (=0.4) is Poisson's ratio of the matrix measured in our laboratory, E_f (=90 GPa) is the tensile modulus of the fiber taken from the Ningbo Dacheng Chemical Fibre Group Co. product specification sheet, and R is the radius of the epoxy beads.

As shown in Fig 2, while the control samples remain constant over the 30-day time interval, the 2 and 4 min He/air plasma treated samples showed a steady but significantly higher value than that of the control group for the first fifteen days, then decreased significantly after 30 days though it is still significantly higher than control. Similar observations can be made in for the 30 second and 1 minute He/O₂/air plasma treated fibers as shown in Fig 3 where the IFSS remained constant up to the 15th day and significantly decreased after 30 days.

2.3.3 Single-fiber tensile strength

Table 2.1 shows the results of single fiber tensile testing of UHMPE fibers for different treatment groups. There appeared to be no statistically significant difference between tensile strengths of the control and the plasma treated fibers. There was however statistically significant difference between the tensile strength of the 1 min He/O₂/air plasma treatment and the 2 min He/air plasma treatment due to slightly improved fiber tensile strength after the 2 min He/air plasma treatment.

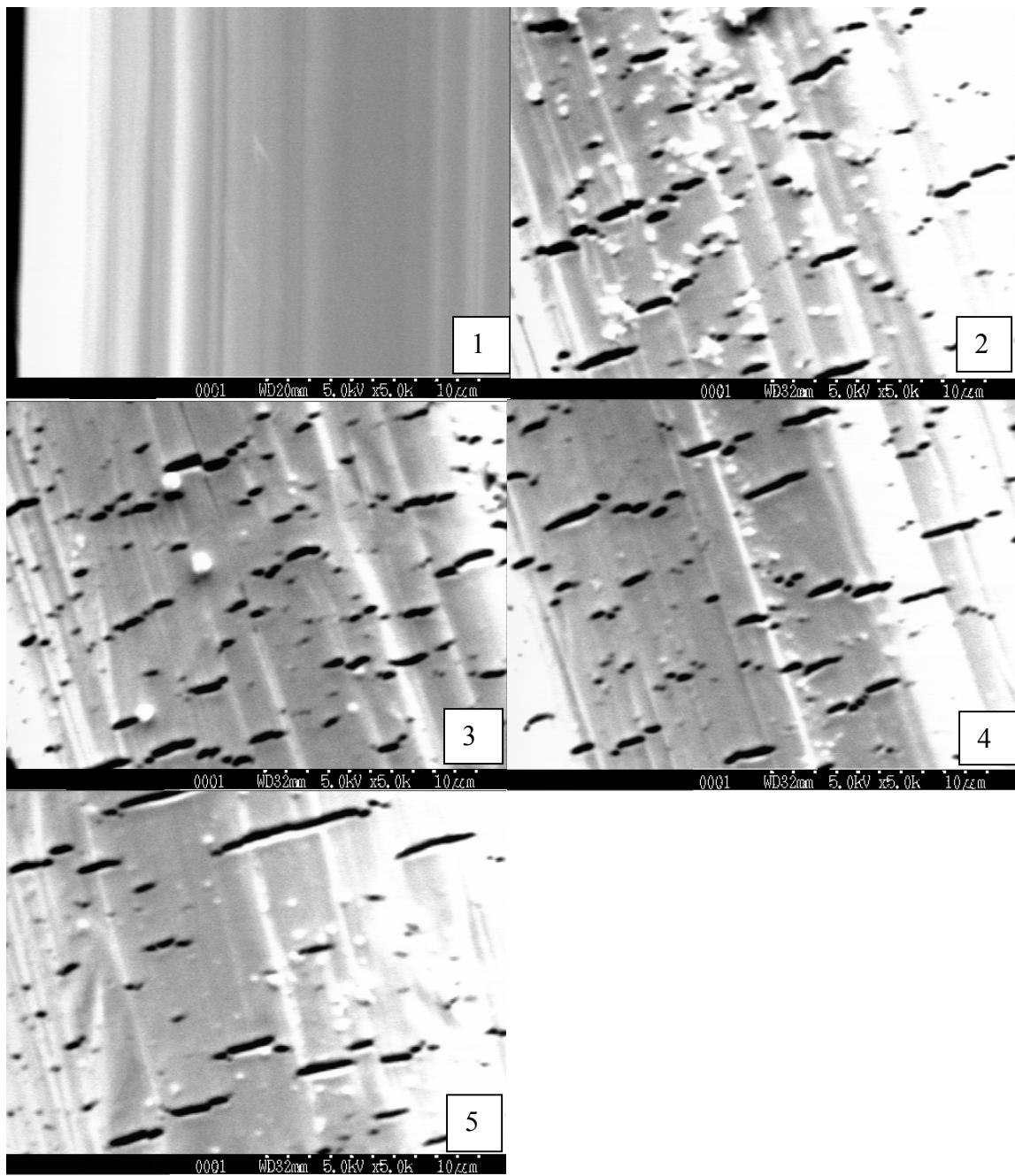


Figure 2.3 SEM Photographs of micro cracks on fiber surface, taken at 30-day time interval, 5000x Magnification: (1) control, (2) 0.5 min He/O₂/air plasma treatment, (3) 1 min He/O₂/air plasma treatment, (4) 2 min He/air plasma treatment, (5) 4 min He/air plasma treatment.

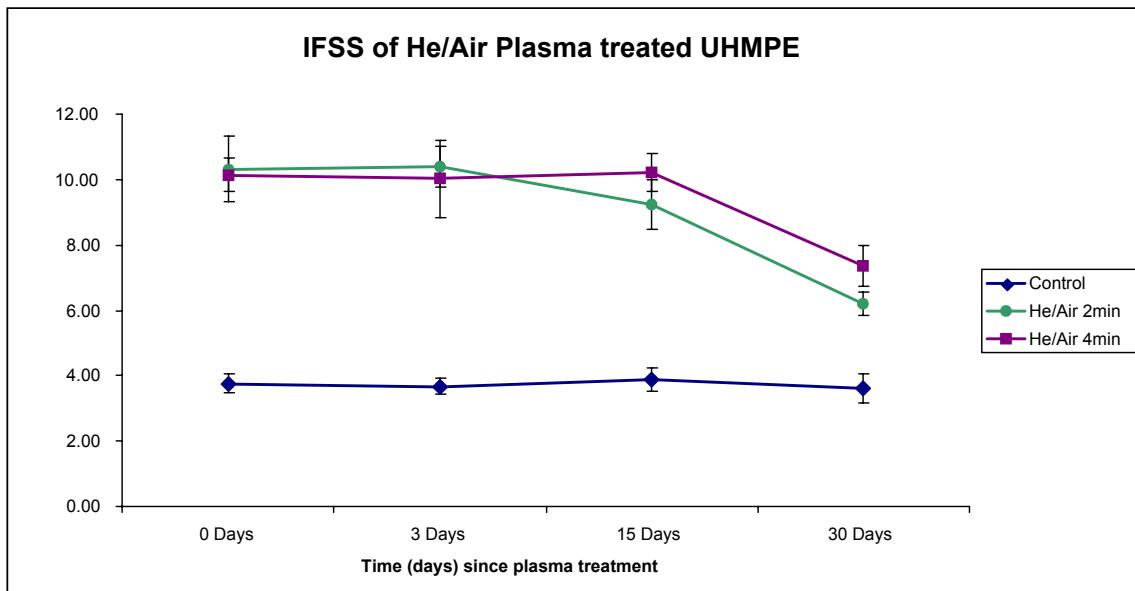


Figure 2.4 Change in interfacial shear strength for He/air plasma treated UHMPE fibers, over a time period of 30 days, using the Microbond technique.

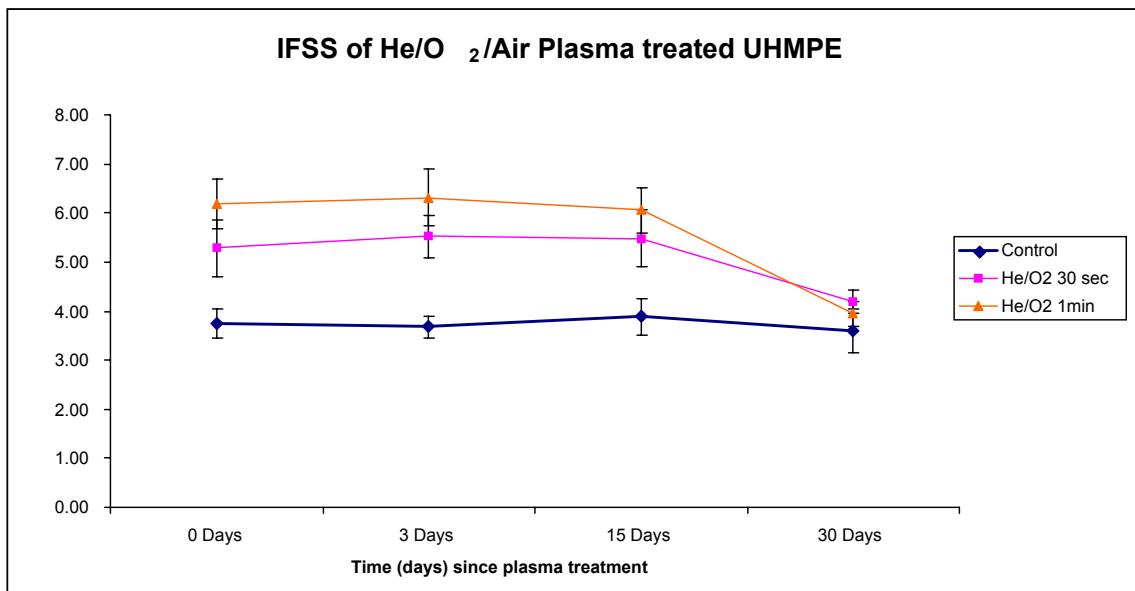


Figure 2.5 Change in interfacial shear strength for He/O₂/air plasma treated UHMPE fibers, over a time period of 30 days, using the microbond technique.

Table 2.1: Tensile Strength

Treatment	Number of Specimens	Tensile Strength (GPa)	Standard Deviation (GPa)
Control	22	2.80 ^{ab*}	0.29
0.5 min He/O ₂ /air	22	2.76 ^{ab}	0.21
1 min He/O ₂ /air	23	2.88 ^a	0.12
2 min He/air	20	2.64 ^b	0.37
4 min He/air	23	2.78 ^{ab}	0.18

*Means with different letters are statistically different at P < 0.05

2.3.4 XPS Analysis

From the XPS data analysis, it can be observed (Table 2.2) that immediately following plasma treatment there is a slight increase in both oxygen content and nitrogen content, except for the 4 minute He/air treatment. The control sample also showed significant oxygen and nitrogen contents, which may likely come from contamination. This can be confirmed by the fact that the 4-minute He/air treatment shows a slight decrease in both contents. For all plasma treated samples, there is also a slight increase in the percentage of C-OH bonds. For the He/O₂/air plasma treated samples, there is an increase in C=O bonds and COOH bonds while O=C-O bonds remain the same as the control. For the He/air plasma treated samples, there is a slight decrease in C=O bonds, COOH bonds and an increase in O=C-O bonds.

After 30 days from initial plasma treatment, it can be observed (Table 2.3) that there is a decrease in the atomic concentration of oxygen for all treated samples, and a decrease in nitrogen for the He/O₂/air plasma treated samples. The He/air treated samples show a slight increase in nitrogen content. The carbon bonds (C-C) observed after 30 days show a similar trend compared to the 0 day samples overall, with an increase seen only in the He/air plasma treated samples. Additionally, for the He/air treated samples, there is a decrease in C-OH, COOH and O=C-O percentages from the 0 day analysis, and for the He/O₂/air treated samples, there is a decrease in C-OH and C=O bonds. Therefore it can be concluded that the decrease of interfacial bonding strength is likely due to the change of the surface chemistry of the fibers. The decrease in surface oxygen content for each of the plasma treated samples, suggests that this may be the critical factor in fiber surface wet-ability. Therefore the decrease in Interfacial Shear Strength after 15 days is directly related to the rearrangement of polar functional groups on the surface and decrease in oxygen. The levels of oxygen present on the fiber surface may have been at a maximum immediately after treatment (contact angle = 0), and at the fifteen day interval, the critical level of oxygen may have dropped to below the optimum causing an increase in contact angle and decrease in surface wet-ability. This theory however, needs to be researched further using contact angle analysis on the individual fiber surface.

The increase in IFSS immediately after plasma treatment is a result of both the plasma gas induced surface chemical composition change and increased surface roughness may not be as effective in increasing the bonding strength because of poor wettability of the fiber to epoxy.

Table 2.2: XPS analysis of fiber immediately after plasma treatment

Treatment	Atomic Concentration							
	C	N	O	C-C	C-OH	C=O	COOH	O=C-O
Control	91.2	2.0	6.8	79.7	14.3	4.0	1.8	0.1
0.5 min He/O ₂	88.0	2.4	9.7	77.2	16.2	4.7	1.8	0.1
1min He/O ₂	87.5	2.9	9.6	77.7	15.1	4.1	2.9	0.1
2min He/air	88.8	2.1	9.1	76.5	18.1	3.2	1.8	0.4
4min He/air	93.2	1.5	5.3	81.0	15.4	2.7	0.5	0.4

Table 2.3: XPS analysis of fiber at 30 days after plasma treatment

Treatment	Atomic Concentration							
	C	N	O	C-C	C-OH	C=O	COOH	O=C-O
Control	92.8	2.0	5.2	79.2	15.8	2.9	1.9	0.2
0.5 min He/O ₂	92.8	1.6	5.6	78.6	15.3	3.3	2.5	0.2
1min He/O ₂	92.0	2.5	5.5	79.5	14.8	3.1	2.5	0.1
2min He/air	91.9	3.2	4.8	84.3	10.7	4.5	0.3	0.2
4min He/air	93.4	2.6	3.9	85.0	11.4	3.1	0.4	0.1

2.4 CONCLUSIONS

The aging effects of an atmospheric plasma treatment on UHMPE fibers were studied over a period of 30 days. The following conclusions can be made:

- (1) The initial increase of IFSS between both sets of plasma treated UHMPE fibers to the control remains at a constant level until at least 15 days after the treatment. After 30 days a significant decrease in IFSS for all the plasma treated groups was observed. The He/air treated fibers still have significantly higher IFSS than that of the control.
- (2) There is no observable difference in surface morphology of the fibers for each treatment group at different time intervals. Increased dust debris, however, was observed after 30 days of aging.
- (3) Compared with the control, there was no statistical significance in the tensile strength of fibers after each corresponding plasma treatment.
- (4) 30 days after the treatment, XPS analysis showed a decrease in oxygen content and in corresponding functional groups for all plasma treated samples. Therefore the decrease of interfacial bonding strength is likely due to the change of the surface chemistry of the fibers and in particular the decrease of oxygen content.

2.5 FUTURE WORK

Now that an understanding of the aging process of atmospheric plasma treatments has been reached, the next step in this research project is to effectively limit or inhibit the aging process. It has been proven that atmospheric plasma stays effective at room temperature for approximately 15 days. Prevention of the loss of surface chemical

functional groups and in particular oxygen content is the next logical step. One suggestion is to study the effect of temperature on the aging process. By keeping the atmospheric plasma treated samples in a freezer (below 0°C) or in liquid nitrogen, this could prevent the surface chemical groups from rearranging. It has been found in one study [20] that the aging process of vacuum plasma treatments on High Density Polyethylene (HDPE) can be reduced by an increase of Argon percentage (in an Argon/Oxygen gas mixture), or by increasing the crystallinity of the fiber itself. This is another interesting topic that may be studied with respect to atmospheric plasma.

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CHAPTER THREE

TRANSVERSE COMPRESSION AND DELAMINATION OF ATMOSPHERIC PLASMA TREATED UHMPE COMPOSITES

3.1 INTRODUCTION

Ultra high modulus polyethylene (UHMPE) fibers have found increasing use in fiber-reinforced composites, especially in load-bearing situations. Newer applications, such as under-water cables and ballistic panels, require a high level of transverse compressive stiffness in order to resist failure [1-3]. Therefore it is essential for composite design purposes to fully understand the load-response and delamination process in the transverse direction, in order to determine lifetimes for those load-bearing structures [4].

The transverse response of unidirectional composites has been studied [5-10] and three main failure characteristics have emerged [4,6,9-12]. In composites with strong fibers and strong bonding between fiber and matrix, failure occurs within the bulk matrix close to the fiber, this is shown in Figure 3.1(a). Figure 3.1(b) shows the failure effect with a weak interface where debonding occurs under transverse compression. The third failure mechanism observed is when the transverse fiber strength is low. shows that in this case, the fiber splits near the aligned surface or across the core as shown in Figure 3.1(c). An assumption here is that currently, polyethylene / epoxy composites fail at the interface region. Atmospheric plasma treatments improve interfacial bonding and

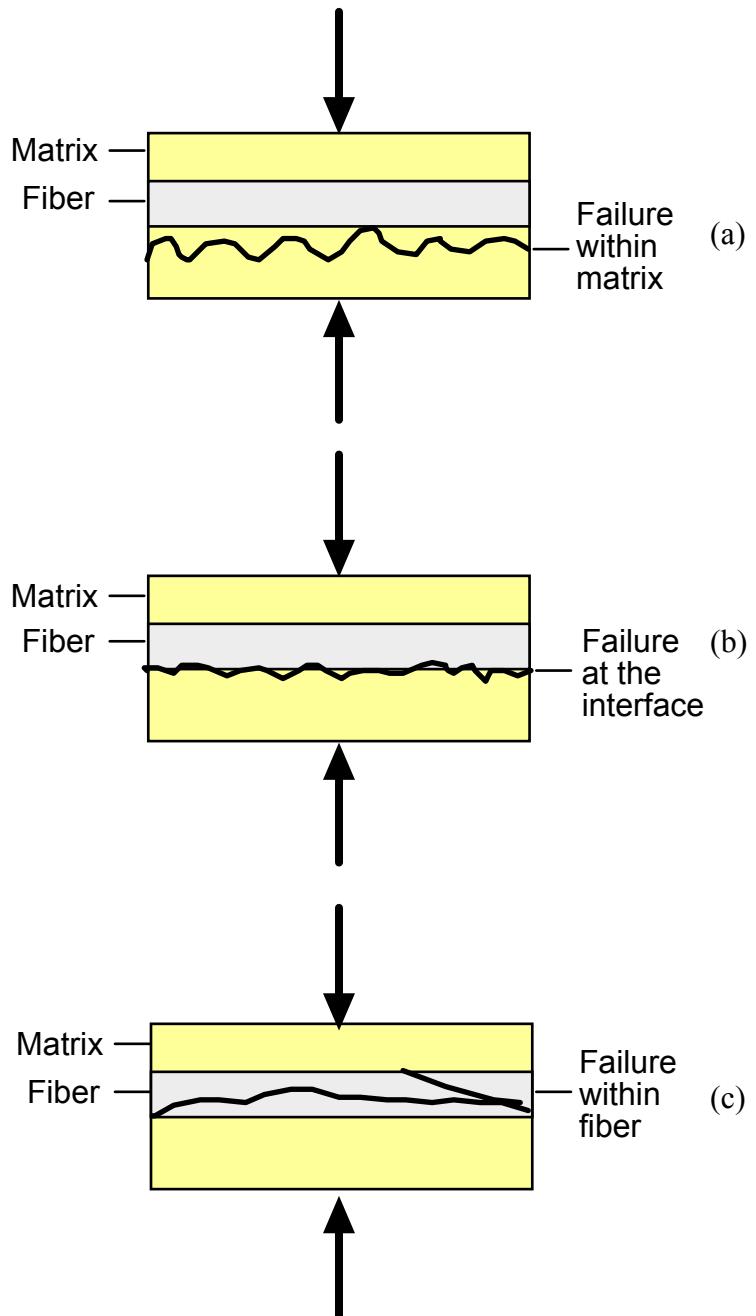


Figure 3.1 (a) Failure within the matrix, (b) Failure at the Interface,
(c) Failure within the fiber

therefore it should be proven (using Scanning Electron Microscopy) that the plasma treated UHMPE fibers should bond more readily with the epoxy, causing failure to occur instead within the bulk matrix.

Debonding of fiber/matrix interface is most likely to be a Mode I and Mode II mixed type of failure [13]. However, in a fiber pull out test, Mode II type of failure is considered dominant while a composite tensile tested along perpendicular to the fiber oriented direction gives Mode I types of failure. At micromechanical level, however, there has been little report on testing Mode I type of failure on fiber-matrix interface. Our research group has made an effort on a new transverse compression test of seven fiber microcomposite in order to characterize the Mode I type of failure that occurs often in compression loading of composites [14].

The purpose of this experiment is to study the debonding between a UHMPE fiber and epoxy resin in a composite, illustrating the weak interface mode of failure. It is well known that UHMPE fibers do not bond easily to resins, due to their inert molecular structure. By modifying the surface of the fiber and therefore strengthening the interface, it has been shown that compressive strength will increase significantly [15]. From our previous research, treatment with atmospheric plasma has proven to be effective in improving interfacial shear strength between UHMPE fibers and epoxy resin [16-18] [Chapter 2]. As a result of the plasma treatment, chemical functional groups were grafted to the fiber surface, and micro-etchings occur, allowing epoxy resins to bond more readily to the fiber [16-18] [Chapter 2]. To effectively study the delaminating effect, three test methods have been implemented. These include transverse compression of

seven-fiber microcomposites, a peeling test of laminated fabric layers, and pure shear test of ten-layer UHMPE fabric composites.

3.2 EXPERIMENTAL PROCEDURE

3.2.1 Plasma Treatment

For the seven-fiber bundle experiment, UHMPE fibers (Ningbo Dacheng Chemical Fibre Group Co.) were wound onto two glass frames to allow effective overall treatment and then rinsed in acetone to remove any finishes. One frame was then placed in the atmospheric plasma chamber for treatment. The second frame was left as the control. The plasma chamber is a capacitively-coupled device developed at the College of Textiles, North Carolina State University. It had a frequency set to 5 kHz with 100-150 watt maximum power input. A constant rate of flow for the gas was used in the treatment. A 5mm thick Plexiglas dielectric material was used and the plate separation distance was 5.5 cm. It has been found from our previous studies [16,18] that a He/air plasma treatment becomes effective for increasing UHMPE/epoxy IFSS at a treatment time of 2 min. Therefore, the fibers were treated with He/air for 2 min duration. After the treatment, the fibers were immediately made into the seven fiber microcomposites.

For the ten-layer composite experiment and the peeling test, 304.8 mm × 304.8 mm plain weave UHMPE fabric samples (Honeywell / Warwick Mills) were cut and placed individually into the atmospheric plasma chamber and treated with He/air for 2 minutes each with the same specifications described above.

3.2.2 Seven-fiber bundle composite sample preparation

Seven-fiber bundle microcomposites have been researched and proven to be an effective scaled study of the composite's response to tensile load and interface delamination [4,14,19-21]. Figure 3.2 shows the apparatus used to manufacture the microcomposites. First, lengths of single fibers were draped over the upper silicon grip and weighted with springs of equal mass. Six fibers hung around the perimeter of the grip at equal distances and one hung in the center. Next, a polypropylene fiber was used to place large droplets of resin on exactly three of the hung fibers, beginning from the top.

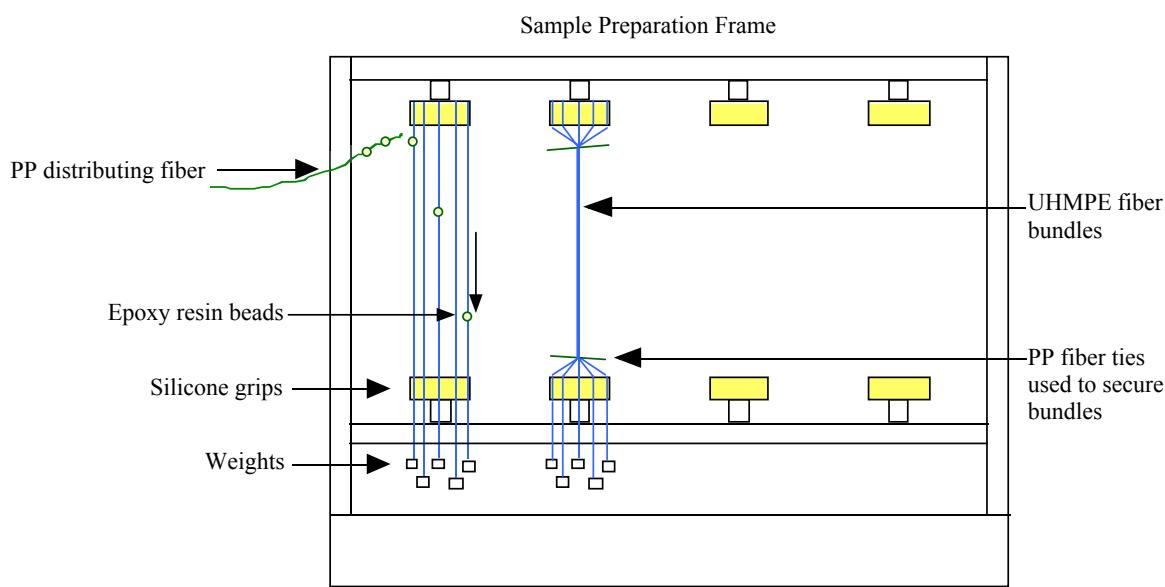


Figure 3.2 Seven-fiber bundle composite manufacturing. The fibers are shown hung from silicone grips with epoxy resin beads coating the fiber length. Polypropylene monofilaments are used to tie off the seven-fiber bundles at the top and bottom.

The beads were then allowed to run down the length of the fibers to coat the surface. The epoxy resin used was 70% DER 331, 30% DER 732 with 12 phr (parts per hundred) of DEH 26 hardener (Dow Chemical). A piece of polypropylene monofilament

was then used to tie together the seven fibers at the top. A second monofilament was then tied around the fibers, just below the knot of the first monofilament. This second monofilament was then slowly dragged down the length of the fiber bundle to allow the resin to fully coat each fiber and to remove excess resin. A knot was tied at the bottom to secure the bundle. The microcomposites were then placed in the oven and cured for 3 hours at 80°C.

3.2.3 Microscopy examination

An Anti-Mould Nikon Labophot2-POL polarized light microscope equipped with a Kodak DC290 camera was used to take cross-sectional images of the microcomposites. The seven-fiber bundle composites were first placed cross-wise in a mold of DER 331 resin, cured and then the surfaces polished. This allowed effective cross-sectional images to be captured.

3.2.4 Transverse compression testing of seven-fiber bundle microcomposites

A compression testing device developed in our laboratory was used to measure the load-displacement of the microcomposites. It consisted of a thermal-coupled set of vices that move at a constant rate (1.67 microns per second) to transversely compress a vertically mounted 10mm section of the fiber bundles. A fifty-pound load cell was used. The time, load and displacement were recorded using a data acquisition system, which was then transferred to an excel spreadsheet for analysis. The data was normalized using a program that took the compliance of the machine into account. The output was then used to plot the load-displacement curves of each specimen. Approximately thirty samples were

tested for each treatment group. Chuyang Zhang contributed Finite Element Analysis modeling of the failure schemes in both the X-direction and the XY-direction (shear).

3.2.5 Finite Element Analysis of seven fiber microcomposites – The Model

The Finite Element Analysis (FEA) modeling using ANSYS was adapted to simulate transverse compression of seven fiber microcomposites. The internal stress state of the microcomposite was recorded. In this model, a two-dimensional four-node viscoelastic element PLANE82 was employed for both the UHMPE fiber and the resin matrix, while a two-dimensional target element TARGE169 was used to simulate compression heads of the lateral compression device. In order to complete the contact pair with the target element, a two-dimensional contact element CONTA172 was modeled on the outer surface of the microcomposite where it touches the compression heads.

Figure 3.3 on the following page shows the two loading conditions modeled. The microcomposite was loaded as either Condition A or Condition B. Three assumptions were made to simplify the model:

1. Plain stress loading, three-dimensional model was simplified to two-dimensional model
2. Perfect bonding between the matrix and fiber
3. Elastic deformation

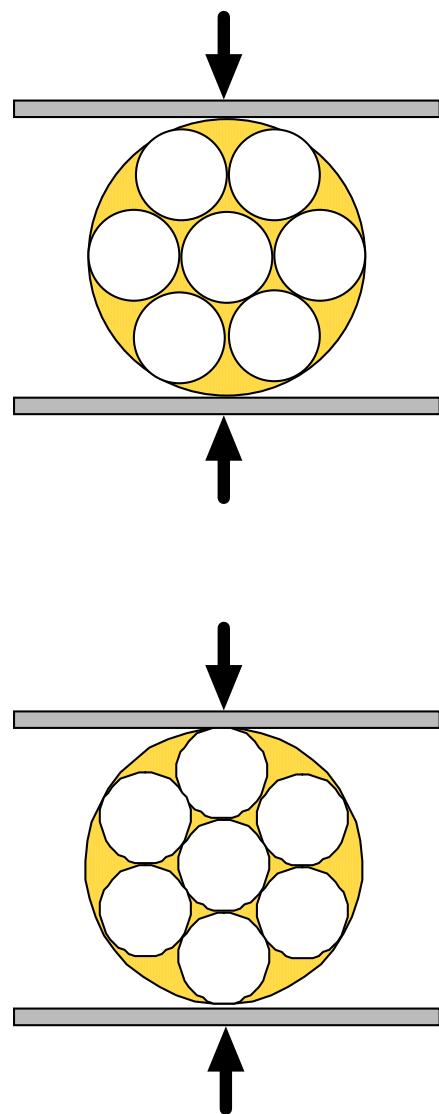


Figure 3.3 Schematic diagram of optimal seven-fiber composite transverse compression. Condition A (top) and Condition B (bottom) fiber arrangements are shown.

Table 3.1 Constants used in the FEA model

Fiber	Modulus(GPa)	Matrix	Fiber Diameter (microns)	Distance between fibers (microns)
1.2		1.4	35	2

Geometrical and mechanical constants used in the FEA model are listed in table 3.1. Both loading conditions are symmetrical about the X and Y axis, therefore, only a quarter of the composite was modeled as shown in figure 3.10. A symmetrical boundary condition about the Y axis was applied at location $x=0$, while all the nodes of x direction deformations (UY) at location $y=0$ were coupled.

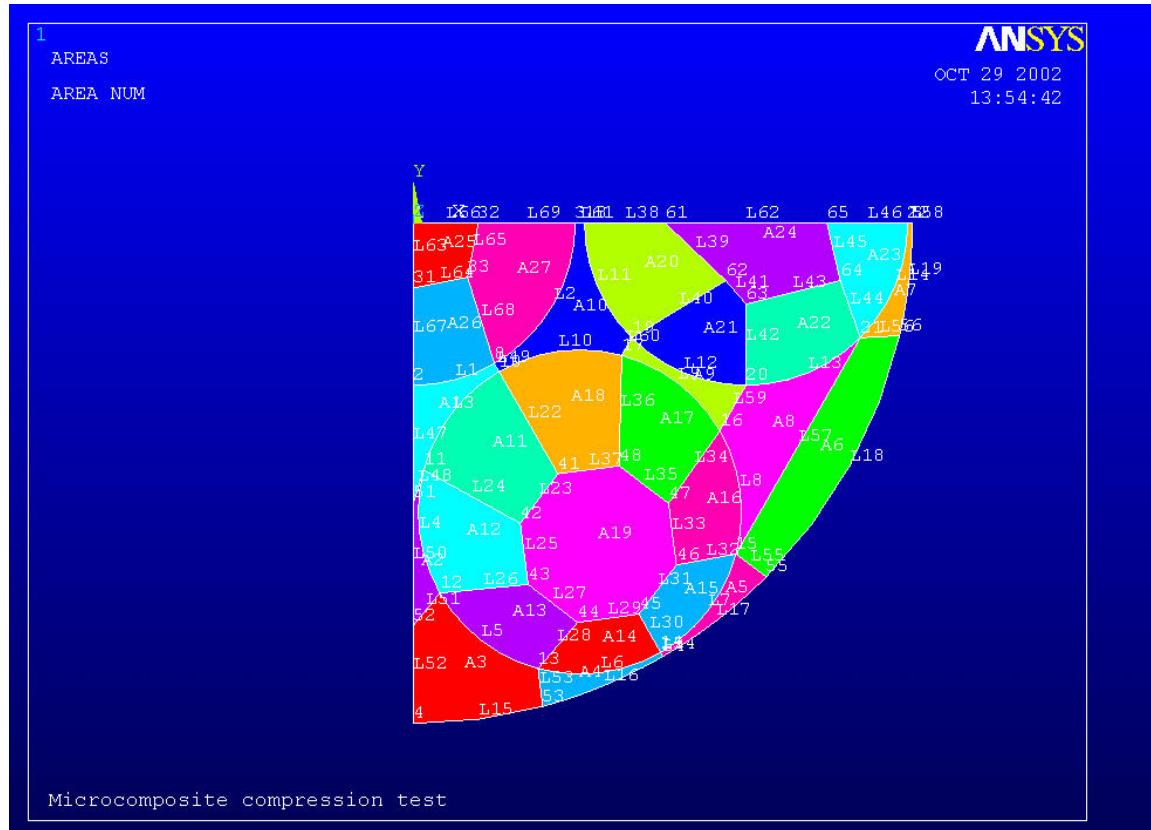


Figure 3.4 Geometric model of seven fiber microcomposite

From experimental data, the linear portion of the load-displacement curve of control sample ends at a displacement around 8 microns. Therefore, $UY = 4$ microns was applied to nodes at $y=0$. Plasma treated sample had a better bonding between fiber and matrix, The linear portion of plasma treated sample ends at a displacement around 12 microns. As a result, $UY = 6$ microns was applied to nodes at $y=0$ for plasma treated samples.

3.2.6 Scanning electron microscopy

The Hitachi S-3200 Variable pressure SEM model was used to inspect the delamination of the seven-fiber bundle microcomposites. In particular, the epoxy adhesion to the individual fibers was observed. The microcomposites were mounted on a sample platform and gold coated for three minutes in an Argon plasma sputtering device. SEM images were taken with a 5000x magnification.

3.2.7 Peeling test of laminated textiles.

ASTM standard D-2724 (Standard Test Methods for Bonded, Fused, and Laminated Apparel Fabrics) was used to determine the bonding strength between a plain weave UHMPE fabric and the epoxy resin. One set of samples was treated with a two min He/air plasma and the second set were left untreated. Fabric was cut in the warp direction into rectangles measuring 76.2 mm x 152.4 mm. A line was drawn across the width, 25.4 mm from the end to record where resin should be placed. Resin was applied using a wooden spatula across the 76.2 mm x 127 mm area until the fabric was wet and excess resin was then scraped off. A second sample was wetted with resin using the same procedure then placed on top of the first fabric leaving the one-inch dry ends free from resin. A piece of release film was placed in between these one-inch dry regions to ensure no transfer of resin to the region. Nine other sets of samples were prepared in the same manner, and then each was placed between layers of release film with a heavy caul plate placed on top. This equal amount of load applied to the samples ensured an even amount of resin to remain on the specimens for later comparison. The samples were cured in an

80°C oven for three hours and then post-cured for 2 hours at 100°C. The samples were then tested on the Sintech tensile tester with one of the resin-free ends in the upper grip and the second in the lower grip. Displacement occurred at a cross-head speed of 5mm/min. A 250lb load cell was used. As the peeling test was performed, the load required to delaminate the two layers was recorded. The five highest and five lowest peaks of the load-displacement graph were averaged to give the final result. A comparison was then made between the untreated and treated samples and a typical load-displacement curve was generated.

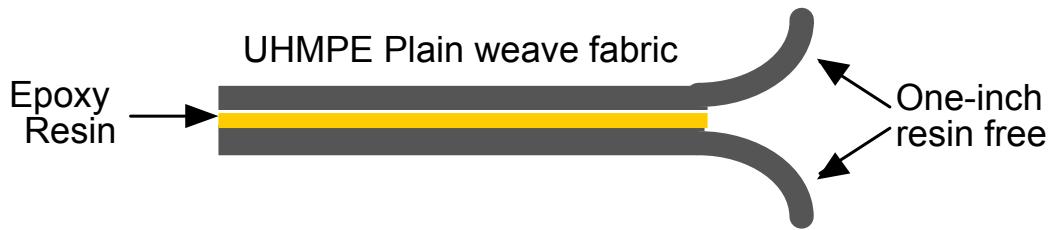


Figure 3.5 Schematic diagram of the peel test sample preparation.

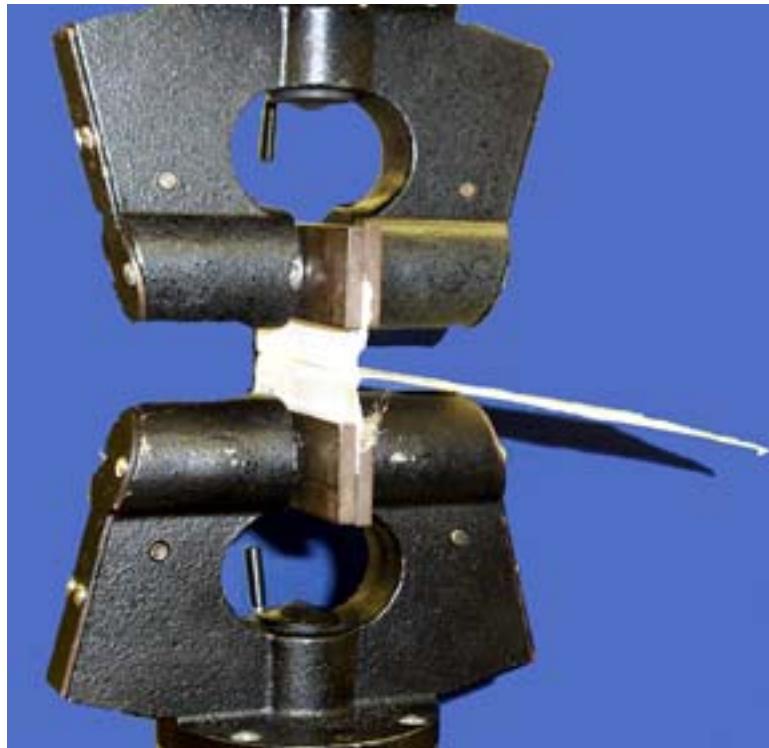


Figure 3.6 Photograph of the Sintech tensile testing machine gripping the resin free ends of the laminated fabric sample for peel strength analysis.

3.2.8 Shear strength of ten-ply UHMPE laminates.

Ten-ply UHMPE composites were manufactured using a hand-lay up process coupled with a vacuum pressured mold to create perfectly flat, void-free samples. The apparatus is shown in Figure 3.7. One composite was manufactured using 2 min He/air plasma treated fabric and the second composite was a control. Initially, to ensure complete wetting out of the fabric, each layer was placed individually into a shallow dish with the epoxy resin. Layer upon layer of wetted fabric was placed on top and excess air pockets were squeezed out. The ten-layer laminate was then placed onto an aluminum plate in between layers of release film and Mylar to allow release after curing. A double layer

vacuum bag seal was placed on top of the composite with a heavy caul plate and breather fabric to ensure even pressure was placed on the laminate. The sample was then vacuumed to remove air and to create a pressurized mold. The composite was cured under vacuum for three hours at 80°C and post-cured at 100°C for two hours. Complying with ASTM method D-3846 (Standard Test Method for In-Plane Shear Strength of Reinforced Plastics), each composite was then removed from the mold and cut into 12.7 mm × 101.4 mm coupons using a water-jet cutter. The thickness of the composites was recorded as 2 mm each. Two 1.2 mm deep cuts were machined using a Diamond Head blade in the center region of the composite strips one on each side, ~10mm separating the cuts. The actual notch separation distance was measured to the nearest 0.1 mm for later calculation. Figure 3.8 is a schematic of the machining required for this test procedure. The coupons were then mounted in an I-shaped support jig described in ASTM D-695 (Standard Test Method for Compressive Properties of Rigid Plastics) and placed in the Instron for shear tensile test. Figure 3.9 shows the support jig used. A torque wrench was used to tighten the four screws on the jig to a standard 1lb.ft.

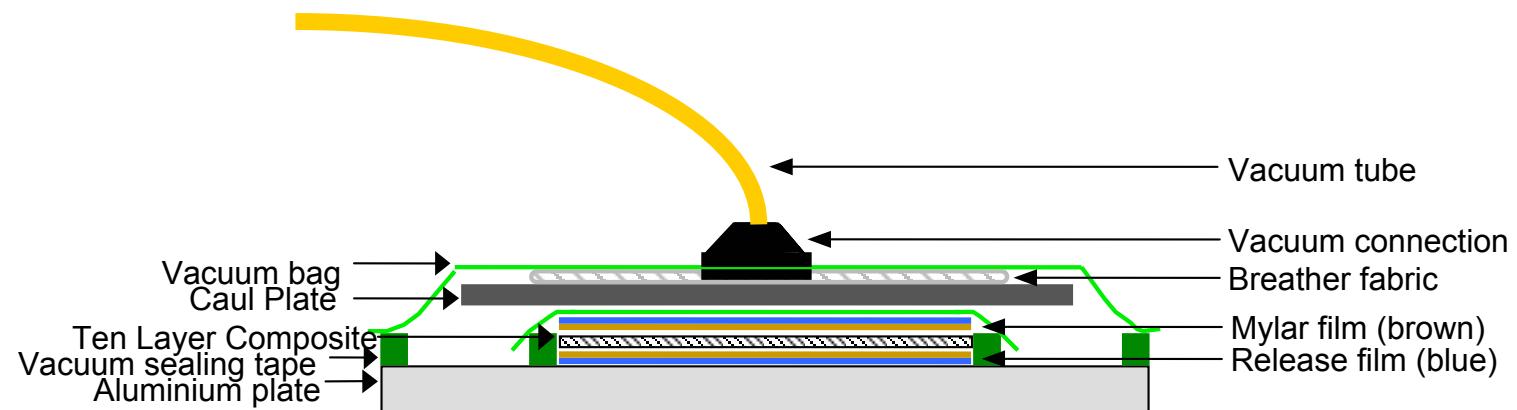


Figure 3.7 Schematic diagram of the composite manufacturing

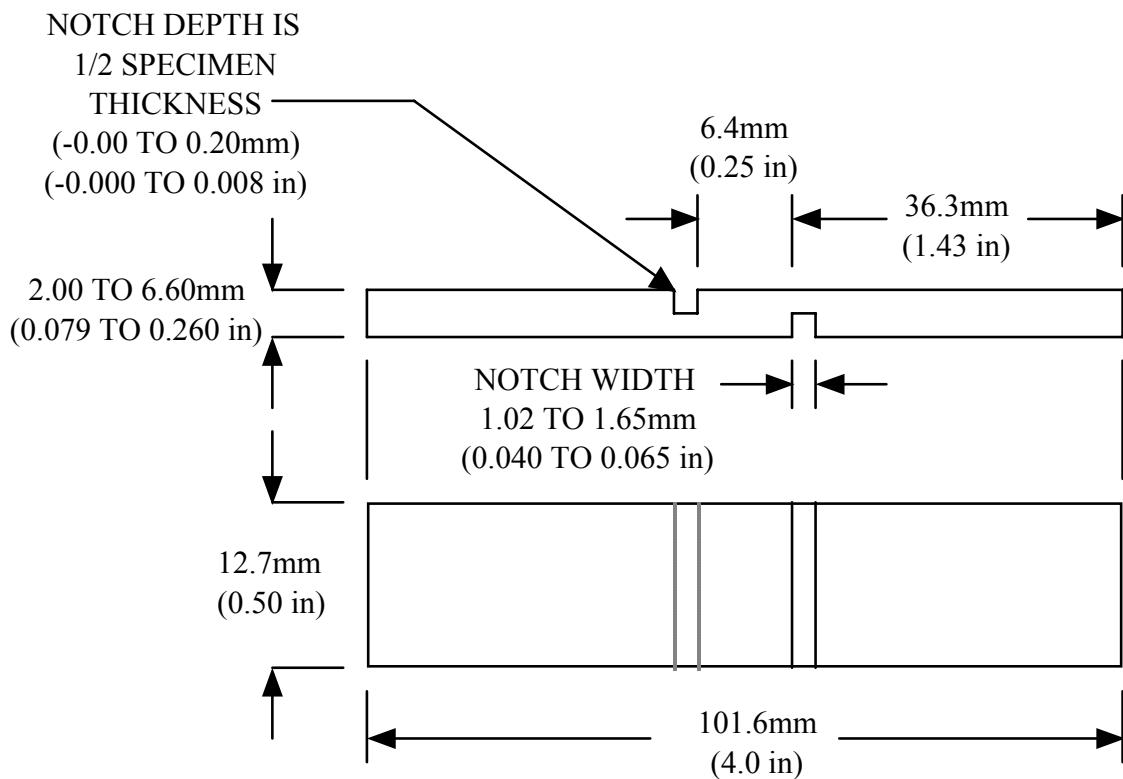


Figure 3.8 Schematic of machined samples for shear strength test (ASTM D-3846).

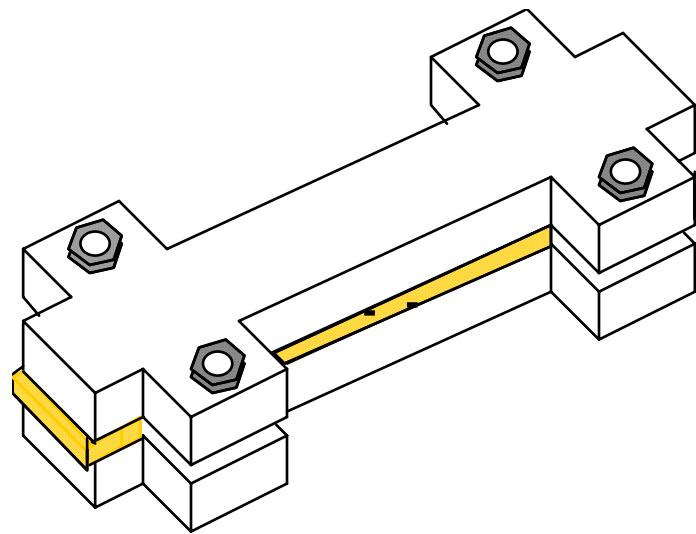


Figure 3.9 Supporting jig for shear strength composite samples. (ASTM D-695)

3.3 RESULTS AND DISCUSSION

3.3.1 Microscopy examination

Figure 3.10 shows typical images of the seven-fiber bundle composites with and without atmospheric plasma treatment. The control sample often shows an irregular array of fibers, due to the poor interfacial bonding and wetting out of the UHMPE fibers. This certainly reduces the transverse stiffness of the overall microcomposite while increasing the variation of the test results. In fact, in another study researching the interfacial bonding of boron/epoxy composites, it was found that the presence of just 10% of unbonded fiber reduced the modulus by 17% [10]. The atmospheric plasma treated samples however, consistently showed perfectly packed seven-fiber bundles that can be attributed to the improved chemical bonding between fiber and matrix. This regular array also contributed to the uniform load-displacement results collected from transverse compression testing and described in detail in the next section.

3.3.2 Transverse compression testing of seven-fiber bundle microcomposites

The transverse compression test generated a very interesting load-displacement trend for the atmospheric plasma treated samples. While the results for the control samples generated much less consistent load-displacement curves, due mostly to the fact that the bundles delaminated very early on, the plasma treated samples generated a very uniform and consistent set of load-displacement curves. Two such curves are demonstrated in Figure 3.11. The control and plasma treated sample both show an initial modulus of

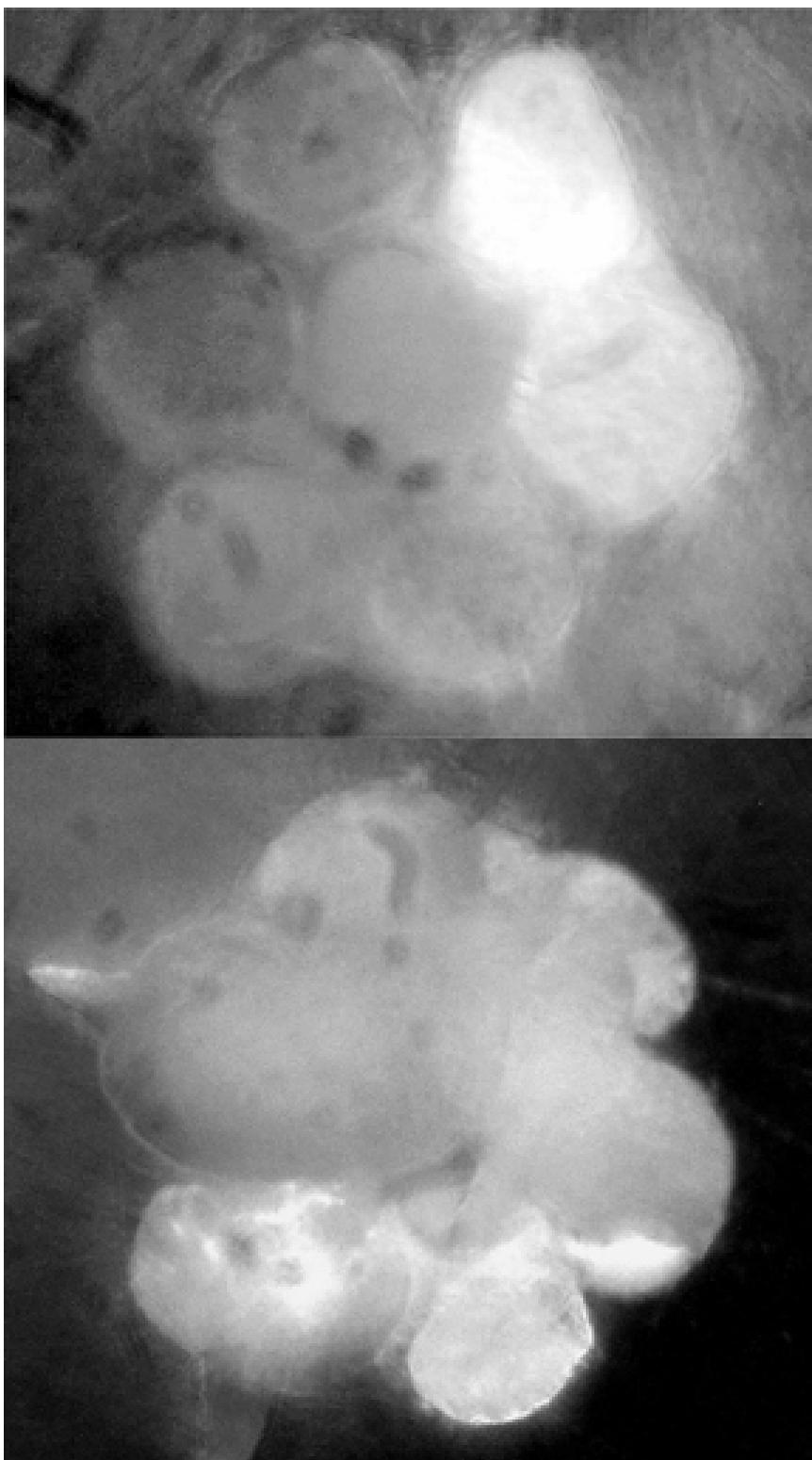


Figure 3.10 Microscopic photograph of seven-fiber microcomposite treated with (top) 2min He/air plasma. (bottom) Control

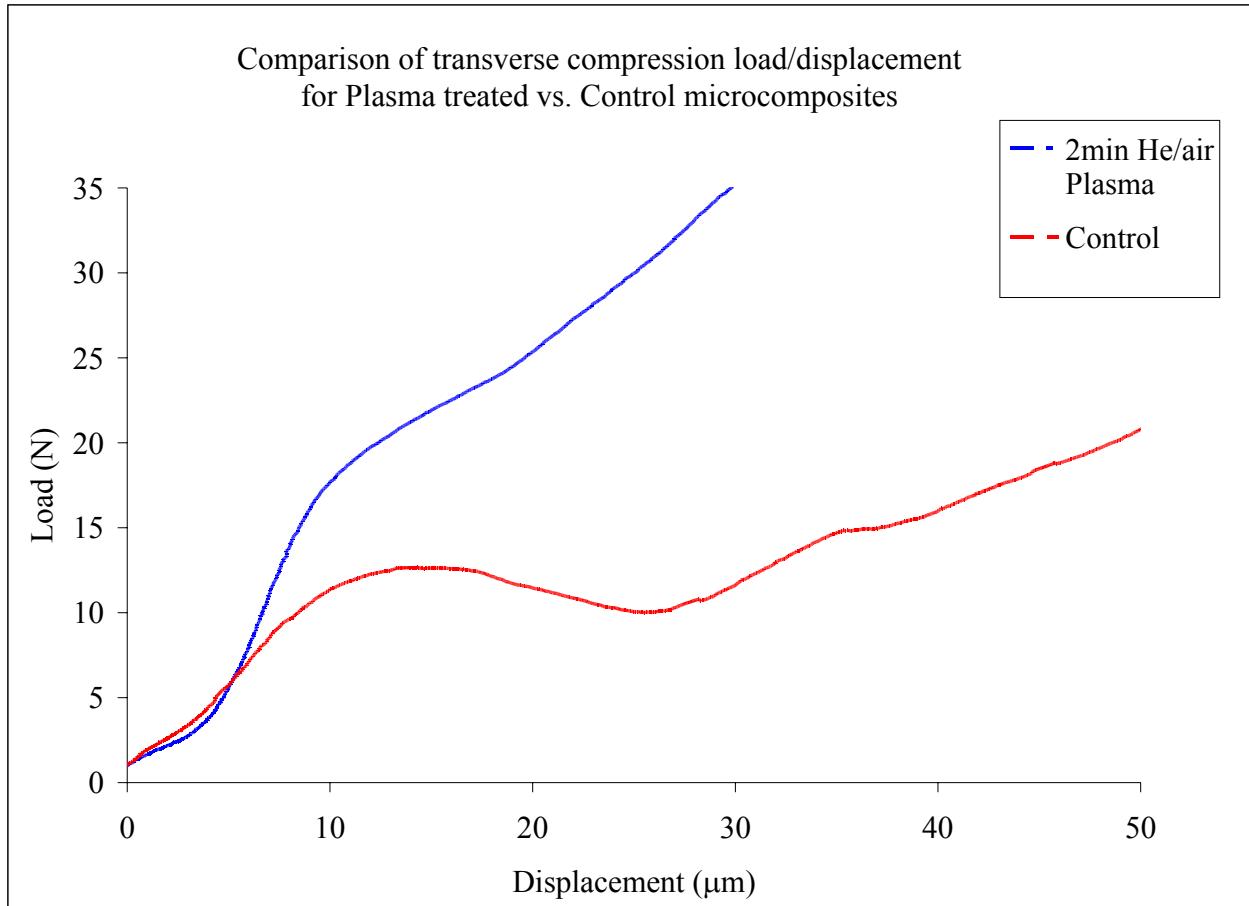


Figure 3.11 Typical Load-Displacement curves of transverse compression testing.

approximately the same value, but as transverse compression continues, the untreated sample yields at a much lower value than the treated sample. In fact, using the Meredith method, the yield modulus of the treated sample is almost 17.7N while the control yields around 11.9N. Considering the yield point corresponds to the onset of interfacial debonding, there is an increase of 49% in debonding load for the treated sample.

The irregular array of fibers observed in some of the control samples under the microscope proves to reinstate the fact that there is poor interfacial bonding between the

fiber and resin. The atmospheric plasma treated samples however, have a more consistent fiber packing arrangement with increased interfacial bonding and therefore yield at a much higher and more consistent load value. The full array of load-displacement curves generated by the transverse compression test can be located in the appendix.

3.3.3 Finite Element Analysis of seven fiber microcomposites

The stress distribution in the composite was obtained as well as deformed shape and reaction force. The load displacement curve was then plotted and used as a judgment of the accuracy of the FEA model. Good agreement was found between experimental data and FEA model as shown in figure 3.12. Therefore, the stress calculated in the FEA model can be used to characterize the interfacial bonding strength and failure strength. Taking real values from the experimental data of the compression test, it was found that in the X-direction for a Type-A fiber arrangement, the Control had a yield stress of 35.8GPa (Fig. 3.13) while the Plasma treated sample yielded at 56.2GPa (Fig. 3.14). This is a 57% increase. For the same Type-A fiber arrangement in the XY-direction (shear), the Control yielded at 160GPa (Fig. 3.15), while the Plasma treated yielded at 251GPa (Fig. 3.16). This is also a 57% increase in yield stress. The high values noted here compared to the experimental values, are a result of the localized study. One would expect a much higher yielding value to occur at the microscopic level than in an actual full-scale experiment.

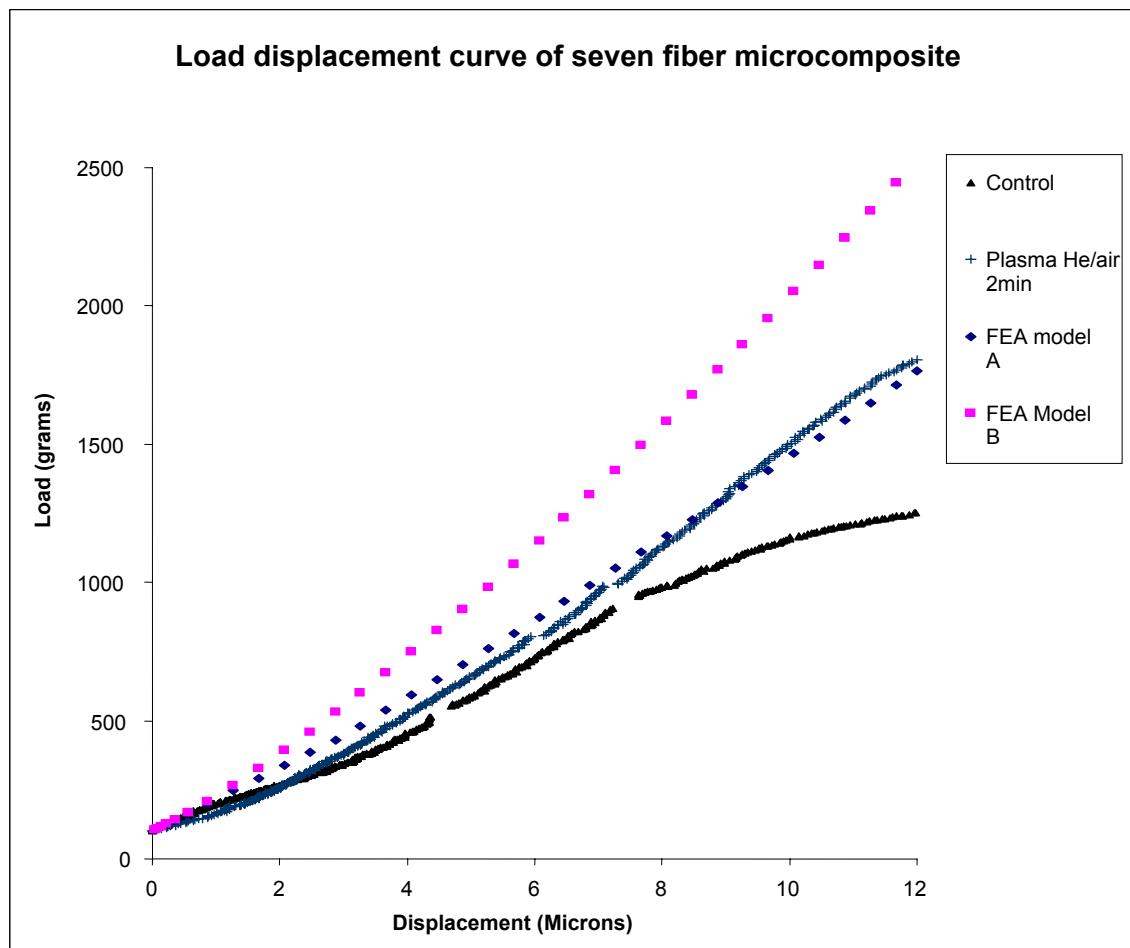


Figure 3.12 Load-displacement curve of seven fiber microcomposite with FEA modeling

The Type-B fiber arrangement was also analyzed using Finite Element Modeling. In the X-direction, the Control yielded at 4.41GPa (Fig. 3.17) while the Plasma treated sample yielded at 7.3GPa (Fig. 3.18). This is a 66% increase. In the XY-direction (shear), the Control yielded at 27.3GPa (Fig. 3.19), while the Plasma treated sample yielded at 31.2GPa (Fig. 3.20). This is a 14.4% increase.

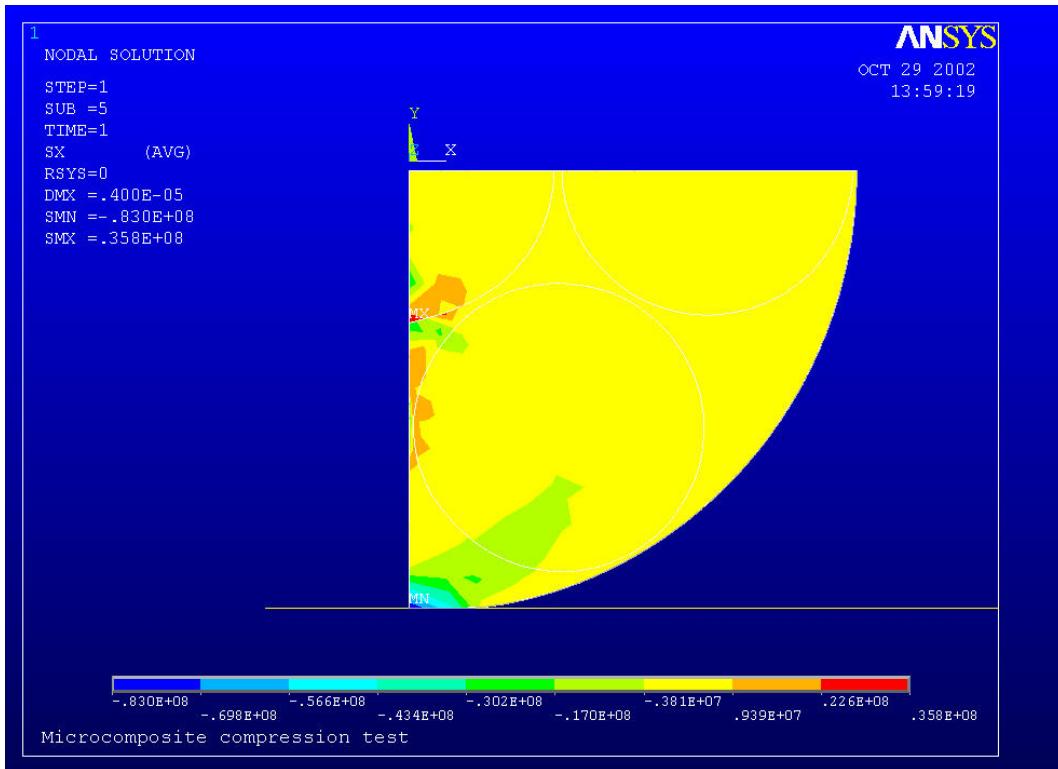


Figure 3.13 Finite Element Analysis (FEA) for failure of Type-A fiber arrangement Control sample in the X-direction.

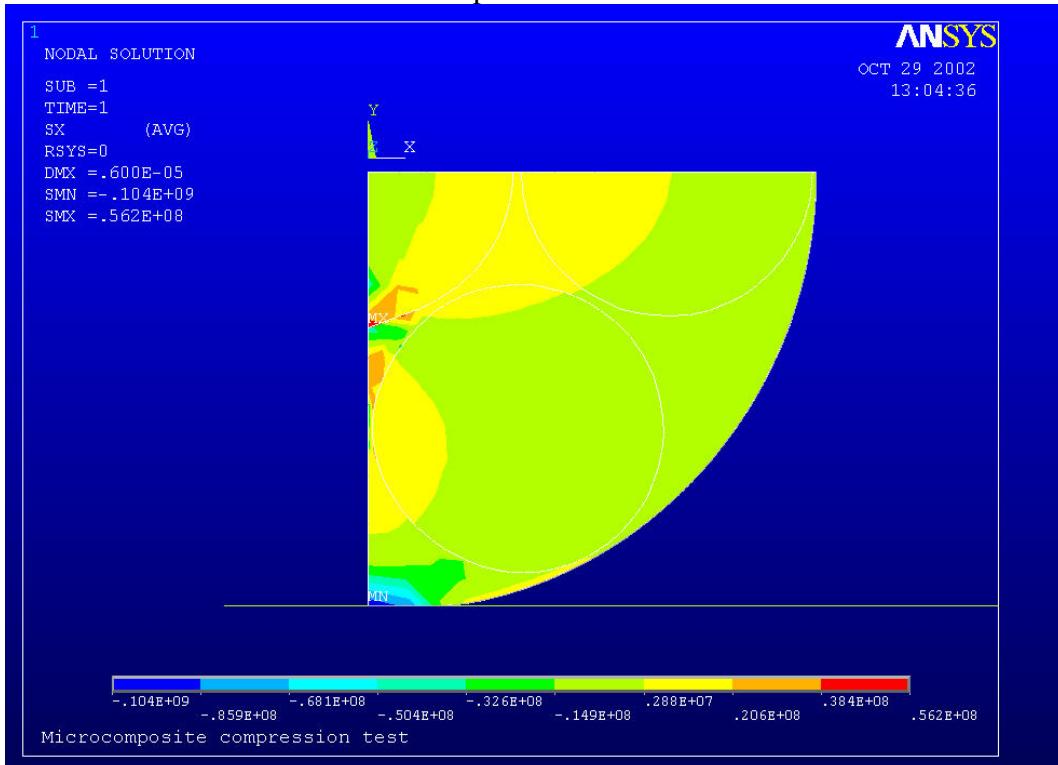


Figure 3.14 FEA for failure of Type-A fiber arrangement Plasma-treated sample in the X-direction.

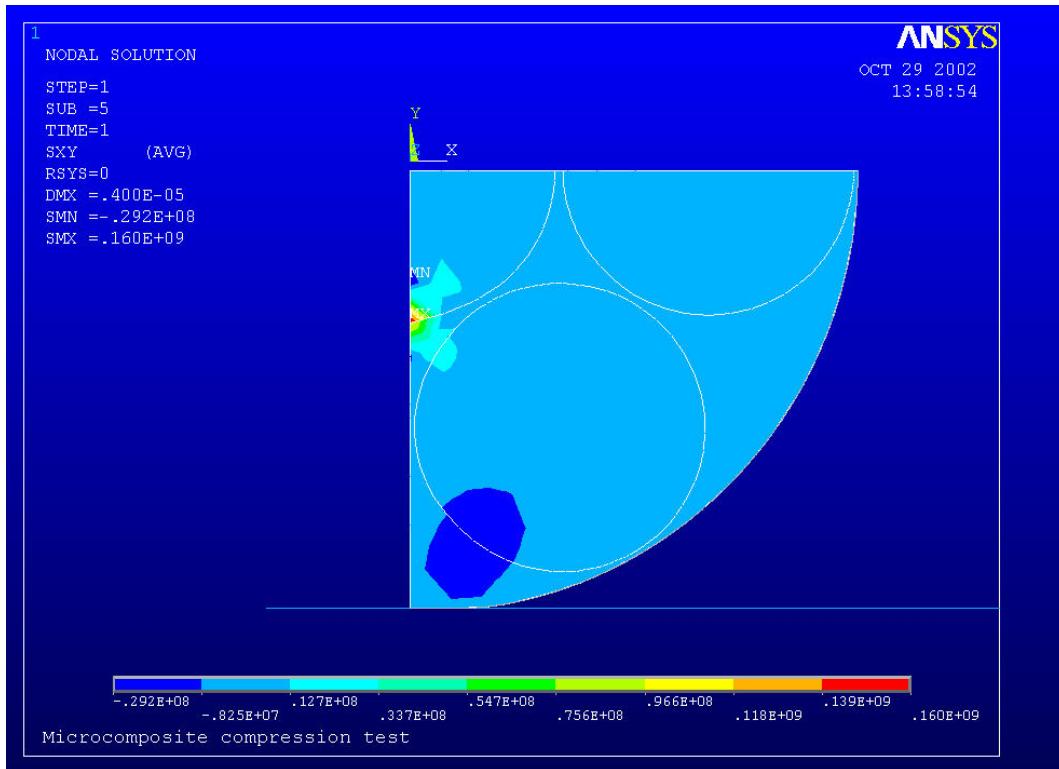


Figure 3.15 FEA for failure of Type-A fiber arrangement
Control sample in the XY-direction.

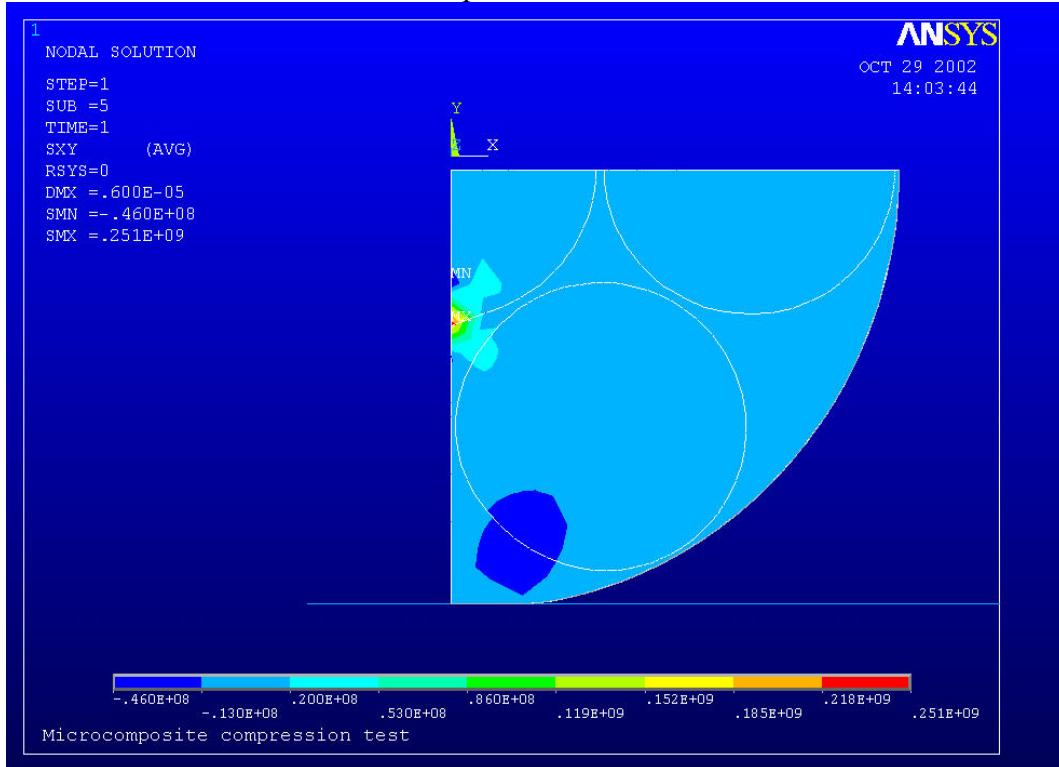


Figure 3.16 FEA for failure of Type-A fiber arrangement
Plasma-treated sample in the XY-direction.

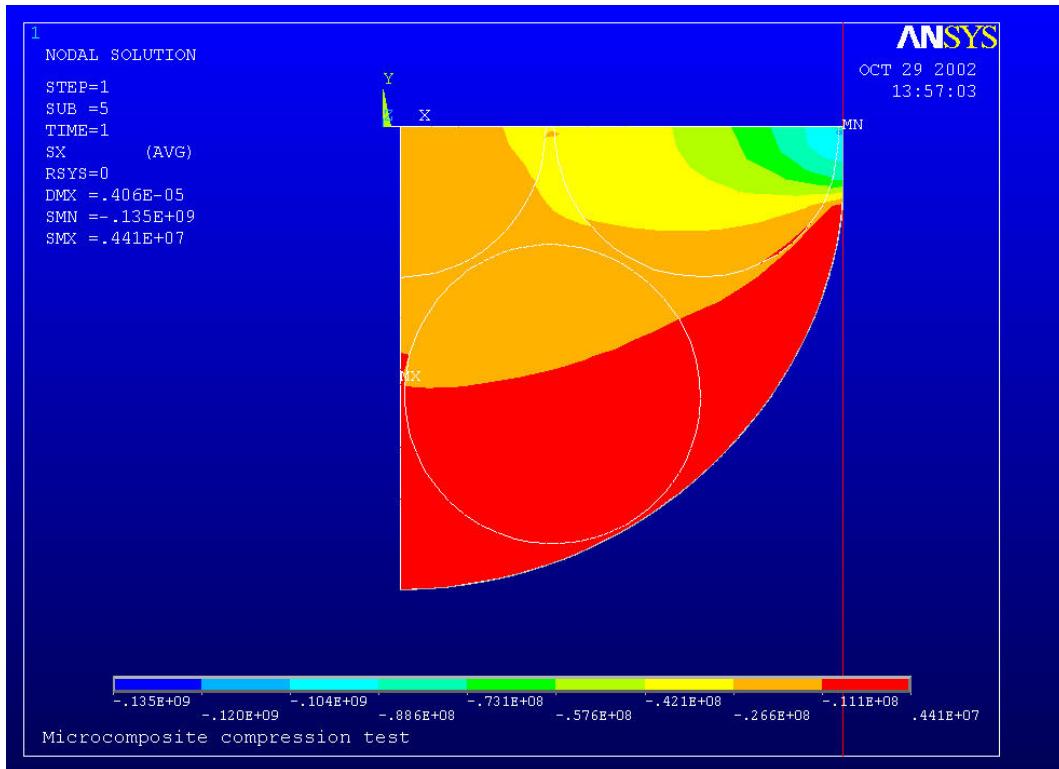


Figure 3.17 FEA for failure of Type-B fiber arrangement
Control sample in the X-direction.

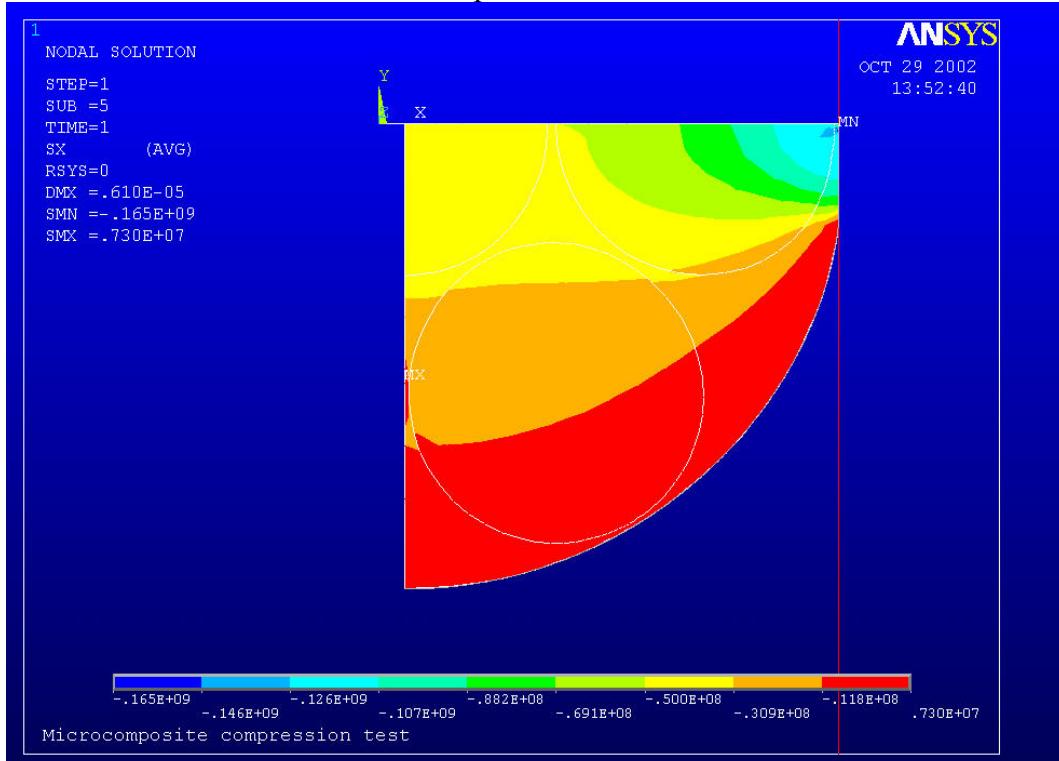


Figure 3.18 FEA for failure of Type-B fiber arrangement
Plasma-treated sample in the X-direction.

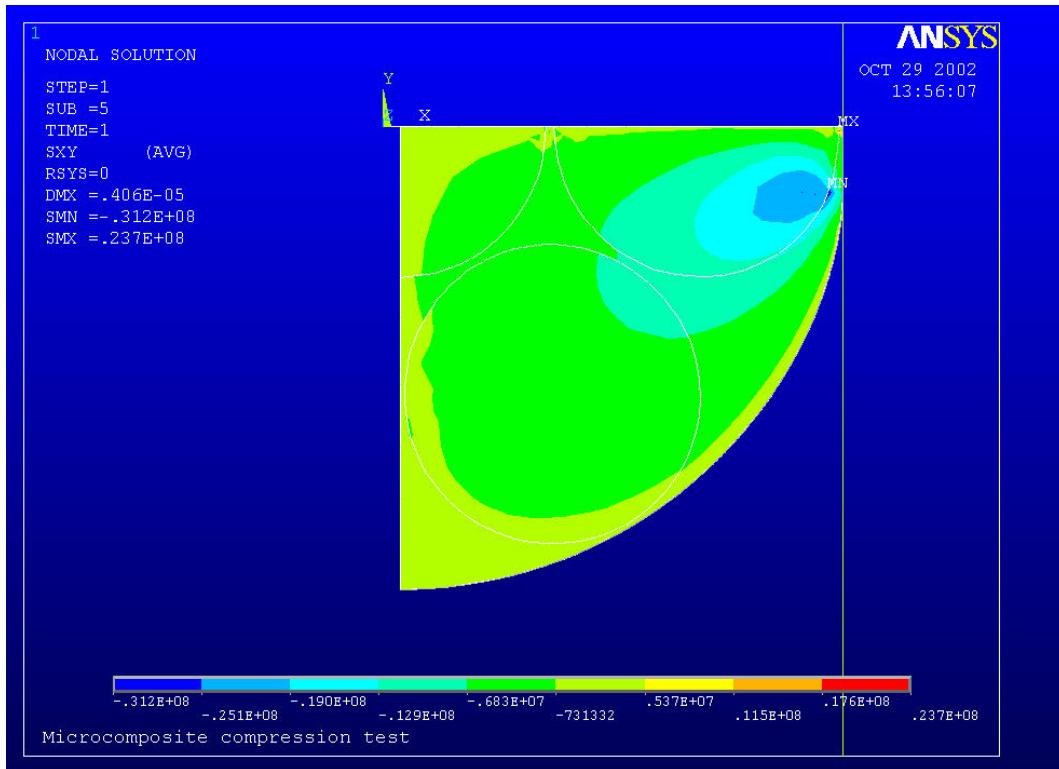


Figure 3.19 FEA for failure of Type-B fiber arrangement
Control sample in the XY-direction.

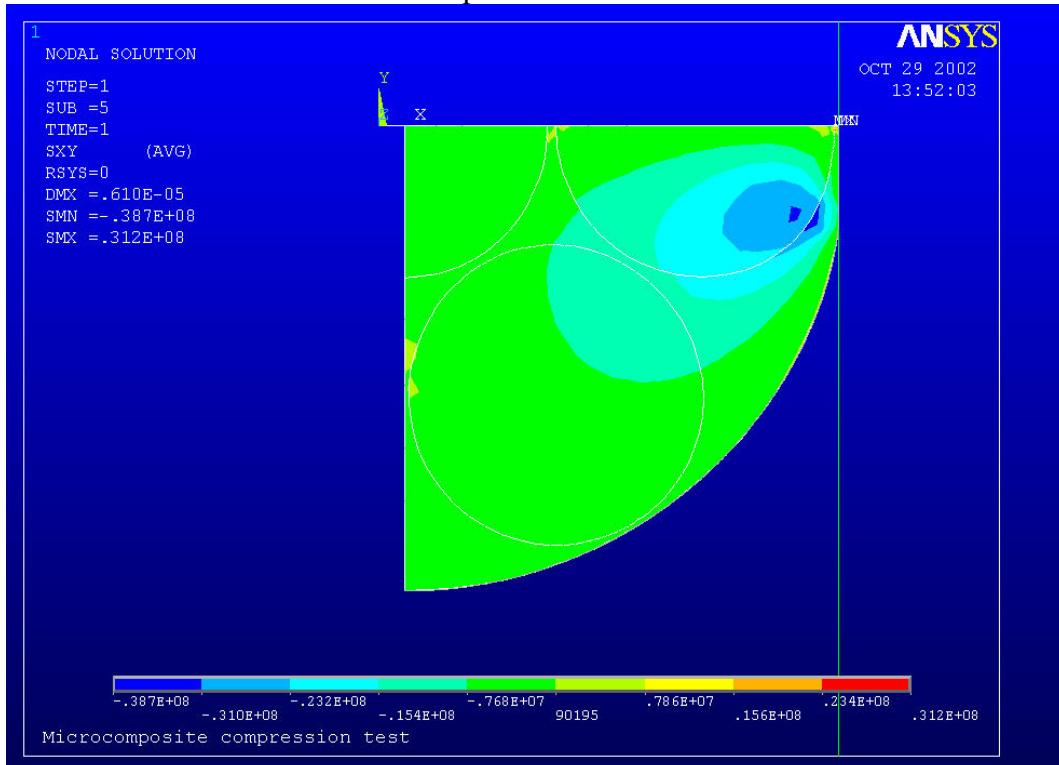


Figure 3.20 FEA for failure of Type-B fiber arrangement
Plasma-treated sample in the XY-direction.

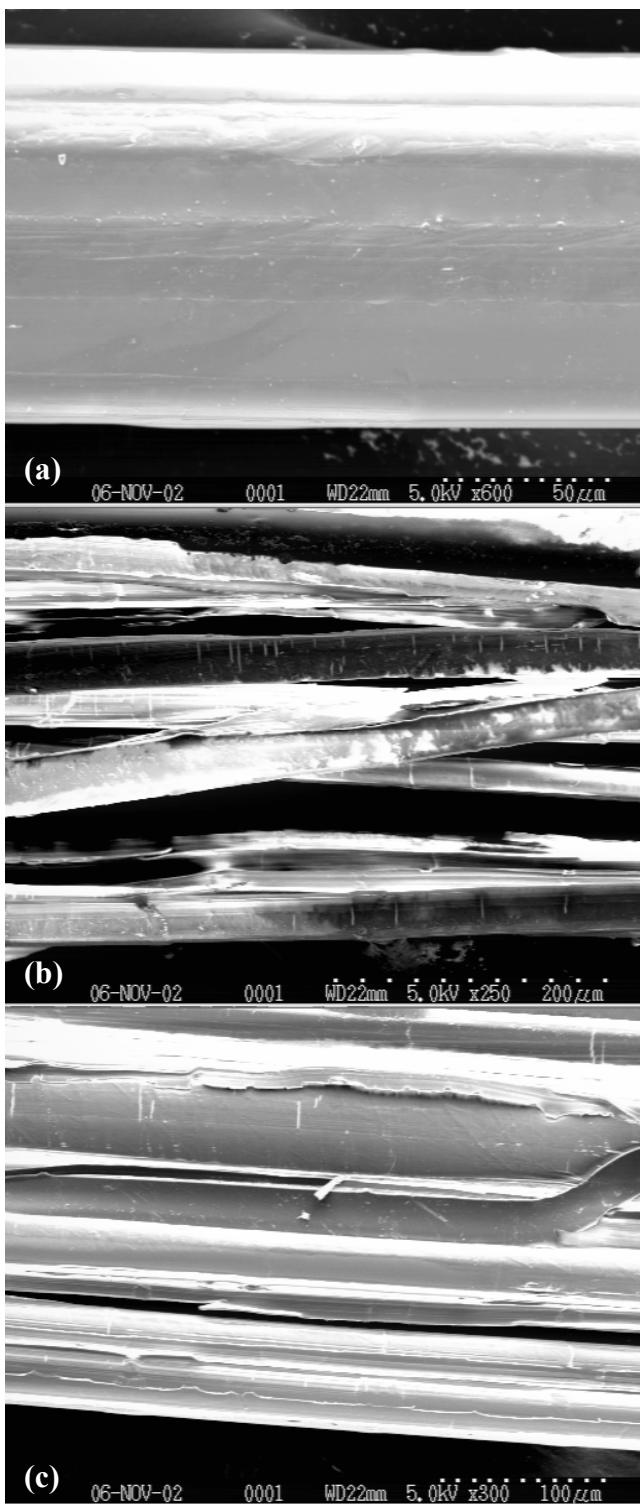


Figure 3.21 (a) SEM of Seven fiber microcomposite before compression, at 600x (b) SEM of Control sample after compression with total delamination at interface, at 250x (c) SEM of Plasma treated sample after compression with bulk matrix failure, interface remains intact, at 300x.

3.3.4 Scanning Electron Microscopy

Figure 3.21 (a) shows the seven-fiber bundle microcomposite as a complete unit, before transverse compression testing. The individual fibers can be distinguished from the epoxy matrix. Figure 3.21 (b) shows the failure mechanism of the control composite. The individual fibers have completely separated from the epoxy resin, leaving a clean fiber surface. The atmospheric plasma treated microcomposite in figure 3.21 (c) shows compression failure in the bulk matrix. The fibers are observed with large pieces of epoxy still attached to the surface. This suggests that the atmospheric plasma treatment has microscopically improved the interfacial bonding strength to an extent that failure occurs in the bulk matrix instead of at the interface region.

3.3.5 Peeling test of laminated textiles

The average load values of the five maximum and five minimum peaks collected from the Peeling test revealed significant differences between the treated and untreated samples. In fact, the average bonding strength recorded for the 2min He/air plasma treated sample tested was 3.94KPa while the control sample bonding strength was recorded at 2.16KPa. That is an 82.5% increase in bonding strength after atmospheric plasma treatment. Table 3.2 shows the values calculated for each sample tested as well as the average totals. The standard deviation and variance was also included. A comparison between typical load-displacement curves for both the treated and control samples is displayed in Figure 3.22. Here the discrepancy between bonding strength of each set of samples is clearly demonstrated. The plasma treated fabric shows a peel strength curve almost double the response of the control sample.

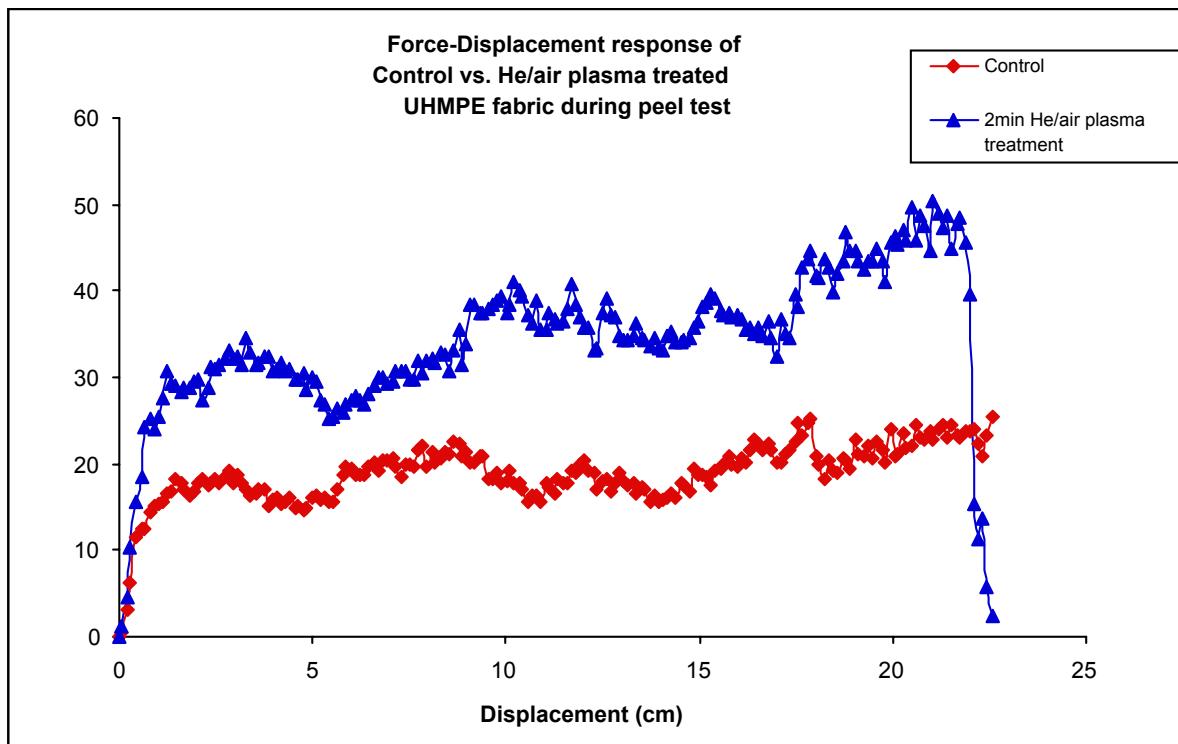


Figure 3.22 Typical load-displacement response of control vs. plasma treated UHMPE fabric during peel test.

Table 3.2 Average values for the five maximum and five minimum peaks of each sample tested during the peel test. An 82.5% increase in strength was calculated by the plasma treatment.

Control

Average Max Peaks (KPa)	Average Min Peaks (KPa)	Average Totals	St Dev	St dev/(n)^-2
2.37	1.99	2.18		
2.61	2.15	2.38		
1.99	1.61	1.80		
2.85	2.58	2.72		
2.81	2.56	2.69		
2.45	1.90	2.18		
1.61	1.41	1.51		
2.22	1.97	2.10		
2.03	1.72	1.88		
2.33	1.99	2.16	0.40	0.13

2min He/Air Plasma treated

Average Max Peaks (KPa)	Average Min Peaks (KPa)	Average Totals	St Dev	St dev/(n)^-2
4.96	4.14	4.55		
4.11	3.55	3.83		
3.47	3.02	3.24		
4.90	4.09	4.50		
4.23	4.23	4.23		
4.94	4.39	4.66		
3.44	2.53	2.99		
3.77	3.27	3.52		
4.23	3.65	3.94	0.64	0.23

3.3.6 Visual inspection of the laminated composites

The first observation noted between the treated and untreated composites was the color of the cured resin. The control sample appeared a lighter shade than the more yellow color of the plasma treated sample. This is an indication of the level of wetting each sample incurred during manufacturing. The control fabrics took much longer to soak up the resin than did the atmospheric plasma treated fabrics. Another observation noted was during the coupon cutting of the samples. The plasma treated composites cut more easily with initial edge trimming using a diamond edge blade compared to the control which showed signs of slight delamination. A water-jet cutter was commissioned to prepare the coupons to ensure straight even edges with little delamination.

3.3.7 Shear strength of ten-ply UHMPE laminates.

Twenty-three control samples and twenty plasma treated samples were tested on the Instron 5500R with a 1000-lb load cell at 1.3mm/min extension rate. The maximum load each specimen withstood before delamination occurred was recorded. Table 3.3 gives the actual data recorded and the conversions used to calculate the final average interlaminar shear strength of each set of samples. The control showed average shear strength of 5.77 MPa while the 2-min He/air plasma treated samples demonstrated average shear strength of 7.25 MPa. That is a statistically significant 25.7% increase in interlaminar shear strength of the plasma treated versus the untreated fiber reinforced composite. Figure 3.23 shows graphically the difference in interlaminar shear strength of the control versus the 2-min He/air atmospheric plasma treated sample. Here it is clearly

shown that the plasma treated samples give a greater response to shear load than do the untreated samples.

Table 3.3 Data collected from Interlaminar Shear Strength test.

CONTROL	LENGTH (cm)	AREA (cm ²)	MAX LOAD (N)	In-Plane Shear Strength (MPa)
A	1.76	2.23	932.74	4.18
B	2.13	2.71	1121.63	4.15
C	1.76	2.23	1314.42	5.89
D	1.70	2.16	1232.06	5.70
E	1.81	2.29	1012.45	4.41
F	1.64	2.09	1328.30	6.37
G	2.14	2.71	1403.76	5.17
H	2.17	2.75	1288.73	4.68
I	2.15	2.73	1078.14	3.95
J	2.15	2.73	1619.00	5.94
K	2.13	2.70	1585.95	5.87
L	1.58	2.00	1404.78	7.01
M	2.14	2.71	1515.48	5.59
N	1.66	2.11	1358.01	6.44
O	2.13	2.70	1596.05	5.91
P	1.79	2.28	1566.59	6.87
Q	2.13	2.70	1725.77	6.38
R	1.77	2.25	1391.19	6.17
S	1.68	2.13	1300.94	6.10
T	1.73	2.20	1560.61	7.08
U	2.14	2.71	1538.76	5.67
V	2.13	2.71	1645.15	6.07
W	1.60	2.03	1440.55	7.08
AVERAGE				5.77
StDev				0.95
Variance				0.20
PLASMA	LENGTH (cm)	AREA (cm ²)	MAX LOAD (N)	In-Plane Shear Strength (MPa)
1	1.73	2.19	1607.04	7.33
2	1.73	2.19	1581.97	7.21
3	0.45	0.57	669.04	11.73
4	1.72	2.18	1335.37	6.12
5	0.28	0.35	322.71	9.20
6	0.34	0.43	496.09	11.55
7	1.72	2.19	1065.92	4.88
8	0.86	1.10	824.44	7.51
9	0.52	0.66	749.54	11.39
10	1.71	2.18	1169.99	5.37
11	1.73	2.19	1029.52	4.69
12	1.40	1.77	1037.61	5.85
13	1.76	2.23	1014.34	4.55
14	1.72	2.19	1155.14	5.28
15	1.11	1.40	991.77	7.06
16	1.31	1.66	1206.80	7.27
17	1.35	1.71	1291.85	7.53
18	1.73	2.19	1124.81	5.13
19	1.19	1.51	1226.84	8.10
20	1.46	1.85	1339.75	7.23
AVERAGE				7.25
StDev				2.24
Variance				0.50

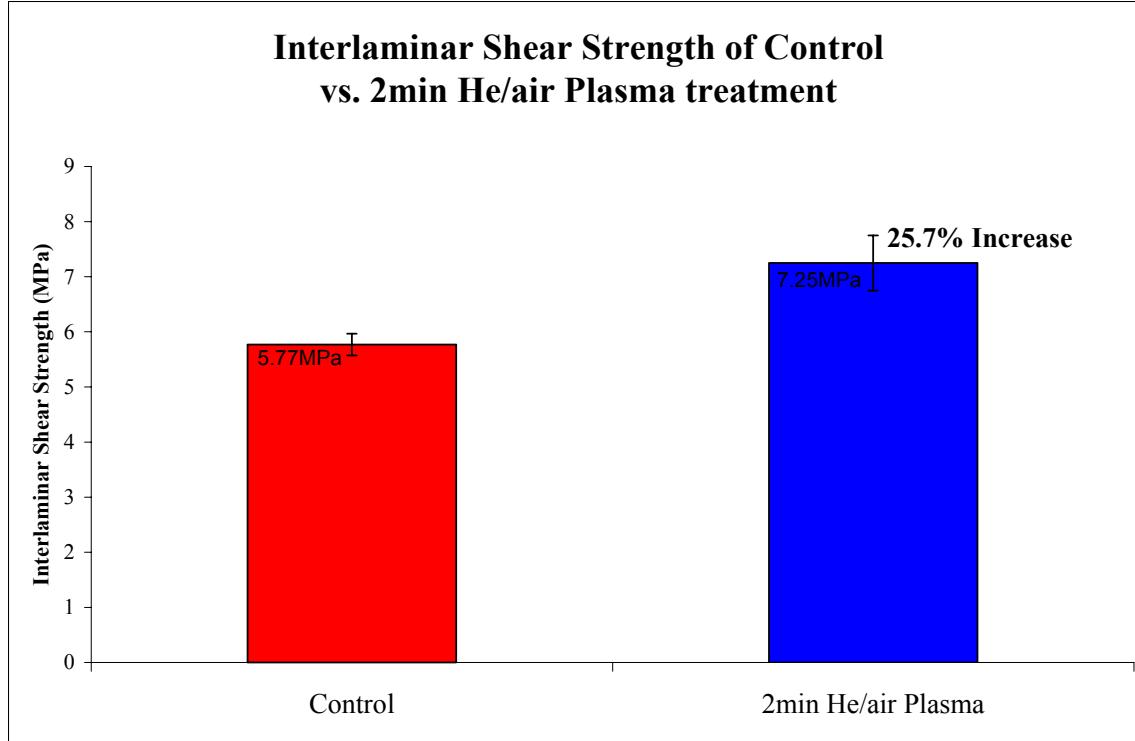


Figure 3.23 Interlaminar Shear Strength graph of Control vs. He/air plasma treatment

3.4 CONCLUSION

The delamination responses of 2-min He/air atmospheric plasma treated UHMPE composites were studied. The following conclusions can be made:

- (1) Transverse compression of the seven-fiber microcomposites demonstrated a 49% increase in yield load for the plasma treated sample compared to the untreated. This can be attributed to greater adhesion between fibers of the plasma treated set and also greater uniformity of the seven-fiber bundle arrangement.
- (2) Compared to the experimental results, the FEA model is most like the Type-A fiber arrangement with a combination of X and XY-direction failure. The 57% increase is close to the 49% increase found in the experiment.
- (3) The peel test demonstrated an 82.5% increase in bonding strength after atmospheric plasma treatment.
- (4) The shear strength test showed that compared with the control; there was a 25.7% increase in interlaminar shear strength of the atmospheric plasma treated samples.

Each set of delamination tests performed in this chapter showed a significant increase in strength between the UHMPE fiber and the epoxy resin after 2-min He/air atmospheric plasma treatments. Since the plasma treatment can increase bonding strength in peel test more than interlaminar shear strength in shear strength test, we conclude that the plasma treatment is more effective in increasing Mode I type of debonding stress than Mode II type debonding stress in a real composite.

Beginning with micromechanical experiments, namely the microbond test and a microcomposite transverse compression test through to a peeling test of laminated plain weave fabric and concluding with a ten-layer plain weave UHMPE flat panel composite,

this experiment has demonstrated clearly that an atmospheric plasma treatment can increase fiber-matrix bonding at the microscopic level as well as the full-scale composite production level.

3.5 REFERENCES

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Appendix A. 0-Days Microbond Data

Report Date: Feb 15, 2002

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS1A – CONTROL 0 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Jan 30, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus	Fiber Tenacity
	Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.000	3.67	0.2	0.45085	0.01	183.39	0.52
2	7.00	50.000	9.54	0.2	0.36672	0.07	371.18	1.36
3	7.00	50.000	7.09	0.4	0.88900	0.04	352.67	1.01
4	7.00	50.000	3.18	0.2	0.40005	0.01	147.61	0.45
5	7.00	50.000	3.18	0.4	0.77470	0.03	157.19	0.45
6	7.00	50.000	9.29	0.7	1.35890	0.07	460.61	1.33
7	7.00	50.000	5.62	0.5	1.01600	0.04	304.04	0.80
8	7.00	50.000	8.56	0.3	0.50800	0.05	465.53	1.22
9	7.00	50.000	8.31	0.4	0.70485	0.06	463.30	1.19
10	7.00	50.000	9.05	0.4	0.80645	0.09	375.11	1.29
11*	7.00	50.000	4.89	0.7	1.38430	0.15	67.37	0.70
12	7.00	50.000	6.11	0.2	0.36195	0.04	318.52	0.87
13	7.00	50.000	5.38	0.4	0.70485	0.04	303.54	0.77
14	7.00	50.000	5.14	0.3	0.69215	0.04	258.90	0.73
15	7.00	50.000	5.14	0.2	0.40005	0.04	190.44	0.73
16	7.00	50.000	2.69	1.3	2.67970	0.24	44.37	0.38
17	7.00	50.000	5.87	0.5	0.90170	0.07	261.98	0.84
18	7.00	50.000	10.03	0.5	0.97155	0.10	522.66	1.43
19	7.00	50.000	12.96	0.4	0.83820	0.24	302.59	1.85
20	7.00	50.000	11.98	0.3	0.64770	0.14	384.08	1.71
21	7.00	50.000	12.47	0.4	0.86360	0.18	342.33	1.78
22	7.00	50.000	10.27	0.4	0.76835	0.14	326.03	1.47
Mean	7.00	50.000	7.64	0.4	0.72128	0.08	324.58	1.09
Stdv	0.00	0.000	3.04	0.1	0.25646	0.06	105.91	0.43

*Exclude

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS2A – HE/O2/AIR 30SEC 0 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Jan 30, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.260	6.50	0	-0.01521	0.40	331.82	0.94
2	7.00	50.381	3.91	-0.2	-0.33756	0.01	189.65	0.56
3	7.00	50.108	5.87	0	0.01227	0.02	428.83	0.84
4	7.00	50.216	4.16	0	0.03161	0.01	293.54	0.59
5	7.00	50.127	7.58	0.2	0.36593	0.07	269.06	1.08
6	7.00	50.044	6.85	0	0.08834	0.02	*****	0.98
7	7.00	50.454	8.31	0.1	0.14474	0.03	*****	1.19
8	7.00	50.038	8.31	0.1	0.16417	0.04	629.24	1.19
9*	7.00	50.190	2.20	0.2	0.31629	0.01	92.04	0.31
10	7.00	50.295	7.34	0.1	0.27002	0.07	356.65	1.05
11	7.00	50.000	10.03	0.3	0.67801	0.04	692.15	1.43
12	7.00	50.003	13.94	0.4	0.79226	0.10	747.71	1.99
13	7.00	50.000	4.89	0.2	0.48895	0.02	435.55	0.70
14	7.00	50.000	7.34	0.5	0.97155	0.05	336.84	1.05
15	7.00	50.000	11.49	0.4	0.83041	0.14	355.99	1.64
16	7.00	50.003	19.07	0.5	1.02863	0.35	570.97	2.72
17	7.00	50.003	11.00	0.4	0.78346	0.08	577.04	1.57
18	7.00	50.000	11.98	5	1.01600	0.12	476.11	1.71
19	7.00	50.000	10.03	0.3	0.55245	0.09	415.43	1.43
20	7.00	50.000	9.54	0.5	0.95106	0.13	301.70	1.36
Mean	7.00	50.102	8.86	0.2	0.46406	0.07	435.78	1.27
Stdv	0.00	0.147	3.67	0.2	0.41899	0.08	158.23	0.52

*Exclude

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS3A – HE/O2/AIR 1MIN 0 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Jan 30, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus	Fiber Tenacity
	Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.000	8.80	0.7	1.43510	0.12	346.84	1.26
2	7.00	50.000	4.16	0.2	0.45085	0.02	*****	0.59
3	7.00	50.000	11.98	0.3	0.61595	0.09	618.94	1.71
4	7.00	50.000	8.07	0.3	0.55880	0.05	437.31	1.15
5	7.00	50.000	19.32	0.6	1.23190	0.23	652.74	2.76
6	7.00	50.000	25.19	0.5	1.06680	0.40	617.54	3.60
7	7.00	50.000	13.94	0.4	0.88900	0.15	537.07	1.99
8	7.00	50.000	12.96	0.3	0.59055	0.15	447.01	1.85
9*	7.00	50.000	14.18	0.6	1.28905	0.24	363.11	2.03
10	7.00	50.000	12.72	0.5	0.95885	0.19	322.05	1.82
11	7.00	50.000	12.23	0.2	0.49530	0.09	683.54	1.75
12	7.00	50.000	9.29	0.3	0.66040	0.10	613.65	1.33
13	7.00	50.000	16.38	0.3	0.53340	0.22	513.49	2.34
14	7.00	50.000	12.23	0.4	0.77470	0.20	330.10	1.75
15	7.00	50.000	10.52	0.3	0.58420	0.12	366.78	1.50
16	7.00	50.000	33.01	1	2.02565	1.84	231.65	4.72
17	7.00	50.000	20.79	0.8	1.64465	1.04	378.24	2.97
18	7.00	50.000	14.18	0.7	1.41605	0.38	324.80	2.03
19	7.00	50.000	5.87	0.6	1.24460	0.21	125.04	0.84
20	7.00	50.000	4.65	0.7	1.42240	0.23	90.44	0.66
21*	7.00	50.000	6.11	0.6	1.16205	0.10	188.22	0.87
Mean	7.00	50.000	13.52	0.5	0.99441	0.30	421.07	1.93
Stdv	0.00	0.000	6.97	0.2	0.44992	0.42	172.27	1.00

*Exclude

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS3A – HE/AIR 2MIN 0 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Jan 30, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus	Fiber Tenacity
	Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.000	15.16	0.8	1.61290	0.59	415.43	2.17
2	7.00	50.000	19.07	0.5	1.00330	0.52	442.82	2.72
3	7.00	50.000	9.78	0.6	1.11125	0.13	415.68	1.40
4	7.00	50.000	11.25	0.4	0.86995	0.18	324.46	1.61
5	7.00	50.000	61.38	0.6	1.29540	2.44	1213.61	8.77
6	7.00	50.000	25.19	0.7	1.38430	0.34	819.64	3.60
7	7.00	50.000	23.97	0.4	0.84455	0.41	485.44	3.42
8	7.00	50.000	15.16	0.4	0.78105	0.32	503.34	2.17
9	7.00	50.000	22.99	0.7	1.45415	0.58	384.08	3.28
10	7.00	50.000	19.56	0.3	0.60325	0.24	640.20	2.79
11	7.00	50.000	39.62	1	1.93675	1.98	664.35	5.66
12	7.00	50.000	26.66	0.9	1.80975	0.59	517.16	3.81
13	7.00	50.000	25.92	0.6	1.27635	0.61	462.14	3.70
14	7.00	50.000	24.94	0.6	1.18745	0.70	392.98	3.56
15	7.00	50.000	18.59	0.3	0.59055	0.17	794.69	2.66
16	7.00	50.000	22.25	0.3	0.54610	0.27	756.48	3.18
17	7.00	50.000	25.92	0.5	1.00965	0.50	515.78	3.70
18	7.00	50.000	24.21	0.5	1.00330	0.41	550.17	3.46
19	7.00	50.000	21.28	0.5	1.06680	0.38	479.18	3.04
20	7.00	50.000	6.11	0.6	1.16205	0.10	188.22	0.87
Mean	7.00	50.000	22.95	0.6	1.12744	0.57	548.29	3.28
Stdv	0.00	0.000	11.61	0.2	0.38389	0.59	222.38	1.66

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS3A – HE/AIR 4MIN 0 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Jan 30, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus	Fiber Tenacity
	Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.000	12.72	0.5	0.94615	0.30	465.53	1.82
2	7.00	50.000	33.75	0.6	1.17966	1.17	627.38	4.82
3	7.00	50.000	31.55	0.9	1.70815	1.02	498.05	4.51
4	7.00	50.000	10.27	0.6	1.18601	0.11	406.08	1.47
5	7.00	50.000	17.61	0.3	0.66675	0.28	402.56	2.52
6	7.00	50.000	10.27	0.3	0.65405	0.07	719.45	1.47
7	7.00	50.000	14.67	0.8	1.68910	0.17	560.55	2.10
8	7.00	50.000	19.32	0.8	1.54305	0.41	487.65	2.76
9	7.00	50.000	22.74	0.7	1.39065	0.75	453.08	3.25
10	7.00	50.000	19.81	0.8	1.67005	0.64	333.00	2.83
11	7.00	50.000	19.81	0.6	1.21531	0.17	858.26	2.83
12	7.00	50.000	26.66	0.6	1.11125	0.33	799.46	3.81
13	7.00	50.000	19.07	0.4	0.74295	0.32	460.61	2.72
14	7.00	50.000	29.10	0.8	1.57336	1.00	357.13	4.16
15	7.00	50.000	25.43	0.6	1.21632	0.72	327.98	3.63
16	7.00	50.000	30.81	1.3	2.61476	2.33	346.68	4.40
17	7.00	50.000	32.77	1.1	2.13606	1.59	324.46	4.68
18	7.00	50.060	23.97	0.2	0.36786	0.21	933.04	3.42
19	7.00	50.057	19.81	0.4	0.79919	0.21	963.89	2.83
20	7.00	50.038	19.32	0.4	0.75364	0.19	724.94	2.76
21	7.00	50.063	19.81	0.2	0.34881	0.23	595.84	2.83
22	7.00	50.197	25.43	0.4	0.74636	0.50	510.91	3.63
Mean	7.00	50.019	22.03	0.6	1.19361	0.58	552.57	3.15
Stdv	0.00	0.045	6.90	0.3	0.56636	0.56	200.42	0.99

Em = 1.4 Gpa, young's modulus of matrix Vm= 0.4, poisson's ratio Ef= 90 Gpa, tensile modulus of Spectra R=bead's radius r=fiber radius															
Pmax/peak load A=cross-section area of fiber L=bead length															
$\tau_i = \frac{n P_{\max} \coth(nL/r)}{2.4}$															
Fiber Diameter	Fiber Radius, r	Bead Diameter D	Bead Radius, R	Bead Length L	Peak Load, g	n	ncoth(nL/r)	A, mm ²	Pmax/2A, kgf/mm ²	Pmax/2A, Mpa	IFSS(Shear Lag), Mpa	IFSS(imbedded length), Mpa	Sf=circumference of fiber		
1 1AA	0.03057918	0.015	0.220	0.110	0.344	3.67	0.075	0.080	0.000734	2.499854249	24.52357019	1.917617532	1.09106043		
2 1AB	0.03604698	0.018	0.205	0.103	0.342	9.54	0.080	0.088	0.00102	4.676391784	45.87540341	4.038487032	2.42021986		
3 1AC	0.03807209	0.019	0.242	0.121	0.368	7.09	0.078	0.086	0.0011378	3.11553802	30.66342797	2.618526921	1.580287255		
4 1AD	0.03260429	0.016	0.205	0.102	0.330	3.18	0.078	0.085	0.0008345	1.905363699	18.69161788	1.583824914	0.923113644		
5 1AE	0.03483192	0.017	0.227	0.113	0.347	3.18	0.077	0.085	0.0009524	1.669447419	16.37727918				
6 1AF	0.03361685	0.017	0.199	0.100	0.307	9.29	0.079	0.088	0.0008871	2.536029162	51.36544608	4.539117993	2.814080544		
7 1AG	0.03159174	0.016	0.181	0.091	0.283	5.62	0.080	0.089	0.000783	3.586655278	35.18508828	3.14822519	1.965929001		
8 1AH	0.03057918	0.015	0.190	0.095	0.282	8.56	0.078	0.087	0.000783	5.830722718	45.19938986	4.996368514	3.10463978		
9 1AI	0.03260429	0.019	0.292	0.146	0.406	8.31	0.071	0.075	0.0008345	4.97911097	48.84507692	3.679672248	1.950411058		
10 1AJ	0.03381936	0.019	0.233	0.107	0.352	9.05	0.076	0.085	0.0008871	3.049656323	49.40531173	2.373983805	1.895258013		
11 1AK	0.03483192	0.016	0.198	0.099	0.307	6.11	0.079	0.087	0.0008449	6.816877443	35.47175755	5.0730533	1.4247487		
12 1AL	0.03361685	0.016	0.253	0.137	0.379	5.89	0.073	0.078	0.000733	3.430492538	35.52685593	2.63375593	1.347407014		
13 1AM	0.03260429	0.016	0.252	0.126	0.367	5.14	0.074	0.079	0.0008345	3.070738809	30.21237771	2.393785111	1.347407014		
14 1AN	0.03057918	0.015	0.190	0.095	0.303	5.14	0.078	0.085	0.000734	3.501158267	34.3634628	2.935213808	1.734548747		
15 1AO	0.03483192	0.016	0.273	0.137	0.379	2.69	0.072	0.078	0.0008345	1.617169921	15.81146293				
16 1AP	0.03179425	0.016	0.199	0.100	0.302	5.87	0.078	0.086	0.0007935	3.698633639	36.283596	3.133734672	1.911585427		
17 1AQ	0.03584447	0.016	0.203	0.101	0.288	10.03	0.080	0.083	0.001008	4.972295369	48.77821757	4.552068555	3.035775144		
18 1AR	0.03381936	0.017	0.196	0.098	0.308	12.96	0.079	0.089	0.0008978	7.217297011	70.80168367	6.28385738	3.881773202		
19 1AS	0.03483192	0.017	0.200	0.100	0.307	11.98	0.080	0.090	0.0009524	6.289301911	61.69805174	5.547629176	3.502331165		
20 1AT	0.03361685	0.017	0.307	0.154	0.428	12.47	0.071	0.075	0.0008781	7.028305444	68.94802073	5.157936113	2.705761517		
21 1AU	0.03381936	0.017	0.189	0.094	0.304	10.27	0.080	0.090	0.0008978	5.719262369	56.10596384	5.041078405	3.123231987	3.123231987	
AVE	0.0333	0.017	0.222	0.111	0.335	7.406	0.077	0.085	0.001	4.224	41.438	3.759	2.1299406	2.1299406	
Fiber Diameter Fiber Radius, r Bead Diameter D Bead Radius, R Bead Length L Peak Load, g n ncoth(nL/r) A, mm ² Pmax/2A, kgf/mm ² Pmax/2A, Mpa IFSS(Shear Lag), Mpa IFSS(imbedded length), Mpa															
1 2AA	0.03057918	0.015	0.220	0.110	0.330	6.60	0.075	0.081	0.000734	4.495650694	44.1022333	3.580588904	2.042776778		
2 2AB	0.03057918	0.015	0.258	0.126	0.358	3.91	0.079	0.082	0.000862	2.66333356	26.1272914				
3 2AC	0.0295666	0.015	0.172	0.086	0.248	5.87	0.073	0.077	0.0008662	4.27695422	41.9569564	13.31638292	13.0343457		
4 2AD	0.0295666	0.015	0.195	0.095	0.276	4.16	0.077	0.088	0.0008662	3.04649276	29.55040754	1.501570329			
5 2AE	0.0230516	0.014	0.258	0.129	0.351	7.59	0.071	0.075	0.000631	6.006413932	56.922674243	4.434686008	2.370663770		
6 2AF	0.0230868	0.016	0.208	0.104	0.299	6.85	0.078	0.087	0.0008449	4.05806934	39.767184602	3.475005983	2.195342263		
7 2AG	0.03179425	0.016	0.237	0.119	0.356	8.31	0.074	0.080	0.0007935	5.236505458	51.36570404	4.103486129	2.296245881		
8 2AH	0.03260429	0.016	0.197	0.099	0.307	8.31	0.079	0.087	0.0008345	4.97911077	48.84507692	4.257435354	2.593868472		
9 2AI	0.03057918	0.015	0.155	0.077	0.257	2.20	0.083	0.094	0.000734	1.498850231	14.70077777				
10 2AK	0.03057918	0.015	0.203	0.101	0.285	7.34	0.077	0.086	0.000734	4.999708499	46.71452416	4.94714037	2.182717183		
11 2AL	0.03483192	0.016	0.216	0.108	0.340	10.03	0.080	0.091	0.0011868	4.2257227	41.4543396	3.762469967	2.373057012		
12 2AL	0.0328068	0.016	0.317	0.158	0.422	13.94	0.070	0.074	0.0008449	4.998465050	80.92901803	5.984082554	3.147023745		
13 2AM	0.0338194	0.017	0.149	0.074	0.233	4.89	0.087	0.094	0.0009978	2.723193085	26.71452416	4.782067819	1.938021518		
14 2AN	0.0295666	0.015	0.169	0.084	0.244	7.34	0.080	0.092	0.0006862	5.348018084	52.4640574	4.839879628	3.183604481		
15 2AO	0.03057918	0.015	0.416	0.208	0.490	11.49	0.065	0.067	0.000734	7.285151962	76.77815298	5.164820489	2.393373472		
16 2AP	0.03260429	0.016	0.279	0.140	0.406	19.07	0.072	0.076	0.0008345	11.426190468	11.426190286	5.622109286	4.500408855		
17 2AQ	0.0315917	0.016	0.209	0.104	0.315	11.00	0.077	0.084	0.000783	7.021043763	68.86761051	5.81020197	3.45001517		
18 2AR	0.0293641	0.015	0.235	0.118	0.352	11.98	0.073	0.078	0.0006769	8.849593708	6.816451428	6.736245824	3.619351515		
19 2AS	0.0305792	0.015	0.231	0.116	0.348	10.03	0.074	0.079	0.000734	6.630206736	67.02218228	5.320469127	2.94538694		
20 2AT	0.0295666	0.015	0.171	0.085	0.248	9.54	0.080	0.091	0.0006862	6.950696284	68.18897924	6.235091982	4.0668262376	4.0668262376	
AVE	0.0313	0.016	0.221	0.110	0.308	8.522	0.076	0.086	0.001	5.535	54.295	5.283	2.491088	3.354	2.533122
Fiber Fiber R Bead Diameter (Bead Radius, R Bead Length L Peak Load, g n ncoth(nL/r) A, mm ² x2/A, kgf/mm ² /2A, MpFSS(Shear Lag), Mpa	IFSS(imbedded length), Mpa														
1 3AA	0.0380721	0.019	0.233	0.117	0.368	8.8	0.078	0.086	0.0011378	3.87	37.93486124				
2 3AB	0.0409072	0.020	0.176	0.088	0.290	4.16	0.087	0.102	0.0012621	1.65	16.1718754				
3 3AC	0.0409072	0.020	0.271	0.136	0.396	11.98	0.076	0.084	0.0008871	4.75	46.558393105	3.918575963	2.360102772		
4 3AD	0.0378696	0.019	0.226	0.113	0.338	8.07	0.078	0.089	0.0011258	3.58	35.16105098	3.12993314	1.972140532		
5 3AE	0.0338194	0.017	0.292	0.146	0.401	19.32	0.072	0.077	0.0008978	10.76	105.5469549	4.453345472			
6 3AF	0.0328068	0.016	0.196	0.099	0.297	13.94	0.079	0.088	0.0008449	8.25	80.92901803	4.465429469			
7 3AG	0.03483192	0.017	0.225	0.112	0.331	12.96	0.077	0.086	0.0009524	6.80	66.74513778	5.70345802	3.510753424		
8 3AH	0.03483192	0.017	0.174	0.087	0.278	14.18	0.082	0.094	0.0008871	7.99	78.40247705	4.732678196			
9 3AI	0.03361685	0.017	0.204	0.149	0.374	12.72	0.088	0.103	0.0009524	6.68	65.50911671	6.731928716	4.543374223		
10 3AJ	0.0390846	0.020	0.195	0.098	0.288	12.23	0.083	0.099	0.0011992	5.10	50.02457011	4.944483866	3.394775679		
11 3AL	0.0348045	0.020	0.224	0.137	0.333	16.38	0.089	0.112	0.0023643	3.46	33.98183617	3.809635647	2.79741118		
12 3AM	0.03260429	0.016	0.231	0.115	0.329	23.97	0.075	0.088	0.0008345	7.33	71.88631658	6.302015114	4.19338467		
13 3AN	0.0338194	0.017	0.180	0.092	0.259	15.16	0.076	0.089	0.0008871	11.08	73.07215292	4.200088183	2.000088183		
14 3AP	0.03057918	0.020	0.244	0.156	0.345	19.55	0.074	0.081	0.0012602	4.954760344	5.539320877	3.243680978			
15 3AQ	0.0378696	0.019	0.224	0.112	0.3										

Appendix B. 3-Days Microbond Data

Report Date: Feb 15, 2002

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS1A – CONTROL 3 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Feb 4, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.044	9.05	0.5	1.03413	0.25	225.22	1.29
2	7.00	50.013	9.78	0.5	1.07778	0.07	487.67	1.40
3	7.00	50.165	4.65	0.2	0.40363	0.03	408.79	0.66
4	7.00	50.038	7.33	0.3	0.50127	0.07	346.01	1.05
5	7.00	50.079	5.38	0.3	0.67837	0.07	205.25	0.77
6	7.00	50.092	4.16	0.3	0.52966	0.01	220.43	0.59
7	7.00	50.492	8.07	0.4	-0.79230	0.01	*****	1.15
8	7.00	50.159	3.67	0.1	-0.17358	0.00	*****	0.52
9	7.00	50.111	7.82	0.4	0.77753	0.22	290.14	1.12
10	7.00	50.495	9.78	0	0.03831	0.10	329.18	1.40
11	7.00	50.337	9.78	0.2	-0.37032	0.03	*****	1.40
12	7.00	50.505	13.45	0.2	-0.48004	0.08	654.38	1.92
13	7.00	50.362	9.78	0.2	-0.41088	0.04	607.65	1.40
14	7.00	50.127	10.02	0.3	0.69382	0.07	485.80	1.43
15	7.00	50.352	7.09	0.1	0.25222	0.04	379.00	1.01
16	7.00	50.038	2.93	0.5	1.02792	0.07	35.51	0.42
17	7.00	50.124	8.56	0.3	0.50041	0.07	534.70	1.22
18	7.00	50.083	8.56	0.3	0.54520	0.13	295.64	1.22
19	7.00	50.194	10.51	0.2	0.37809	0.10	404.03	1.50
20	7.00	50.095	15.40	0.4	0.71475	0.29	296.75	2.20
Mean	7.00	50.195	8.29	0.2	0.34630	0.09	365.07	1.18
Stdv	0.00	0.165	3.13	0.3	0.54245	0.08	157.31	0.45

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS2A – HE/O2/AIR 30SEC 3 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Feb 4, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.394	10.76	0	0.04361	0.20	621.58	1.54
2	7.00	50.498	9.54	-0.3	-0.58099	0.06	517.22	1.36
3	7.00	50.137	8.31	0	0.02533	0.06	409.72	1.19
4	7.00	50.184	7.58	0.4	0.86043	0.13	243.94	1.08
5	7.00	50.117	14.18	0.5	1.03190	0.43	289.45	2.03
6	7.00	50.562	13.20	0.3	0.57285	0.21	386.27	1.89
7	7.00	50.267	6.36	-0.1	-0.23184	0.02	467.91	0.91
8	7.00	50.264	7.09	0.1	0.25123	0.05	456.78	1.01
9	7.00	50.530	11.25	0.3	0.50753	0.16	400.24	1.61
10	7.00	50.321	9.54	0	0.06264	0.14	289.37	1.36
11	7.00	50.098	12.96	0.3	0.51968	0.18	371.17	1.85
12	7.00	50.156	24.69	0.3	0.66468	0.64	509.32	3.53
13	7.00	50.197	16.14	0.4	0.74636	0.34	431.42	2.31
14	7.00	50.051	12.22	0.6	1.10234	0.41	403.78	1.75
15	7.00	50.079	17.60	0.8	1.52158	0.81	311.39	2.51
16	7.00	50.102	15.89	0.2	0.38656	0.17	697.07	2.27
17	7.00	50.133	14.91	0.4	0.85058	0.25	528.54	2.13
18	7.00	50.121	10.27	0.3	0.58769	0.08	565.89	1.47
19	7.00	50.048	28.85	0.7	1.37520	1.28	489.40	4.12
20	7.00	50.057	19.80	0.5	1.02609	0.62	376.07	2.83
21	7.00	50.048	21.76	0.5	0.98966	0.66	605.63	3.11
Mean	7.00	50.208	13.948	0.295	0.586	0.329	446.293	1.993
Stdv	0.00	0.164	5.945	0.273	0.528	0.317	118.225	0.850

*Exclude

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS3A – HE/O2/AIR 1MIN 3 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Feb 4, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.048	21.76	0.5	0.98966	0.66	605.63	3.11
2	7.00	50.073	13.20	0.6	1.26527	0.44	526.37	1.89
3	7.00	50.149	16.38	0.5	1.04953	0.47	551.69	2.34
4	7.00	50.054	17.11	0.5	1.01346	0.53	487.11	2.44
5	7.00	50.063	22.25	0.9	1.70163	0.57	394.70	3.18
6	7.00	50.022	10.02	0.3	0.57759	0.16	264.96	1.43
7	7.00	50.102	5.38	0.4	0.77313	0.11	254.39	0.77
8	7.00	50.203	5.62	0	0.01599	0.04	241.63	0.80
9	7.00	50.079	7.09	0.2	0.41844	0.08	255.79	1.01
10	7.00	50.044	15.65	0.2	0.40604	0.12	770.75	2.24
11	7.00	50.051	14.18	0.1	0.22693	0.08	*****	2.03
12	7.00	50.003	25.43	0.5	1.03210	0.41	613.21	3.63
13	7.00	50.003	28.36	0.8	1.52246	0.61	540.26	4.05
14	7.00	50.070	18.09	0.3	0.53900	0.31	663.81	2.58
15	7.00	50.063	22.98	0.3	0.65178	0.41	596.64	3.28
16	7.00	50.051	12.22	0.5	1.03890	0.14	456.60	1.75
17	7.00	50.048	19.80	0.6	1.19379	0.69	457.08	2.83
18	7.00	50.060	26.41	0.8	1.51438	1.05	568.48	3.77
Mean	7.00	50.066	16.774	0.444	0.885	0.382	485.241	2.396
Stdv	0.00	0.048	7.036	0.245	0.468	0.276	157.395	1.005

North Carolina State Univ.
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Sample ID: CHRIS3A – HE/AIR 2MIN 3 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Feb 4, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.000	14.18	0.5	0.98425	0.34	567.79	2.03
2	7.00	50.000	16.38	0.2	0.31115	0.17	835.26	2.34
3	7.00	50.000	21.27	0.4	0.86360	0.62	532.31	3.04
4	7.00	50.000	25.92	0.3	0.53340	0.41	609.52	3.70
5	7.00	50.057	27.87	0.6	1.27980	0.75	446.78	3.98
6	7.00	50.013	27.14	0.2	0.35493	0.24	1053.56	3.88
7	7.00	50.019	16.63	0.4	0.70314	0.35	779.53	2.38
8	7.00	50.025	13.69	0.2	0.41254	0.13	524.12	1.96
9	7.00	50.016	14.48	0.2	0.38723	0.14	491.27	2.03
10	7.00	50.003	14.91	0.3	0.53337	0.20	372.12	2.13
11	7.00	50.019	22.74	0.7	1.38911	0.47	634.92	3.25
12	7.00	50.029	31.05	0.7	1.33157	0.72	666.23	4.44
13	7.00	50.038	34.23	0.5	1.07202	0.90	536.71	4.89
14	7.00	50.000	42.05	0.7	1.36381	1.52	473.04	6.01
15	7.00	50.003	24.94	0.5	1.02719	0.56	433.72	3.56
16	7.00	50.270	21.03	0.1	0.15158	0.39	1017.56	3.00
17	7.00	50.006	22.98	0.7	1.35035	0.63	412.59	3.28
18	7.00	50.022	34.47	0.9	1.79625	1.71	355.03	4.92
19	7.00	50.000	30.32	1	2.08136	1.79	248.65	4.33
20	7.00	50.000	20.78	1.1	2.12437	1.25	196.86	2.97
Mean	7.00	50.026	23.853	0.510	1.003	0.665	559.379	3.406
Stdv	0.00	0.059	7.868	0.286	0.587	0.518	225.695	1.126

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Sample ID: CHRIS3A – HE/AIR 4MIN 3 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD-12

Test Date: Feb 4, 2002

Operator ID: GRAD

Sample Inputs:

		Type of Test	Straight					
		Grip Pressure						
		Lab Conditions	70F, 65% R					
	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus	Fiber Tenacity
	Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.057	21.26	0.8	1.62230	0.39	419.56	3.11
2	7.00	50.010	18.09	0.7	1.32546	0.41	283.41	2.58
3	7.00	50.029	19.07	0.7	1.39620	0.65	220.15	2.72
4	7.00	50.006	13.94	0.8	1.53261	0.56	157.18	1.99
5	7.00	50.016	9.29	1	2.00596	0.55	99.44	1.33
6	7.00	50.003	6.36	0.5	1.09848	0.23	125.02	0.91
7	7.00	50.000	27.63	0.4	0.71611	0.44	858.62	3.95
8	7.00	50.003	15.16	0.5	1.04768	0.38	465.46	2.17
9	7.00	50.013	58.68	1	2.00609	3.84	719.93	8.38
10	7.00	50.000	21.52	0.7	1.47955	1.19	409.48	3.07
11	7.00	50.000	15.65	0.9	1.75607	0.51	300.03	2.24
12	7.00	50.016	13.94	0.3	0.69193	0.22	579.18	1.99
13	7.00	50.003	14.67	0.4	0.83180	0.18	570.46	2.10
14	7.00	50.000	18.09	0.7	1.40970	0.50	456.14	2.58
15	7.00	50.000	15.89	0.5	1.02870	0.38	436.80	2.27
16	7.00	50.000	16.87	0.5	1.06680	0.47	376.35	2.41
17	7.00	50.000	22.98	1	1.99390	1.42	275.03	3.28
18	7.00	50.016	11.00	0.3	0.52918	0.09	550.23	1.57
19	7.00	50.013	34.47	0.4	0.78085	0.72	775.91	4.92
20*	7.00	50.035	15.16	0.4	0.87569	0.22	435.76	2.17
21	7.00	50.003	17.85	0.5	0.95244	0.22	514.21	2.55
22	7.00	50.013	13.94	0.3	0.51422	0.14	481.42	1.99
23	7.00	50.003	18.83	0.5	1.06529	0.50	359.24	2.69
Mean	7.00	50.009	19.326	0.609	1.221	0.636	428.784	2.764
Stdv	0.00	0.013	10.624	0.233	0.467	0.782	200.846	1.517

*Exclude

Em = 1.4 Gpa, young's modulus of matrix												Pmax/peak load											
Vm= 0.4, poison's ratio												A=cross-section area of fiber											
Ef= 90 Gpa, tensile modulus of Spectra												L=beam length											
r=Rebar's radius												$\tau_i = \frac{n P_{\max} \coth(L/r)}{2.4}$											
r=fiber radius												$F_i = \frac{P_{\max}}{S_f L}$											
n = $\left[\frac{E_u}{E_f (1 + V_m) \ln(R/r)} \right]^{1/2}$												Sf=circumference of fiber											
1 1BA	0.032604293	0.016	0.18727825	0.094	0.278452815	9.05	0.080	0.091	0.0008345	5.422497319	53.194696869	IFSS(imbedded length), Mpa											
2 1BA	0.037059538	0.019	0.287970838	0.144	0.389226407	9.78	0.074	0.081	0.0010781	4.535646448	44.49649166	3.114307814											
3 1BC	0.034831916	0.017	0.225799919	0.113	0.310247064	4.65	0.077	0.088	0.0009524	2.441173112	23.94790823	2.118243645											
4 1BD	0.031794249	0.016	0.183475091	0.092	0.268934791	7.33	0.080	0.091	0.0007935	4.61856367	45.30813606	1.344334274											
5 1BC	0.035844471	0.016	0.23835561	0.119	0.329080599	5.38	0.077	0.086	0.0010086	2.6677093627	26.16184849	2.678229428											
6 1BF	0.038072094	0.019	0.258404212	0.120	0.343863913	4.16	0.076	0.087	0.0011378	1.828016666	17.93284349	1.424941958											
7 1BG	0.034831916	0.017	0.333130822	0.167	0.444106926	8.07	0.070	0.074	0.0009524	4.236520501	41.56120584	3.083066143											
8 1BG	0.034831916	0.017	0.20878633	0.044	0.361687941	3.67	0.079	0.082	0.000824	1.828069916	18.3008219	3.568908108											
9 1BI	0.0368957027	0.018	0.196233293	0.098	0.273187525	7.82	0.082	0.087	0.0010864	3.698625050	35.8995913	3.508377707											
10 1BJ	0.036404293	0.016	0.263302493	0.118	0.329080594	3.78	0.077	0.087	0.0010030	7.94036991	47.0282501	4.084037178											
11 1BK	0.034831916	0.017	0.259416768	0.130	0.368165249	9.78	0.074	0.081	0.0009524	5.13438288	50.3678586	2.426414239											
12 1BL	0.039098465	0.020	0.2633669502	0.132	0.383961118	13.45	0.076	0.084	0.0011992	5.608620152	55.01475617	2.50371953											
13 1BM	0.0368957027	0.018	0.287970838	0.144	0.407047388	9.78	0.074	0.081	0.0010664	4.585625646	44.98498759	2.036633766											
14 1BN	0.037869583	0.019	0.2633669502	0.132	0.350141758	10.02	0.076	0.085	0.0011258	4.450276831	43.65721571	2.360873146											
15 1BO	0.036049465	0.020	0.189955448	0.095	0.275451548	7.09	0.084	0.101	0.0011992	2.956202201	20.99034359	2.057744968											
16 1BP	0.0362604293	0.016	0.216709384	0.108	0.431384724	2.93	0.077	0.079	0.0008345	1.755570995	17.22215107	1.629851313											
17 1BQ	0.034831916	0.017	0.301741598	0.151	0.421830701	8.56	0.072	0.076	0.0009524	4.49385846	44.0847515	1.820109759											
18 1BR	0.035844471	0.016	0.257391657	0.129	0.345889024	8.56	0.075	0.084	0.0010086	4.243554713	41.62926844	2.157019953											
19 1BS	0.032342828	0.021	0.322802754	0.161	0.438841636	10.51	0.074	0.081	0.0014062	3.739610208	36.656908914	1.767824096											
20 1BT	0.035844471	0.018	0.277440229	0.136	0.360672337	15.4	0.074	0.082	0.0010086	7.634345173	49.789377373	6.118135102											
AVE	0.036	0.018	0.250	0.125	0.353	8.289	0.076	0.085	0.001	4.037	39.601	3.676											
Fiber Diameter D												IFSS(IFSS Lag), Mpa											
1 2BA	0.030579182	0.015	0.21182665	0.106	0.322600243	10.76	0.077	0.083	0.0009734	7.329272949	7.901167673	5.970046611											
2 2BB	0.034831916	0.017	0.236127987	0.118	0.354540747	9.54	0.074	0.081	0.0009524	4.72538387	46.39523386	2.339587716											
3 2BC	0.0368957027	0.018	0.105799137	0.073	0.2629162	8.31	0.078	0.086	0.0010664	3.822330165	38.232231877	3.322190138											
4 2CD	0.0323428405	0.016	0.203002693	0.145	0.329080594	7.59	0.073	0.086	0.0012323	3.023276209	34.001417373	1.93001565											
5 2DE	0.035844471	0.018	0.204743096	0.142	0.369520944	14.18	0.078	0.083	0.0010066	7.029625955	68.96063002	3.116934152											
6 2EF	0.035844471	0.018	0.225799919	0.113	0.342851537	13.2	0.077	0.085	0.0010066	6.545798491	64.1946632	5.470058628											
7 2BG	0.032860804	0.016	0.218307007	0.109	0.310044552	6.36	0.075	0.084	0.0008449	3.763285684	7.92313769	3.104304643											
8 2BH	0.034831916	0.017	0.251319283	0.127	0.348116646	7.09	0.075	0.083	0.0009524	3.722132767	36.51412244	1.826788197											
9 2BI	0.030604983	0.018	0.265897124	0.133	0.360874748	11.25	0.085	0.091	0.0010102	5.146152953	44.95830537	2.701881381											
10 2BJ	0.0368957027	0.018	0.251109113	0.125	0.371202916	14.18	0.076	0.084	0.0010664	6.030921700	62.255260561	4.030014595											
11 2BK	0.0304097205	0.020	0.256614018	0.127	0.355407047	18.09	0.073	0.078	0.0009734	10.049105103	12.00691106	3.000629144											
12 2BL	0.034831916	0.017	0.256614018	0.127	0.360853737	22.74	0.075	0.081	0.0009524	12.736320538	11.79372538	1.964047167											
13 2BM	0.034831916	0.017	0.225799919	0.112	0.305944333	5.62	0.077	0.088	0.0009524	2.95	28.94349339	2.552479526											
14 2BN	0.030857027	0.018	0.21000405	0.105	0.305944333	7.09	0.080	0.092	0.0010664	3.32	31.6181616	3.000629144											

Appendix C. 15-Days Microbond Data

Report Date: Feb 15, 2002

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS1A – CONTROL 15 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD13

Test Date: Feb 15, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.044	11.00	0.4	0.76132	0.26	401.74	1.57
2	7.00	50.051	13.69	0.3	0.65339	0.30	734.15	1.96
3	7.00	50.124	8.80	0.4	0.88047	0.25	208.31	1.26
4	7.00	50.127	4.16	0.1	0.10768	0.01	551.45	0.59
5	7.00	50.102	9.78	0.1	0.17744	0.10	392.18	1.40
6	7.00	50.070	9.05	0.1	0.29803	0.08	528.79	1.29
7	7.00	50.041	9.78	0.5	0.95806	0.30	734.01	1.40
8	7.00	50.140	5.38	0.3	0.54458	0.09	154.44	0.77
9	7.00	50.054	6.60	0.5	0.95836	0.17	224.34	0.94
10	7.00	50.260	4.65	0.3	0.60644	0.04	-1105.83	0.66
11	7.00	50.146	10.51	0.2	0.41295	0.16	377.45	1.50
12	7.00	50.063	7.82	0.5	0.92592	0.21	360.84	1.12
13	7.00	50.283	4.89	-0.2	-0.48620	-0.01	*****	0.70
14	7.00	50.178	7.58	0.1	0.16941	0.03	579.61	1.08
15	7.00	50.362	8.56	-0.2	-0.32859	0.04	488.85	1.22
16	7.00	50.029	3.42	0	0.03808	0.01	*****	0.49
17	7.00	50.051	11.49	0.4	0.77882	0.20	512.64	1.64
18	7.00	50.130	30.56	0.6	1.13370	1.21	661.78	4.37
19	7.00	50.070	20.54	0.6	1.10188	0.63	637.79	2.93
20	7.00	50.063	15.89	0.3	0.55175	0.21	529.57	2.27
21	7.00	50.083	21.27	0.4	0.86218	0.48	436.97	3.04
Mean	7.00	50.118	10.734	0.271	0.529	0.227	389.952	1.533
Stdv	0.00	0.088	6.639	0.233	0.453	0.277	398.220	0.949

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Sample ID: CHRIS2A – HE/O2/AIR 30SEC 15 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD13

Test Date: Feb 15, 2002

Operator ID: GRAD

Sample Inputs:

	Name							
	Class/Project							
	Type of Test		Straight					
	Grip Pressure							
	Lab Conditions		70F, 65% R					
	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.133	6.60	-0.1	-0.10766	0.02	696.65	0.94
2	7.00	50.076	5.38	0.4	0.75100	0.09	267.10	0.77
3	7.00	50.076	13.45	0.4	0.73548	0.19	514.16	1.92
4	7.00	50.352	9.54	-0.2	-0.46031	0.05	461.61	1.36
5	7.00	50.273	3.42	0.1	0.13119	0.01	*****	0.49
6	7.00	50.187	31.30	0.4	0.71487	0.90	460.09	4.47
7	7.00	50.076	15.16	0.1	0.18645	0.10	688.61	2.17
8	7.00	50.079	8.80	0	0.05072	0.05	462.78	1.26
9	7.00	50.102	31.05	0.4	0.74144	0.63	664.23	4.44
10	7.00	50.168	7.09	0.5	0.95563	0.19	347.49	1.01
11	7.00	50.156	19.80	0.3	0.63936	0.34	461.94	2.83
12	7.00	50.032	23.23	0.4	0.87746	0.57	379.59	3.32
13	7.00	50.060	21.27	0.2	0.42494	0.18	1003.09	3.04
14	7.00	50.140	7.82	0.2	0.37361	0.09	324.46	1.12
15	7.00	50.073	10.27	0.1	0.10779	0.07	*****	1.47
16	7.00	50.568	8.80	0.1	0.26370	0.06	438.30	1.26
17	7.00	50.200	24.69	0.4	0.88569	0.80	392.71	3.53
18	7.00	50.054	18.58	0.3	0.61529	0.26	619.47	2.65
19	7.00	50.156	8.80	0.1	0.12661	0.02	965.58	1.26
20	7.00	50.038	19.80	0.3	0.63452	0.33	680.84	2.83
21	7.00	50.146	17.36	0.3	0.65283	0.37	474.42	2.48
22	7.00	50.025	9.78	0.2	0.32224	0.08	429.95	1.40
Mean	7.00	50.144	14.636	0.223	0.437	0.245	536.654	2.092
Stdv	0.00	0.124	8.148	0.182	0.368	0.261	197.482	1.164

North Carolina State Univ.
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Sample ID: CHRIS3A – HE/O2/AIR 1MIN 15 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD13

Test Date: Feb 15, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

Denier	Adj gage	Peak	Elong @	%Strn @ Pk	Energy to	Fiber	
	len1	Load	Pk Ld	Ld	Pk Ld	Modulus	
Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.016	10.76	0.1	0.15870	0.05	902.37
2	7.00	50.054	23.72	0.2	0.48208	0.42	1326.55
3	7.00	50.025	8.07	0.4	0.76161	0.13	308.72
4	7.00	50.038	11.25	0.4	0.88134	0.27	519.89
5	7.00	50.162	6.85	0.3	0.63151	0.07	312.70
6	7.00	50.010	15.89	0.4	0.86062	0.45	564.63
7	7.00	50.020	21.27	0.3	0.58394	0.36	621.30
8	7.00	50.210	22.25	0.5	1.01176	0.82	355.09
9	7.00	50.079	21.27	0.4	0.83337	0.53	341.05
10	7.00	50.019	14.43	0.6	1.11717	0.45	408.26
11	7.00	50.324	12.47	0.1	0.23344	0.07	771.10
12	7.00	50.044	21.52	0.3	0.53293	0.30	727.50
13	7.00	50.019	13.94	0.3	0.60547	0.19	481.48
14	7.00	50.035	6.60	0.4	0.82578	0.10	355.12
15	7.00	50.019	18.83	0.6	1.14256	0.55	362.37
16	7.00	50.022	12.71	0.1	0.28936	0.07	825.44
17	7.00	50.133	25.18	0.3	0.60165	0.40	693.84
18	7.00	50.070	19.56	1.6	3.15174	0.48	511.47
19	7.00	50.070	14.91	0.4	0.86874	0.28	449.35
20	7.00	50.054	12.22	0.6	1.17348	0.23	399.66
21	7.00	50.000	14.67	0.4	0.81280	0.31	396.55
Mean	7.00	50.068	15.637	0.414	0.836	0.311	554.021
Stdv	0.00	0.079	5.564	0.310	0.602	0.200	251.189
							0.795

North Carolina State Univ.
 College of Textiles
 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS3A – HE/AIR 2MIN 15 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD13

Test Date: Feb 15, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.022	17.60	0.4	0.86898	0.42	639.53	2.51
2	7.00	50.000	13.94	0.4	0.82550	0.35	565.77	1.99
3	7.00	50.016	22.49	0.4	0.71097	0.43	523.38	3.21
4	7.00	50.060	17.60	0.4	0.89427	0.39	926.20	2.51
5	7.00	50.067	20.54	0.5	0.93073	0.63	611.98	2.93
6	7.00	50.022	14.91	0.4	0.70395	0.28	639.05	2.13
7	7.00	50.181	19.80	0.6	1.15153	0.44	700.15	2.83
8	7.00	50.073	13.69	0.7	1.31887	0.60	525.23	1.96
9	7.00	50.079	19.32	0.5	1.08125	0.42	448.90	2.76
10	7.00	50.079	7.58	0.2	0.31556	0.07	422.37	1.08
11	7.00	50.079	34.96	0.2	0.44380	0.50	929.01	4.99
12	7.00	50.000	34.96	0.7	1.32080	1.06	718.43	4.99
13	7.00	50.060	21.76	0.3	0.60887	0.30	627.20	3.11
14	7.00	50.035	17.11	0.5	0.98991	0.34	456.07	2.44
15	7.00	50.044	30.07	0.6	1.10593	0.92	760.27	4.30
16	7.00	50.117	35.94	0.8	1.66614	1.83	962.34	5.13
17	7.00	50.022	29.10	0.4	0.79340	0.62	786.14	4.16
18	7.00	50.006	26.65	0.3	0.65397	0.46	662.64	3.81
19	7.00	50.003	27.14	0.5	0.95532	0.66	479.56	3.88
20	7.00	50.051	15.40	0.6	1.13807	0.43	616.68	2.20
Mean	7.00	50.051	22.028	0.470	0.924	0.558	650.045	3.146
Stdv	0.00	0.045	7.969	0.163	0.318	0.371	160.698	1.138

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Sample ID: CHRIS3A – HE/AIR 4MIN 15 DAYS

Method: Tensile microbond – 5lb

Test Date: Feb 15, 2002

Data Path: C:\GRAD13

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus	Tenacity
	Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.013	20.05	0.5	1.05383	0.57	758.88	2.86
2	7.00	50.054	24.45	0.2	0.45671	0.34	814.00	3.49
3	7.00	50.035	13.69	0.2	0.48671	0.15	642.17	1.96
4	7.00	50.102	19.80	0.2	0.46323	0.26	627.72	2.83
5	7.00	50.022	10.76	0.5	0.48162	0.22	409.19	1.54
6	7.00	50.041	22.00	0.9	0.96477	1.11	591.28	3.14
7	7.00	50.063	17.36	0.3	1.82095	0.31	402.47	2.48
8	7.00	50.054	25.18	0.2	0.68493	0.36	1365.60	3.60
9	7.00	50.038	23.72	0.1	0.49072	0.19	1081.28	3.39
10	7.00	50.060	17.60	0.2	0.25840	0.21	774.44	2.51
11	7.00	50.057	38.14	0.6	0.35054	1.17	532.92	5.45
12	7.00	50.073	28.61	0.6	1.14660	1.06	459.04	4.09
13	7.00	50.063	33.25	0.1	1.22520	0.37	2822.59	4.75
14	7.00	50.095	21.27	0.3	0.28539	0.43	927.46	3.04
15	7.00	50.073	24.94	0.2	0.50703	0.35	739.72	3.56
16	7.00	50.117	23.72	0.5	0.37410	0.50	492.27	3.39
17	7.00	50.063	37.65	0.7	0.93126	1.34	470.64	5.38
18	7.00	50.137	42.30	0.3	1.30500	0.66	957.56	6.04
19	7.00	50.067	30.56	0.2	0.65155	0.42	814.20	4.37
20	7.00	50.092	32.52	0.6	0.43609	0.92	654.39	4.65
21	7.00	50.038	31.05	0.5	1.25917	0.76	499.66	4.44
22	7.00	50.048	38.39	0.9	1.83000	1.87	353.94	5.48
Mean	7.00	50.064	26.228	0.400	0.794	0.617	781.428	3.747
Stdv	0.00	0.030	8.466	0.243	0.473	0.451	518.230	1.209

Em = 1 Gpa, young's modulus of matrix													
Vm= 0.4, poison's ratio													
Ef= 90 Gpa, tensile modulus of Spectra													
R=bead's radius													
r=fiber radius													
$n = \left[\frac{E_m}{E_f(1 + v_m) \ln(R/r)} \right]^{1/2}$													
Fiber Diameter, r													
1 15AA	0.030579182	0.015	0.197245849	0.099	0.295261239	11	0.077	0.085	0.000734	7.492751156	73.50388884	5.763092996	4.098308235
2 15AB	0.031794249	0.016	0.218307007	0.109	0.325030377	13.69	0.076	0.083	0.0007935	8.625945451	84.62051604	5.155022061	4.180669905
3 15AC	0.032604293	0.016	0.124341839	0.062	0.205751316	8.8	0.091	0.111	0.0008345	5.272704757	51.72523188	5.370785017	3.232554912
4 15AD	0.034831916	0.017	0.141352774	0.071	0.240583232	4.16	0.089	0.106	0.0009524	2.183931214	21.42436521	2.264387457	1.406762261
5 15AE	0.037869583	0.019	0.131834751	0.066	0.192939315	9.78	0.094	0.127	0.001258	4.343638374	42.6115339	3.807908714	1.92415214
6 15AF	0.038072094	0.019	0.212029162	0.106	0.288983394	9.05	0.080	0.096	0.0011378	3.976815103	39.01255616	3.735905775	2.569853034
7 15AG	0.036857027	0.018	0.2509113	0.125	0.355407047	9.78	0.076	0.085	0.001064	44.98498759	3.807908714	3.07353008	1.406762261
8 15AH	0.035844471	0.018	0.230052653	0.115	0.333333333	5.38	0.077	0.087	0.0010086	2.6677093627	26.16418849	2.264400089	1.309294411
9 15AI	0.034831916	0.017	0.212029162	0.106	0.297295135	6.6	0.074	0.087	0.00102	3.232530595	31.73770047	2.911206331	1.770985931
10 15AJ	0.03381916	0.017	0.227825054	0.114	0.31855002	4.65	0.077	0.087	0.0009524	2.441171112	23.947909823	2.077081413	1.309294411
11 15AK	0.034831916	0.017	0.214304932	0.101	0.287939308	10.51	0.070	0.085	0.001064	54.12769524	4.747475588	3.07353008	1.406762261
12 15AL	0.031591738	0.016	0.1883031	0.094	0.262261292	7.82	0.079	0.080	0.0007835	4.990840108	40.45651038	2.769853034	1.406762261
13 15AM	0.030579182	0.015	0.20496322	0.101	0.303007467	4.89	0.077	0.084	0.0004344	3.930983483	32.32851893	8.757748911	1.640482931
14 15AN	0.03381916	0.017	0.232602043	0.161	0.435763474	7.58	0.070	0.074	0.0008978	4.22.1227727	41.410244	3.069941407	1.615018858
15 15AO	0.035844471	0.018	0.242608343	0.121	0.331302822	8.56	0.076	0.086	0.0010086	4.243551174	41.62626644	3.576276133	2.251949262
16 15AP	0.034831916	0.017	0.258404212	0.129	0.356419603	3.42	0.074	0.082	0.0009524	1.79544345	17.61330025	1.442221416	0.860649898
17 15AQ	0.037869583	0.019	0.171931956	0.086	0.275415148	11.49	0.086	0.101	0.001258	5.103161755	14.32897095	50.06201682	3.441763656
18 15AR	0.036857027	0.017	0.387201296	0.194	0.526326448	30.56	0.069	0.071	0.001064	7.30987599	140.566587	1.92415214	1.309294411
19 15AS	0.032604293	0.016	0.320755132	0.160	0.445119482	20.54	0.070	0.073	0.0008345	12.30697181	120.7319395	1.309294411	1.309294411
20 15AT	0.034831916	0.017	0.30255164	0.151	0.431551235	15.89	0.072	0.076	0.0009524	8.341987259	9.973032463	6.212979732	3.302581403
21 15AU	0.036857027	0.016	0.32381513	0.162	0.452409883	21.27	0.072	0.076	0.001064	9.973032463	97.835448464	5.666	1.5267
AVE	0.035	0.017	0.232	0.116	0.332	10.734	0.078	0.089	0.001	5.666	55.580	3.881	1.5267
Fiber Diameter, r													
1 15BA	0.037869583	0.019	0.405832321	0.203	0.571688943	6.6	0.068	0.071	0.0001258	2.931320068	27.5624987	2.0323612	0.952429813
2 15BP	0.0340997205	0.020	0.26164430	0.131	0.355407047	5.38	0.077	0.088	0.0012621	2.131348236	10.2095625	1.834089637	1.179455324
3 15BC	0.035844471	0.018	0.244633455	0.122	0.353331936	13.45	0.076	0.084	0.0001086	6.667734096	65.41047121	5.496847585	3.317379199
4 15BD	0.036857027	0.018	0.254151478	0.127	0.374434094	9.54	0.076	0.083	0.0001258	4.47309444	43.88106151	3.648540642	2.15964121
5 15BE	0.03381916	0.019	0.202803058	0.101	0.300526523	3.42	0.076	0.081	0.0001258	1.51687653	14.93965024	1.442221416	0.860649898
6 15BF	0.035844471	0.018	0.164401377	0.091	0.230261236	3.13	0.076	0.083	0.0001258	1.07707352	14.24561635	1.442221416	0.860649898
7 15BG	0.036856028	0.015	0.240880599	0.124	0.329080599	15.16	0.072	0.078	0.0008962	11.0457703	102.35800098	8.480219337	4.86782
8 15BH	0.029566626	0.015	0.230982697	0.115	0.320575132	8.8	0.074	0.080	0.0008682	6.411792708	62.99986735	5.022539627	2.900617294
9 15BI	0.030579182	0.015	0.127784528	0.064	0.204536249	31.05	0.088	0.107	0.000734	21.14993904	20.47481317	1.340469273	1.251949262
10 15BJ	0.032806804	0.016	0.195018226	0.098	0.275415148	7.09	0.079	0.091	0.0008449	4.195831216	41.1617201	3.743245092	2.45151098
11 15BK	0.032604293	0.016	0.222559741	0.111	0.326852977	19.8	0.076	0.084	0.0008345	11.86358529	13.38617177	9.731949776	5.804667053
12 15BL	0.032806804	0.016	0.214054273	0.107	0.326042932	23.23	0.077	0.085	0.0008449	7.743473578	134.862345	1.340469273	1.340469273
13 15BM	0.037869583	0.018	0.163088821	0.086	0.251923856	7.82	0.087	0.105	0.001064	3.666625005	35.9659513	3.766803155	2.631216084
14 15BN	0.034831916	0.018	0.182462535	0.091	0.291080505	10.27	0.083	0.095	0.00102	5.034228892	49.3857543	4.694225642	3.058688172
15 15BP	0.038882139	0.019	0.185500203	0.094	0.292232572	8.8	0.084	0.099	0.0011868	3.707513436	36.3707068	3.595033108	2.419672802
17 15BQ	0.038882139	0.019	0.209801539	0.105	0.315512353	24.69	0.081	0.094	0.0011868	10.42010304	102.0446308	9.564737131	6.287737078
18 15BR	0.036857027	0.018	0.241393277	0.121	0.361864893	18.58	0.077	0.085	0.001064	8.711795729	7.246727082	4.354460924	3.124171875
19 15BS	0.037667072	0.019	0.289709838	0.144	0.397956987	8.8	0.074	0.081	0.0011138	3.950556852	38.75505074	1.825845856	1.825845856
20 15BT	0.034831916	0.017	0.381936006	0.191	0.498987444	19.8	0.068	0.071	0.0009524	3.94678261	10.39467261	1.01.9713783	7.323008674
22 15BV	0.031591738	0.016	0.301741956	0.151	0.386072337	17.36	0.073	0.079	0.0008735	6.010621508	8.02682153	4.612197106	4.612197106
AVE	0.035	0.017	0.243	0.124	0.348642932	14.636	0.077	0.087	0.001	7.891	7.7415	5.445	2.470242
Fiber Diameter, r													
1 15CA	0.034831916	0.017	0.210814095	0.105	0.31855002	10.76	0.079	0.088	0.0009524	5.65	55.14194464	4.874520421	3.029679109
2 15CB	0.03381916	0.017	0.235115431	0.118	0.356622114	23.72	0.076	0.082	0.0008978	13.21	129.584563	5.155022061	2.364040253
3 15CC	0.035191738	0.016	0.246861077	0.123	0.337586067	8.07	0.074	0.080	0.0007835	5.15	50.52378335	4.049514271	1.770985931
4 15CD	0.036857027	0.018	0.137100041	0.069	0.201498582	11.25	0.092	0.120	0.001064	5.27	51.7465348	6.229664735	4.732597655
5 15CE	0.038869583	0.019	0.256379101	0.129	0.360672337	6.85	0.078	0.081	0.001258	3.04	29.8455176	2.53873467	1.566849948
6 15CF	0.035844471	0.016	0.273187525	0.137	0.372251547	15.89	0.072	0.077	0.0007835	10.14	99.48239373	7.641719275	4.221777318
7 15CG	0.032806804	0.016	0.236330498	0.118	0.339611179	21.27	0.075	0.082	0.0008449	12.59	123.483516	1.309294411	1.309294411
8 15CH	0.032604293	0.016	0.263669502	0.132	0.349129202	22.25	0.073	0.080	0.0008345	13.33	130.7825465	1.309294411	1.309294411
10 15CJ	0.036857027	0.018	0.252126367	0.126	0.347104091	14.43	0.076	0.085	0.001064	6.77	66.37355531	5.656124432	3.52391688
11 15CK	0.036857027	0.018	0.245848522	0.123	0.367152693	12.47	0.077	0.084	0.001064	5.85	53.5581902	4.825287076	2.87890895
12 15CL	0.036857027	0.018											

Appendix D. 30-Days Microbond Data

Report Date: Mar 1, 2002

North Carolina State Univ.
College of Textiles
2401 Research Dr
Raleigh NC 27695-8301

Sample ID: CHRIS1A – CONTROL 30 DAYS

Method: Tensile microbond – 5lb

Test Date: Mar 1, 2002

Data Path: C:\GRAD13

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.000	5.38	0.3	0.64770	.02	359.82	0.77
2	7.00	50.000	7.09	0.5	0.91440	0.04	493.39	1.01
3	7.00	50.000	7.09	0.2	0.38735	0.04	598.17	1.01
4	7.00	50.000	7.34	0.3	0.57150	0.09	366.88	1.05
5	7.00	50.000	8.56	0.3	0.50165	0.05	453.20	1.22
6	7.00	50.060	9.54	0.3	0.62155	0.12	528.94	1.36
7	7.00	50.067	15.90	0.3	0.66197	0.25	332.71	2.27
8	7.00	50.019	20.79	0.3	0.56350	0.31	550.53	2.97
9	7.00	50.032	21.77	0.4	0.71506	0.43	479.29	3.11
10	7.00	50.000	8.81	0.5	0.93345	0.14	641.20	1.26
11	7.00	50.019	33.02	0.3	0.52050	0.43	945.78	4.72
12	7.00	50.000	17.12	0.2	0.41910	0.19	670.70	2.45
13	7.00	50.000	10.76	0.3	0.66040	0.10	550.32	1.54
14	7.00	50.102	35.47	0.7	1.30844	1.60	768.07	5.07
15	7.00	50.048	9.05	0.5	1.04041	0.08	367.23	1.29
16	7.00	50.041	14.43	0.3	0.56468	0.10	1078.60	2.06
17	7.00	50.063	12.72	0.1	0.20294	0.07	944.60	1.82
18	7.00	50.054	13.94	0.3	0.53917	0.21	472.21	1.99
19	7.00	50.076	13.45	0.2	0.38676	0.14	523.60	1.92
20	7.00	50.000	14.43	0.2	0.48260	0.19	428.03	2.06
21	7.00	50.013	10.52	0.3	0.69832	0.13	375.31	1.50
Mean	7.00	50.028	14.151	0.324	0.635	0.236	568.028	2.021
Stdv	0.00	0.032	8.006	0.134	0.248	0.341	209.545	1.145

North Carolina State Univ.
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 2401 Research Dr
 Raleigh NC 27695-8301

Sample ID: CHRIS2A – HE/O2/AIR 30SEC 30 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD13

Test Date: Mar 1, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus	Fiber Tenacity
	Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.000	12.48	0.3	0.62865	0.11	1048.23	1.78
2	7.00	50.000	14.19	0.3	0.59690	0.23	536.56	2.03
3	7.00	50.394	8.81	0.1	0.18901	0.04	739.54	1.26
4	7.00	50.000	14.19	0.3	0.52705	0.18	522.80	2.03
5	7.00	50.251	17.86	0.4	0.89088	0.44	355.55	2.55
6	7.00	50.010	28.62	0.3	0.58205	0.39	927.03	4.09
7	7.00	50.314	14.1/	-0.2	-0.40386	0.07	830.67	2.03
8	7.00	50.333	15.41	-0.2	-0.35324	0.12	751.84	2.20
9	7.00	50.025	16.88	0.3	0.57061	0.20	550.60	2.41
10	7.00	50.000	21.77	0.5	0.92075	0.48	497.91	3.11
11	7.00	50.013	15.90	0.3	0.50009	0.18	619.27	2.27
12	7.00	50.016	21.53	0.4	0.88237	0.37	660.59	3.08
13	7.00	50.006	21.04	0.4	0.88889	0.34	550.39	3.01
14	7.00	50.025	24.71	0.4	0.85047	0.55	479.55	3.53
15	7.00	50.022	22.75	0.3	0.61568	0.29	1234.02	3.25
16	7.00	50.010	22.99	0.3	0.67297	0.30	1081.87	3.28
17	7.00	50.000	21.77	0.2	0.38100	0.21	928.66	3.11
18	7.00	50.000	14.19	0.1	0.24130	0.10	678.73	2.03
19	7.00	50.124	8.56	0.1	0.27238	0.06	424.37	1.22
Mean	7.00	50.081	17.981	0.242	0.498	0.245	706.220	2.541
Stdv	0.00	0.133	5.506	0.189	0.381	0.151	244.049	0.775

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Sample ID: CHRIS3A – HE/O2/AIR 1MIN 30 DAYS

Method: Tensile microbond – 5lb

Data Path: C:\GRAD13

Test Date: Mar 1, 2002

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.010	7.34	0.3	0.57139	0.10	582.80	1.05
2	7.00	50.000	18.10	0.3	0.53975	0.19	817.14	2.59
3	7.00	50.000	15.90	0.3	0.55880	0.16	746.86	2.27
4	7.00	50.003	9.30	0.4	0.82545	0.08	609.32	1.33
5	7.00	50.010	13.94	0.4	0.83660	0.33	366.95	1.99
6	7.00	50.222	14.43	0.2	0.44253	0.17	583.47	2.06
7	7.00	50.013	17.12	0.3	0.56357	0.26	637.37	2.45
8	7.00	50.200	16.39	0.3	0.60574	0.24	491.13	2.34
9	7.00	50.010	14.43	0.3	0.60948	0.21	352.27	2.06
10	7.00	50.000	33.51	1.1	2.18296	1.78	363.42	4.79
11	7.00	50.000	30.09	0.6	1.11760	0.58	742.29	4.30
12	7.00	50.000	15.66	0.3	0.62865	0.18	761.98	2.24
13	7.00	50.000	17.12	0.5	1.09855	0.22	550.32	2.45
14	7.00	50.000	17.86	0.3	0.50165	0.25	489.17	2.55
15	7.00	50.003	10.27	0.3	0.60321	0.10	361.17	1.47
16	7.00	50.346	11.01	-0.3	-0.55496	0.02	1200.61	1.57
17	7.00	50.010	12.48	0.2	0.39363	0.07	963.24	1.78
18	7.00	50.038	9.30	0.1	0.26650	0.05	786.77	1.33
19	7.00	50.086	13.45	0.2	0.32330	0.11	620.17	1.92
20	7.00	50.013	25.20	0.5	1.00332	0.56	471.82	3.60
21	7.00	50.019	30.58	0.6	1.29862	1.01	550.53	4.37
Mean	7.00	50.047	16.832	0.343	0.686	0.317	621.371	2.405
Stdv	0.00	0.093	7.242	0.256	0.511	0.405	213.284	1.035

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Sample ID: CHRIS3A – HE/AIR 2MIN 30 DAYS

Method: Tensile microbond – 5lb
 2002

Test Date: Mar 1,

Data Path: C:\GRAD13

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus	Fiber Tenacity
	Denier	mm	Gm	mm	%	Gm-cm	Gm/Denier	Gm/Denier
1	7.00	50.184	19.81	0.2	0.41124	0.22	773.28	2.83
2	7.00	50.003	26.91	0.3	0.60321	0.41	733.80	3.84
3	7.00	50.000	19.57	0.4	0.83185	0.34	550.32	2.80
4	7.00	50.010	17.86	0.4	0.84439	0.30	450.35	2.55
5	7.00	50.000	20.06	0.3	0.60181	0.13	1034.60	2.87
6	7.00	50.000	28.38	0.5	0.90805	0.53	842.68	4.05
7	7.00	50.000	20.03	0.5	0.90497	0.21	724.81	2.90
8	7.00	50.000	35.96	0.6	1.17475	1.03	550.32	5.14
9	7.00	50.000	19.57	0.3	0.69850	0.32	450.26	2.80
10	7.00	50.000	27.40	0.5	1.03505	0.40	917.20	3.91
11	7.00	50.003	24.46	0.4	0.87153	0.30	738.28	3.49
12	7.00	50.006	22.02	0.4	0.84301	0.32	525.37	3.15
13	7.00	50.003	19.81	0.6	1.25578	0.33	434.49	2.83
14	7.00	50.000	23.48	0.5	0.92710	0.53	408.30	3.35
15	7.00	50.000	29.11	0.2	0.41910	0.31	1018.09	4.16
16	7.00	50.000	18.35	0.2	0.41275	0.15	790.20	2.62
17	7.00	50.000	30.82	0.5	0.94615	0.62	715.41	4.40
18	7.00	50.000	28.38	0.7	1.49860	0.73	521.35	4.05
19	7.00	50.000	31.80	0.8	1.62560	0.95	444.94	4.54
20	7.00	50.000	28.38	0.8	1.59385	0.78	398.10	4.05
21*	7.00	50.006	10.27	0.4	0.71991	0.04	835.78	1.47
Mean	7.00	50.010	24.608	0.455	0.920	0.446	651.108	3.517
Stdv	0.00	0.041	5.263	0.182	0.364	0.256	204.364	0.749

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Sample ID: CHRIS3A – HE/AIR 4MIN 30 DAYS

Method: Tensile microbond – 5lb
 2002

Test Date: Mar 1,

Data Path: C:\GRAD13

Operator ID: GRAD

Sample Inputs:

Name	
Class/Project	
Type of Test	Straight
Grip Pressure	
Lab Conditions	70F, 65% R

	Denier	Adj gage len1	Peak Load	Elong @ Pk Ld	%Strn @ Pk Ld	Energy to Pk Ld	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier
	Denier	mm	Gm	mm	%	Gm-cm		
1	7.00	50.000	15.41	0.4	0.71120	0.10	884.44	2.20
2	7.00	50.003	22.50	0.3	0.51432	0.29	639.60	3.21
3	7.00	50.003	23.24	0.5	1.00180	0.40	535.07	3.32
4	7.00	50.010	13.94	0.4	0.83804	0.20	366.95	1.99
5	7.00	50.013	37.43	0.5	0.96987	1.01	1206.77	5.35
6	7.00	50.010	19.08	0.2	0.48886	0.14	978.53	2.73
7	7.00	50.016	17.37	0.2	0.43801	0.15	754.96	2.48
8	7.00	50.000	28.62	0.6	1.14300	0.71	598.88	4.09
9	7.00	50.000	47.94	1.2	2.38125	2.84	440.25	6.85
10	7.00	50.000	30.58	0.6	1.19380	1.00	453.20	4.37
11	7.00	50.073	39.63	0.3	0.53119	0.53	1013.35	5.66
12	7.00	50.000	34.00	0.4	0.74930	0.53	928.66	4.86
13	7.00	50.006	29.84	0.3	0.60952	0.47	722.39	4.26
14	7.00	50.022	44.03	0.6	1.22501	1.40	566.76	6.29
15	7.00	50.010	10.76	0.2	0.41267	0.09	440.34	1.54
16	7.00	50.019	57	1.1	2.13914	3.26	448.58	8.14
17	7.00	50.032	27.15	0.3	0.69806	0.24	1292.87	3.88
18	7.00	50.000	36.45	0.5	0.91931	0.54	990.57	5.21
19	7.00	50.006	29.11	0.5	1.09452	0.47	770.54	4.16
20	7.00	50.003	27.15	0.8	1.56691	0.50	600.39	3.88
21	7.00	50.006	32.78	0.8	1.62396	0.98	495.35	4.68
22	7.00	50.006	41.10	0.9	1.78903	1.98	448.46	5.87
Mean	7.00	50.011	30.232	0.527	1.047	0.810	708.042	4.319
Stdv	0.00	0.016	11.596	0.281	0.552	0.861	267.771	1.656

Em = 1.4 Gpa, young's modulus of matrix											
Pmax=peak load											
Vm= 0.4, position's ratio											
Ef= 90 Gpa, tensile modulus of Spectra											
Re=bead's radius											
rf=fiber radius											
$\tau_f = \frac{n P_{max} \coth(\pi L/r)}{2.4}$											
Fiber Diameter Fiber Radius, r											
1 30AA	0.044957473	0.022	0.200690195	0.115	0.326254545	5.38	0.089	0.0015866	1.656428298	16.632126501	6.493765194
2 30AB	0.044957473	0.022	0.272782503	0.136	0.400769542	7.09	0.079	0.0015866	2.2342036789	21.9185496	1.943676974
3 30AC	0.044957405	0.022	0.246625209	0.155	0.448523171	7.09	0.078	0.0015866	2.020150945	23.153152026	1.230388077
4 30AD	0.043539895	0.022	0.247496811	0.124	0.388213852	7.34	0.076	0.0014980	2.469162325	24.19305241	1.209214981
5 30AE	0.046375051	0.023	0.406433984	0.203	0.551235318	8.56	0.076	0.0016883	2.535159678	24.86991644	1.091977229
6 30AF	0.042324828	0.021	0.269947347	0.135	0.390846497	9.54	0.077	0.0014062	3.392019352	33.27705984	2.889763523
7 30AG	0.040704739	0.020	0.340216712	0.170	0.476711121	15.9	0.072	0.0013096	6.112340342	59.96205876	4.640661056
8 30AH	0.04957473	0.022	0.286757572	0.143	0.413527744	20.79	0.077	0.0015868	6.55165559	64.27174134	5.588736377
9 30AI	0.043539895	0.022	0.253138923	0.127	0.376873228	21.77	0.079	0.0014881	7.31486921	71.75514318	4.493713658
10 30AJ	0.04090725	0.022	0.292628595	0.146	0.40927501	8.81	0.075	0.0013136	3.353253221	32.8961214	2.729089045
11 30AK	0.046375051	0.022	0.684892665	0.342	0.884568651	33.02	0.064	0.005	0.0016883	9.779319226	6.255160446
12 30AM	0.043539895	0.022	0.303766707	0.152	0.442891859	17.12	0.076	0.0014881	5.752138829	56.42848191	4.880005667
13 30AM	0.044957473	0.022	0.267112191	0.134	0.392246407	10.76	0.079	0.0015866	3.390508252	33.26425863	2.983198249
14 30AN	0.042122317	0.021	0.335965978	0.166	0.461320373	35.47	0.073	0.0013928	12.73318466	124.9125415	9.910357335
15 30AN	0.045159984	0.021	0.264277035	0.132	0.376873228	9.05	0.078	0.0016000	3.265450421	27.72747863	1.66126484
16 30AP	0.049210207	0.022	0.410692588	0.205	0.579384366	14.43	0.072	0.0017091	3.795389071	37.23276679	1.581188803
17 30AQ	0.049210207	0.025	0.36863776	0.184	0.507695423	12.72	0.074	0.0017091	3.346526363	32.82056781	1.590625843
18 30AQ	0.046577562	0.025	0.334795111	0.184	0.482351531	13.93	0.075	0.0016000	4.092567232	31.495074651	1.383832124
19 30AS	0.047792629	0.024	0.350141758	0.175	0.509113031	13.05	0.075	0.001793	3.750801514	36.79339006	2.36959465
20 30AT	0.047792629	0.024	0.362645443	0.163	0.468990684	14.43	0.076	0.001793	4.023878362	39.47424673	3.328099597
21 30AU	0.050627785	0.025	0.32199271	0.161	0.479548375	10.52	0.077	0.0020121	2.74169575	25.645284684	2.21020159
AVE	0.045	0.023	0.328	0.164	0.465	14.151	0.076	0.004	4.482	43.957	3.602
Fiber Diameter Fiber Radius, r											
1 30BA	0.046375051	0.023	0.350141758	0.175	0.509113001	12.48	0.074	0.0016883	3.696120652	35.26954934	2.9037425
2 30BB	0.046375051	0.023	0.244835966	0.122	0.385378696	14.19	0.082	0.0016883	4.202506281	41.22711616	3.84607754
3 30BC	0.044957473	0.022	0.251923856	0.126	0.374038072	8.81	0.080	0.0015868	2.776338901	23.515869517	1.636807359
4 30BD	0.046375051	0.023	0.364317573	0.182	0.542932361	14.19	0.073	0.0017093	4.202506261	41.22711616	3.227789586
5 30BE	0.047792629	0.024	0.265897124	0.133	0.420615634	17.86	0.080	0.001793	4.980351181	48.857249411	2.775712528
6 30BF	0.046375051	0.023	0.371405427	0.186	0.54272985	28.62	0.073	0.0016883	8.47619795	63.1515197	6.467203977
7 30BG	0.047792629	0.024	0.244633455	0.122	0.385378696	14.19	0.082	0.001793	3.95695315	38.87117044	3.683379369
8 30BH	0.049210207	0.025	0.237478076	0.119	0.385378696	15.41	0.084	0.0017091	5.051493824	39.76139544	3.586728243
9 30BI	0.045159984	0.023	0.248868169	0.124	0.392466586	16.88	0.081	0.0016000	5.217876587	51.71710931	4.710372189
10 30BJ	0.047792629	0.024	0.347036602	0.174	0.493722155	21.77	0.075	0.001793	6.070644224	55.53313161	4.881155384
11 30BK	0.043944917	0.022	0.339003645	0.170	0.507695423	15.9	0.074	0.001793	5.244212221	51.44571207	4.053731261
12 30CL	0.04090725	0.023	0.407814742	0.184	0.584757213	21.53	0.075	0.0017091	5.682333053	55.525422565	4.276319054
13 30CM	0.047792629	0.024	0.265691634	0.134	0.377072093	21.04	0.076	0.001793	5.040202003	51.50020205	2.226509677
14 30CN	0.042122317	0.021	0.330498177	0.165	0.469825854	17.86	0.073	0.001793	6.070644224	51.44791171	2.032932124
15 30CO	0.042122317	0.021	0.330498177	0.165	0.469825854	17.86	0.073	0.001793	6.070644224	51.44791171	2.032932124
16 30CP	0.047792629	0.024	0.287055449	0.141	0.436060048	10.27	0.075	0.001793	5.00160883	3.04	2.98817363
17 30CQ	0.047792629	0.024	0.204634985	0.203	0.572296476	11.01	0.072	0.001793	3.07	30.11860405	2.312103402
18 30CR	0.047792629	0.024	0.34872418	0.174	0.499392467	12.48	0.075	0.001793	3.48	34.13988090	2.077261269
19 30CR	0.046375051	0.023	0.277237748	0.139	0.427501013	9.3	0.078	0.0016883	2.75	27.01988568	2.376224462
20 30CU	0.047792629	0.024	0.295513973	0.150	0.433171324	25.2	0.078	0.0016883	0.001793	7.03	68.93631446
21 30CT	0.047792629	0.024	0.398136898	0.190	0.558323208	30.58	0.072	0.001793	8.53	83.65367048	28.21424096
AVE	0.047	0.024	0.323	0.161	0.466	16.832	0.077	0.005	4.767	50.566	4.204
Fiber Diameter D Bead Radius, R											
1 30CA	0.050627785	0.025	0.279667882	0.140	0.410629588	7.34	0.081	0.0016883	1.82	17.9811851	1.102883479
2 30CB	0.047792629	0.024	0.258606723	0.120	0.392246075	18.1	0.081	0.001793	5.05	49.517817412	3.016327417
3 30CC	0.045159984	0.023	0.337586067	0.160	0.500610753	15.9	0.074	0.0016000	4.97	47.155752	2.197279609
4 30CD	0.043539895	0.022	0.230457675	0.115	0.351761847	9.3	0.084	0.0014881	3.12	30.6532325	2.511595159
5 30CE	0.043742406	0.022	0.246051033	0.123	0.359862292	13.94	0.080	0.0015050	4.64	45.5225264	1.767072540
6 30CF	0.047792629	0.024	0.335164964	0.169	0.464152447	14.43	0.075	0.001793	4.02	43.7447673	1.203293247
7 30CG	0.046375051	0.023	0.302349129	0.151	0.444309437	17.12	0.077	0.0016883	5.07	4.250214113	2.226509677
8 30CH	0.049210207	0.025	0.326591694	0.133	0.399351594	14.43	0.081	0.0016883	3.04	37.240207	2.294076576
9 30CI	0.04090725	0.030	0.591610639	0.299	0.768492823	33.51	0.070	0.001793	5.62	57.1087137	2.109324129
10 30CJ	0.047792629	0.024	0.413527744	0.207	0.573714054	30.09	0.072	0.001793	8.39	82.1323424	1.26568444
11 30CK	0.047792629	0.024	0.306004804	0.180	0.499189055	17.86	0.073	0.0014881	6.000779988	58.8675673	4.587504409
12 30CL	0.046375051	0.023	0.267314702	0.134	0.400769542	15.66	0.080	0.0016883	4.64	45.49800134	4.116947119
13 30CM	0.044957473	0.022	0.271567436	0.136	0.413527744	17.12	0.079	0.0015868	5.40	52.02603224	4.212705443
14 30CN	0.042122317	0.021	0.330498177	0.165	0.469825854	17.86	0.073	0.001793	6.41	62.89647565	4.981709642
15 30CO	0.042122317	0.021	0.287055449	0.141	0.436060048	10.27	0.075	0.001793	3.04	29.838089	1.586837525
16 30CP	0.047792629	0.024	0.203422485	0.121	0.392466586	35.95	0.082	0.001793	10.02762757	9.3710265	5.989567145
17 30CQ	0.047792629	0.024	0.34872418	0.174	0.499392467	19.57	0.081	0.0016883	6.000721502	56.0014324	2.024051926
18 30CR	0.046375051	0.023	0.272737748	0.139	0.427501013	9.3	0.078	0.0016883	2.75	27.01988568	2.376224462
19 30CS	0.0449										

Appendix E. Compiled Microbond data for graphs

Data collected from Microbond tests to generate comparison curves between Control / 30sec He-air plasma / 1min He-air plasma / 2min He-O2-air plasma / 4min He-O2-air plasma treated UHMPE fibers over a thirty day time study.

0 DAYS				
		IFSS(Shear Lag), MPa	SD	SD/(n)^-2
Control	0 Days	3.76	1.30	0.298104
He/O2 30 sec.	0 Days	5.28	2.49	0.587155
He/O2 1.0 min.	0 Days	6.19	2.04	0.509988
He/Air 2.0 min.	0 Days	10.32	4.36	1.000716
He/Air 4.0 min.	0 Days	10.14	2.27	0.521082
3 DAYS				
Control	3 Days	3.68	0.96	0.233526
He/O2 30 sec.	3 Days	5.52	1.88	0.43111
He/O2 1.0 min.	3 Days	6.31	2.32	0.580511
He/Air 2.0 min.	3 Days	10.39	2.76	0.632572
He/Air 4.0 min.	3 Days	10.026	5.150	1.181449
15 DAYS				
Control	15 Days	3.88	1.53	0.370279
He/O2 30 sec.	15 Days	5.49	2.47	0.582242
He/O2 1.0 min.	15 Days	6.06	1.91	0.462583
He/Air 2.0 min.	15 Days	9.24	3.19	0.773527
He/Air 4.0 min.	15 Days	10.23	2.25	0.580771
30 DAYS				
Control		3.60	2.03	0.442747
He/O2 30 sec.	30 Days	4.20	1.04	0.23921
He/O2 1.0 min.	30 Days	3.94	1.12	0.244158
He/Air 2.0 min.	30 Days	6.22	1.60	0.358321
He/Air 4.0 min.	30 Days	7.37	2.62	0.616719
Control	0 Days	3.76	1.30	0.298104
Control	3 Days	3.68	0.96	0.233526
Control	15 Days	3.88	1.53	0.370279
Control	30 Days	3.60	2.03	0.442747
He/O2 30 sec.	0 Days	5.28	2.49	0.587155
He/O2 30 sec.	3 Days	5.52	1.88	0.43111
He/O2 30 sec.	15 Days	5.49	2.47	0.582242
He/O2 30 sec.	30 Days	4.20	1.04	0.23921
He/O2 1.0 min.	0 Days	6.19	2.04	0.509988
He/O2 1.0 min.	3 Days	6.31	2.32	0.580511
He/O2 1.0 min.	15 Days	6.06	1.91	0.462583
He/O2 1.0 min.	30 Days	3.94	1.12	0.244158
He/Air 2.0 min.	0 Days	10.32	4.36	1.000716
He/Air 2.0 min.	3 Days	10.39	2.76	0.632572
He/Air 2.0 min.	15 Days	9.24	3.19	0.773527
He/Air 2.0 min.	30 Days	6.22	1.60	0.358321
He/Air 4.0 min.	0 Days	10.14	2.27	0.521082
He/Air 4.0 min.	3 Days	10.026	5.150	1.181449
He/Air 4.0 min.	15 Days	10.23	2.25	0.580771
He/Air 4.0 min.	30 Days	7.37	2.62	0.616719

Appendix F. Random Order data for Epoxy bead placement

Each set of samples prepared for the Microbond test had five frames with four fibers hanging off each frame. Each fiber had five beads applied to it. Each of the five frames had a different fiber treatment:

1. Control, 2. He/O₂/air 30sec, 3. He/O₂/air 1min, 4. He/air 2min, 5. He/air 4min.

A random number generator was used (table below) to apply the beads to the five sets of fibers to ensure no bias during the Microbond testing.

Frame	Random Order						
4	-2.01468	1	-0.92649	2	-1.98165	2	-1.82127
1	-1.85763	4	-0.65821	4	-1.1789	5	-0.23193
3	-0.18745	5	-0.65389	1	0.09422	1	0.13823
5	0.87657	3	-0.4488	3	0.2498	4	0.2671
2	1.4978	2	0.47048	5	1.04751	3	1.384
2	-0.60411	4	-1.25992	1	-0.19002	5	-0.46324
4	-0.36742	1	-1.09858	3	-0.07863	1	0.33763
3	0.6108	2	0.10579	2	0.32124	3	0.70703
1	0.62671	3	0.81324	4	0.66857	4	0.96365
5	0.71128	5	0.83772	5	0.74887	2	1.4664
1	-2.40921	1	-1.83672	2	-0.34809	2	-1.32133
4	-2.21455	4	-0.31126	4	-0.24884	3	-0.96542
3	0.93066	3	-0.02517	3	-0.23031	4	-0.1672
5	1.31748	5	0.53554	5	0.46928	1	-0.12352
2	1.48191	2	1.21169	1	1.68994	5	0.41486
2	-2.39524	2	-1.68592	1	-1.25695	3	-1.09921
5	-1.90171	5	-1.5164	2	0.07928	1	-0.77541
1	0.72756	3	-0.25731	5	0.54662	4	-0.76795
3	1.24946	4	0.07416	4	1.47501	2	0.55889
4	1.46836	1	0.19021	3	1.53586	5	1.09178
3	-1.07659	3	0.12313	2	-1.27427	5	-2.40896
1	0.10829	2	0.43696	5	-1.12189	1	-0.79859
4	0.88234	4	0.7147	3	-0.39762	3	-0.42847
5	1.0191	1	0.76549	1	0.20849	4	-0.22523
2	1.4328	5	2.54009	4	0.85957	2	0.65959

Appendix G. Single fiber tensile data

Test Method: FiberTensNoSlackCp(3822)-5lb c

Sample I. D.	Specm No.	Denier Denier	Peak Load Gm	Elong @ Pk Ld mm	Break Load Gm	Elong @ Break mm	%Strn @ Break %	Fiber Modulus Gm/Denier	Fiber Tenacity Gm/Denier	Time @ Break Sec
CONTROL 30 Days	1	8	281.03	4.6	279.07	4.56	4.56	1076.56	35.12	5.5
	2	8	297.64	4.7	297.64	4.69	4.69	1193.8	37.21	5.6
	3	8	314.99	4.8	314.75	4.81	4.81	1192.99	39.37	5.8
	4	8	266.36	4.3	160.55	2.22	2.22	1104.97	33.3	2.7
	5	8	271.25	4.3	271.25	4.35	4.35	1078.39	33.91	5.2
	6	8	263.68	4.4	263.68	4.38	4.38	1081.7	32.96	5.3
	7	8	275.69	4.7	275.45	4.67	4.67	1066.11	34.46	5.6
	8	8	268.6	4.5	268.6	4.46	4.46	1069.72	33.58	5.4
	9	8	282.78	5	282.78	5.01	5.01	1076.56	35.35	6
	10	8	252.23	4.4	252.23	4.42	4.42	1026.74	31.53	5.3
	11*	8	162.78	28.6	162.78	28.62	28.62	114.5	20.35	34.3
	12	8	274.71	4.2	274.71	4.23	4.23	1138.19	34.34	5.1
	13	8	290.85	4.7	290.85	4.66	4.66	1111.46	36.36	5.6
	14	8	287.91	5.2	287.91	5.19	5.19	1045.72	35.99	6.2
	15	8	301.11	4.8	301.11	4.78	4.78	1123.48	37.64	5.7
	16	8	275.45	5	275.45	4.99	4.99	1053.67	34.43	6
	17	8	270.32	4.8	270.32	4.78	4.78	982.62	33.79	5.7
	18	8	276.43	4.5	276.43	4.53	4.53	1109.4	34.55	5.4
	19	8	279.11	4.6	279.11	4.62	4.62	1110.64	34.89	5.5
	20	8	271.29	4.7	271.29	4.75	4.75	1123.5	33.91	5.7
	21	8	291.58	5.3	291.58	5.33	5.33	1069.15	36.45	6.4
	22	8	290.6	4.7	290.36	4.73	4.73	1136.26	36.33	5.7
He/O ₂ /air 30sec 30 Days	1	8	252.47	4.6	250.76	4.62	4.62	972.64	31.56	5.6
	2	8	256.63	4.7	256.63	4.72	4.72	876.23	32.08	5.7
	3	8	251.5	4.5	251.5	4.53	4.53	893.5	31.44	5.4
	4	8	305.51	5	305.51	5	5	1000.93	38.19	6
	5	8	302.58	5.5	291.09	5.47	5.47	1045.72	37.82	6.6
	6	8	240.01	3.9	240.01	3.92	3.92	1003.18	30	4.7
	7	8	286.2	4.7	286.2	4.71	4.71	1075.12	35.78	5.7
	8	8	298.67	5.2	297.69	5.18	5.18	1077.13	37.33	6.2
	9	8	283.76	5.3	283.76	5.26	5.26	995.71	35.47	6.3
	10	8	271.05	4.8	271.05	4.83	4.83	986.17	33.88	5.8
	11	8	250.27	4.7	250.27	4.68	4.68	972.47	31.28	5.6
	12	8	282.78	4.6	282.78	4.62	4.62	1072.49	35.35	5.5
	13	8	276.18	4.6	275.94	4.58	4.58	1066.57	34.52	5.5
	14	8	229.25	4.7	229.25	4.68	4.68	907.39	28.66	5.6
	15	8	277.65	4.9	277.65	4.93	4.93	1056.57	34.71	5.9
	16	8	291.33	5	281.07	5.05	5.05	1038.45	36.42	6.1
	17	8	282.29	5.1	282.29	5.14	5.14	1013.37	35.29	6.2
	18	8	261.76	4.6	261.76	4.58	4.58	948.36	32.72	5.5
	19	8	253.21	4.7	253.21	4.74	4.74	986.05	31.65	5.7
	20	8	271.05	4.6	271.05	4.56	4.56	983.04	33.88	5.5
	21	8	253.45	4.5	253.45	4.51	4.51	990.4	31.68	5.4
	22	8	273.49	5	273.49	5	5	1036.25	34.19	6
He/O ₂ /air 1min 30 Days	1	8	306.73	5.2	306.73	5.22	5.22	1070.53	38.34	6.3
	2	8	283.02	4.4	281.8	4.45	4.45	1075.44	35.38	5.3
	3	8	280.82	4.7	280.82	4.66	4.66	1033.36	35.1	5.6
	4	8	280.82	4.6	272.03	4.65	4.65	1105.92	35.1	5.6
	5	8	269.09	4.6	268.85	4.58	4.58	977.21	33.64	5.5
	6	8	281.8	5	281.31	4.96	4.96	979.27	35.23	6
	7	8	278.87	4.8	278.87	4.77	4.77	976.96	34.86	5.7
	8	8	292.31	5	292.31	5	5	923.75	36.54	6

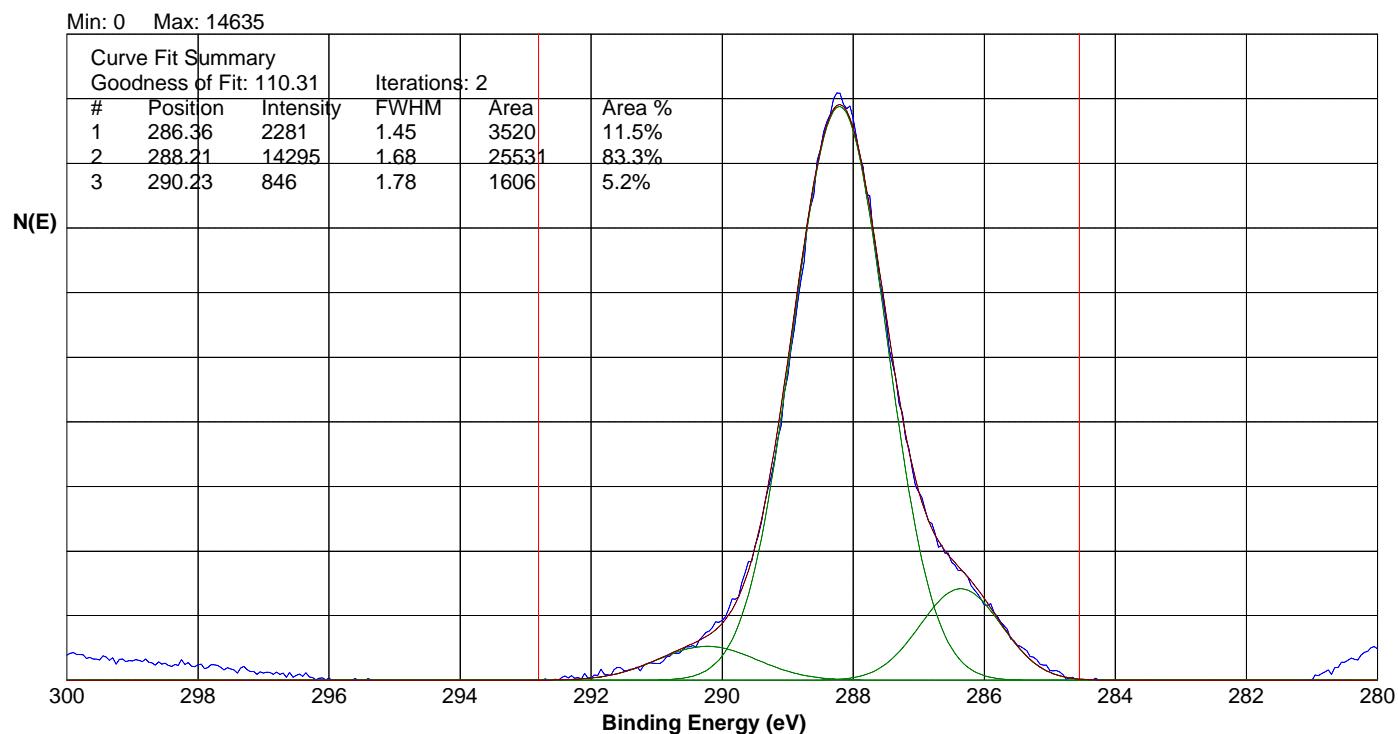
9*	8	270.8	5.7	270.8	5.76	5.76	865.84	33.85	6.9
10	8	280.58	4.9	280.34	4.88	4.88	951.7	35.07	5.9
11	8	297.44	4.9	296.71	4.88	4.88	1056.74	37.18	5.9
12	8	273.25	3.8	273.25	3.8	3.8	1070.24	34.16	4.6
13	8	274.71	4.4	273.49	4.38	4.38	988.24	34.34	5.3
14	8	283.51	5	283.51	5	5	932.82	35.44	6
15	8	262.98	4.4	262.74	4.42	4.42	951.78	32.87	5.3
16	8	272.76	5.2	272.76	5.18	5.18	874.38	34.09	6.2
17	8	284.98	4.5	284.98	4.5	4.5	1026.12	35.62	5.4
18	8	269.83	4.5	269.83	4.46	4.46	982.08	33.73	5.4
19	8	306	4.4	306	4.42	4.42	1218.34	38.25	5.3
20	8	271.05	4.8	271.05	4.77	4.77	860.28	33.88	5.7
21	8	283.51	4.7	283.51	4.75	4.75	1069.75	35.44	5.7
22	8	290.85	5	290.85	4.98	4.98	1046.1	36.36	6
23	8	296.71	5.2	296.22	5.2	5.2	969.64	37.09	6.2
He/air 2min	1	284.74	4.6	279.85	4.65	4.65	1015.69	35.59	5.6
30 Days	2	294.27	4.7	294.27	4.68	4.68	1166.35	36.78	5.6
	3	253.7	5.1	250.27	5.11	5.11	1013.69	31.71	6.1
	4	233.65	4.4	233.65	4.36	4.36	920.4	29.21	5.2
	5	241.23	4.5	241.23	4.47	4.47	903.56	30.15	5.4
	6	262.49	4.6	258.58	4.66	4.66	1038.54	32.81	5.6
	7	276.09	5.6	264.61	5.61	5.61	974.46	34.51	6.7
	8	270.96	5.1	270.96	5.08	5.08	1009.79	33.87	6.1
	9	256.3	4.8	254.83	4.84	4.84	1008.84	32.04	5.8
10*	8	296.61	5.5	296.61	5.49	5.49	1144.5	37.08	6.6
	11	231.87	4.2	231.13	4.19	4.19	1255.26	28.98	5
	12	284.4	4.4	284.4	4.42	4.42	1134.28	35.55	5.3
	13	285.86	4.7	285.62	4.66	4.66	1166.12	35.73	5.6
14*	8	261.43	5.3	261.43	5.31	5.31	1140.99	32.68	6.4
	15	213.05	3.7	213.05	3.73	3.73	924.63	26.63	4.5
	16	273.16	4.9	273.16	4.94	4.94	1230.08	34.14	5.9
	17	300.77	4.9	300.77	4.92	4.92	1140.99	37.6	5.9
18*	8	139.51	3.9	139.51	3.87	3.87	679.39	17.44	4.6
	19	274.62	4.8	274.62	4.78	4.78	1187.55	34.33	5.7
	20	243.35	4.6	242.62	4.65	4.65	915.67	30.42	5.6
He/air 4min	1	291.24	4.8	291.24	4.79	4.79	1147.09	36.4	5.8
30 Days	2	288.3	5.5	279.51	5.58	5.58	1036.93	36.04	6.7
	3	303.45	5	291.73	5.05	5.05	1065.9	37.93	6.1
	4	267.05	4.7	266.8	4.73	4.73	1023.64	33.38	5.7
	5	276.58	5.3	276.58	5.27	5.27	970.62	34.57	6.3
	6	258.99	4.2	258.99	4.19	4.19	1296.2	32.37	5
	7	279.75	4.6	279.26	4.57	4.57	1085.37	34.97	5.5
	8	246.04	4.3	246.04	4.35	4.35	1000.17	30.75	5.2
	9	285.13	4.8	284.88	4.77	4.77	1104.42	35.64	5.7
	10	277.8	4.8	277.8	4.83	4.83	1010.01	34.72	5.8
	11	284.15	5.5	284.15	5.47	5.47	1097.39	35.52	6.6
	12	299.79	4.8	299.54	4.84	4.84	1094.42	37.47	5.8
	13	274.87	4.6	274.62	4.63	4.63	1076.2	34.36	5.6
	14	285.37	5	285.37	5	5	998.49	35.67	6
	15	258.01	5	258.01	5	5	917.74	32.25	6
	16	272.67	4.7	267.29	4.67	4.67	1018.5	34.08	5.6
	17	255.08	4.8	254.34	4.84	4.84	987.68	31.88	5.8
	18	281.46	4.5	281.46	4.48	4.48	1145.4	35.18	5.4
	19	250.43	5	250.43	5	5	903.87	31.3	6
	20	247.75	4.1	247.75	4.12	4.12	1028.25	30.97	4.9
	21	292.21	4.6	291.97	4.63	4.63	1092.01	36.53	5.6
	22	246.77	4.2	246.77	4.23	4.23	1047.29	30.85	5.1
	23	259.96	4.7	259.96	4.68	4.68	1067.67	32.5	5.6

* Exclude due to slippage or error

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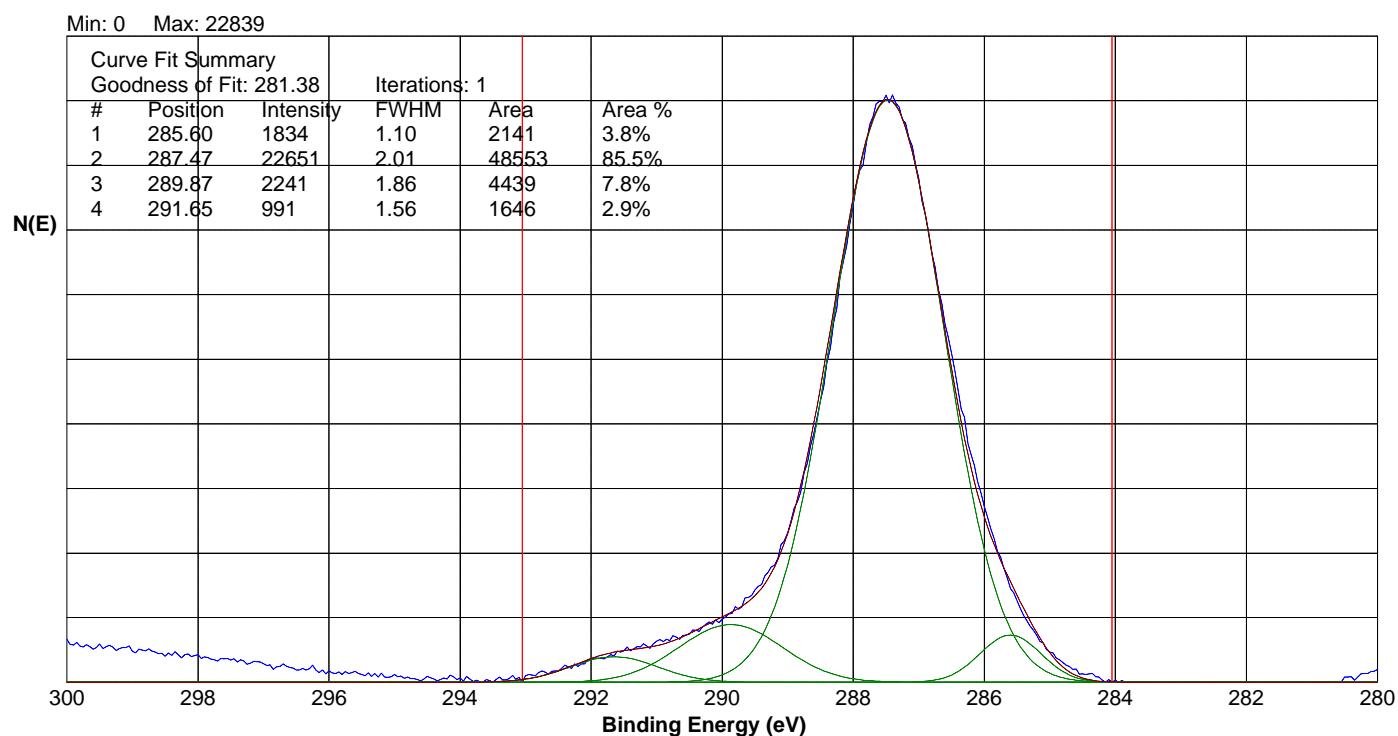
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XPS Multiplex

C 1s

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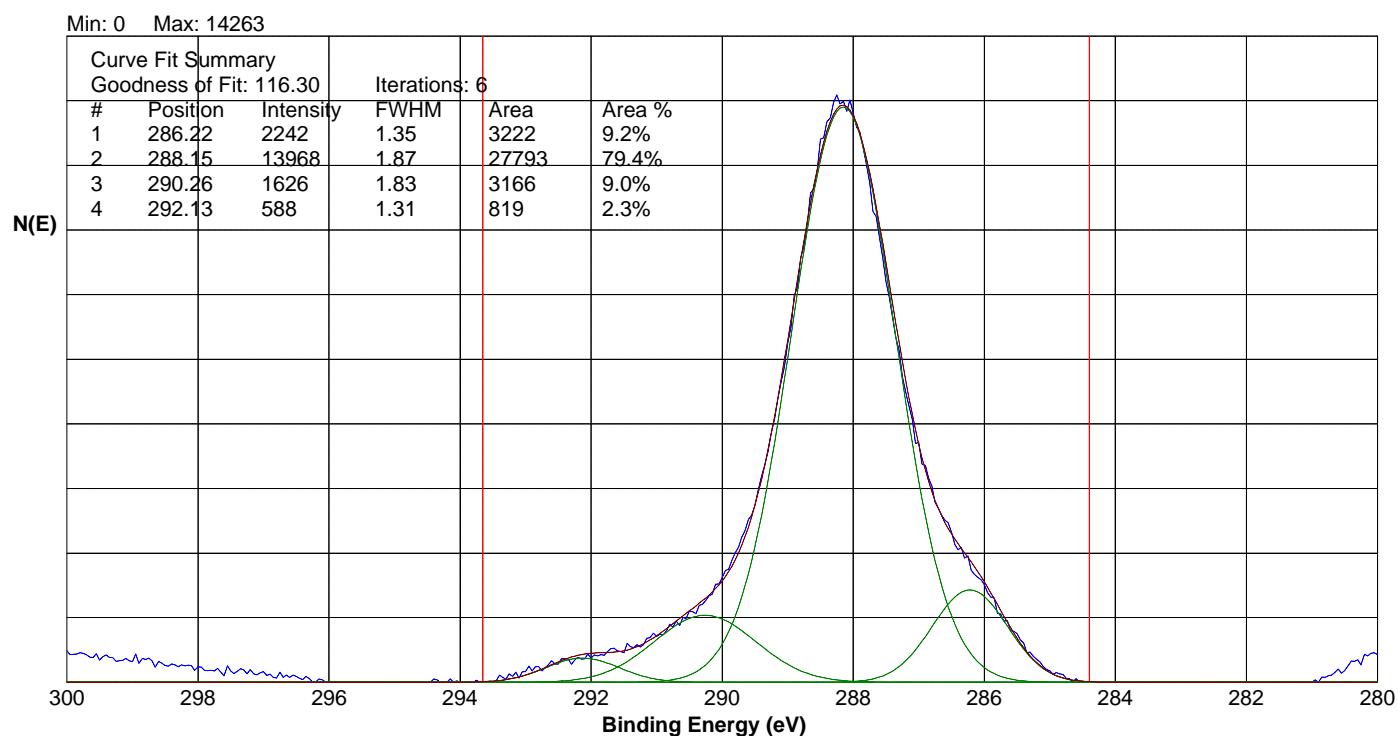
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XPS Multiplex

C 1s

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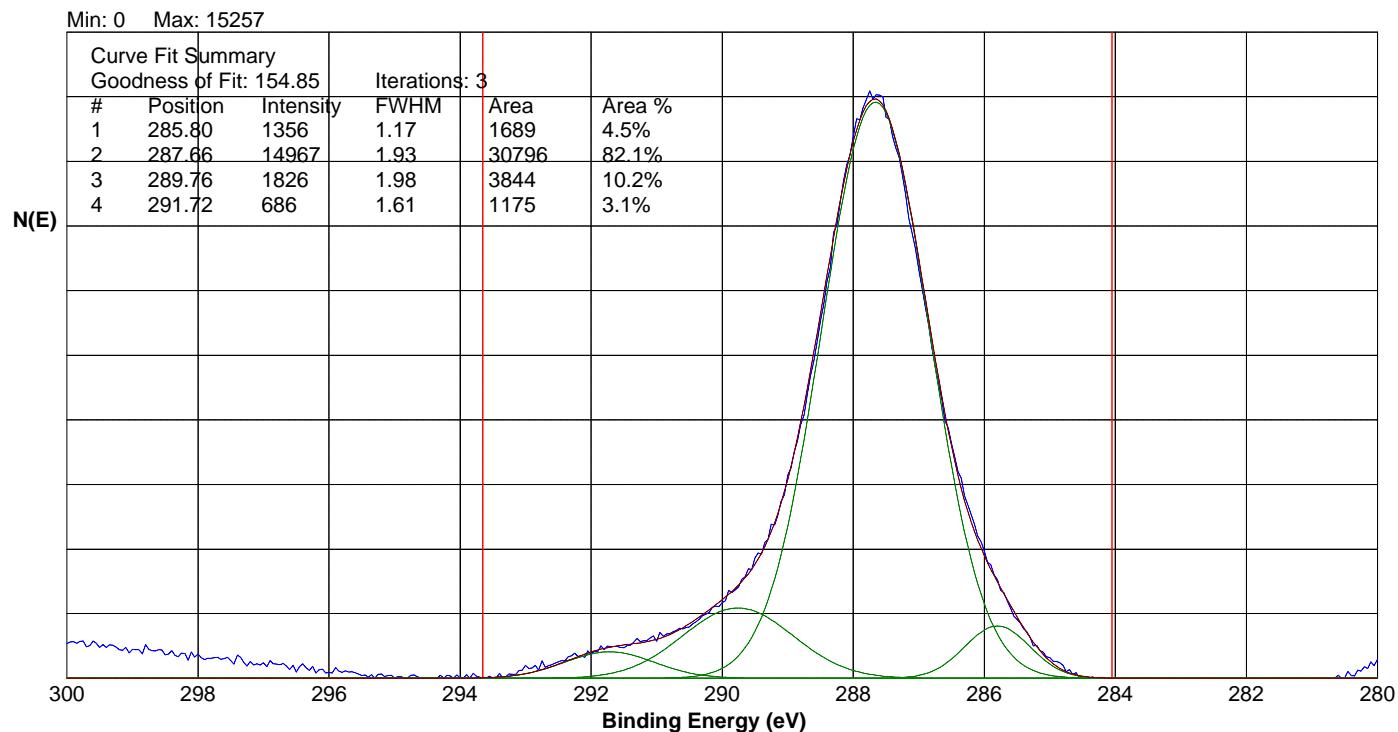
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C 1s

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Source: Mg, Pass Energy: 35.75 eV, Work Function: 4 eV



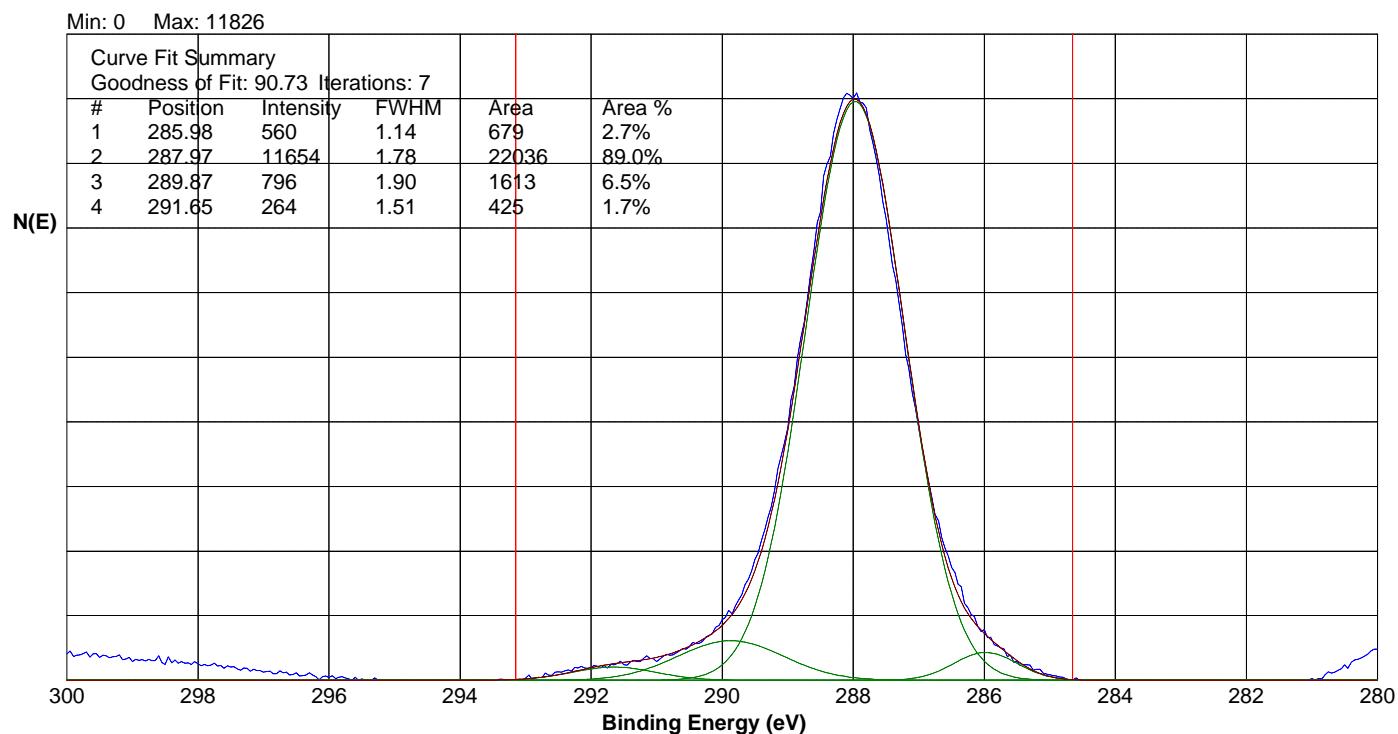
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C 1s

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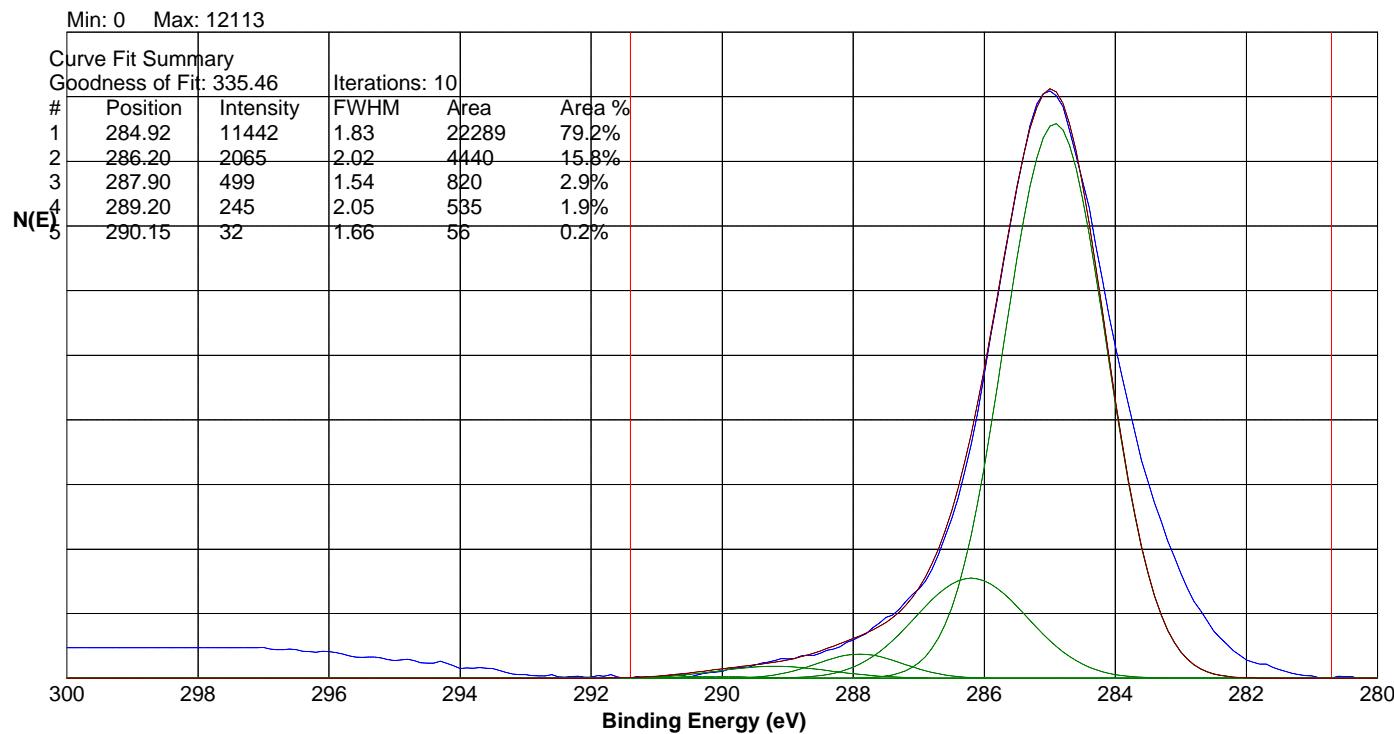
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XPS Multiplex

C 1s

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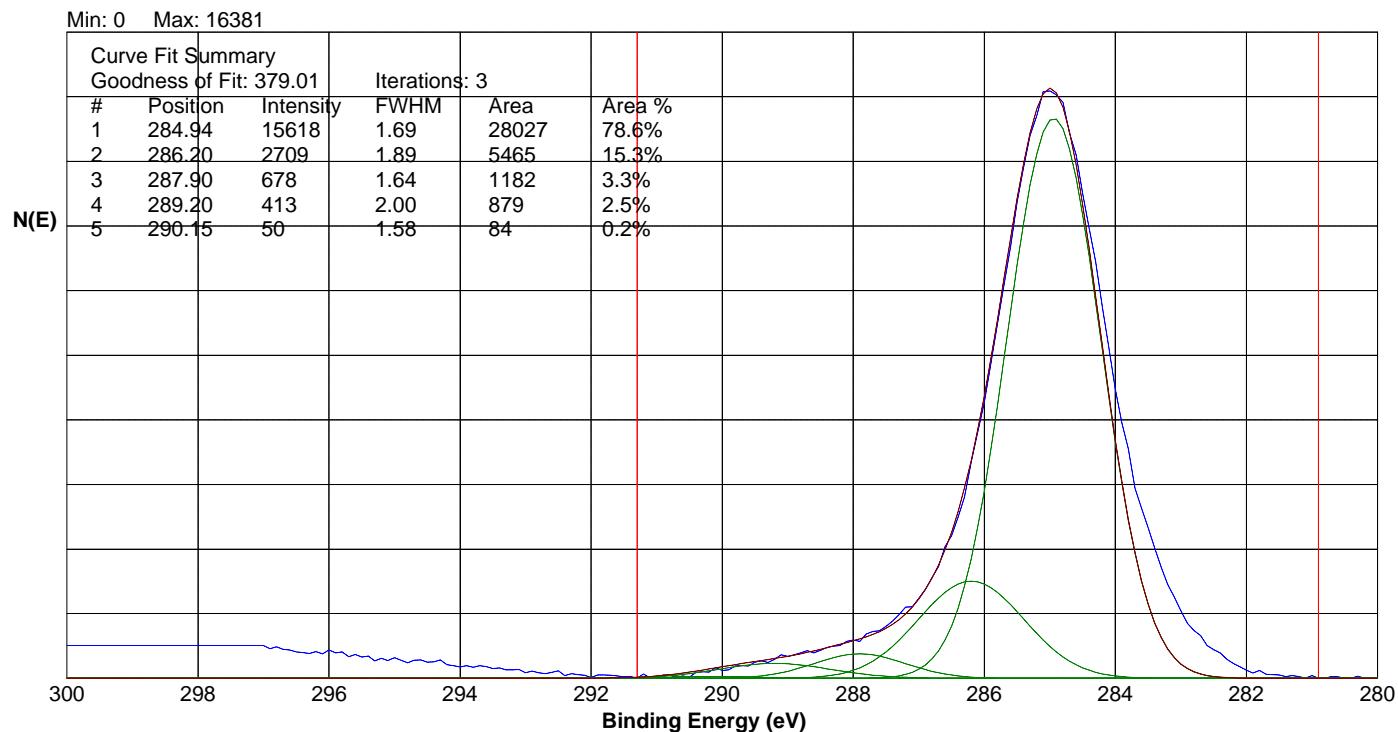


AugerScan 2
Fibers 1, Sample #: 5, Angle: 45

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Acquired: Monday, June 24, 2002 11:12:01

XPS Multiplex
C 1s

EV/Step: 0.1 eV, Time/Step: 100 mSec, Sweeps: 15
Source: Mg, Pass Energy: 35.75 eV, Work Function: 4 eV



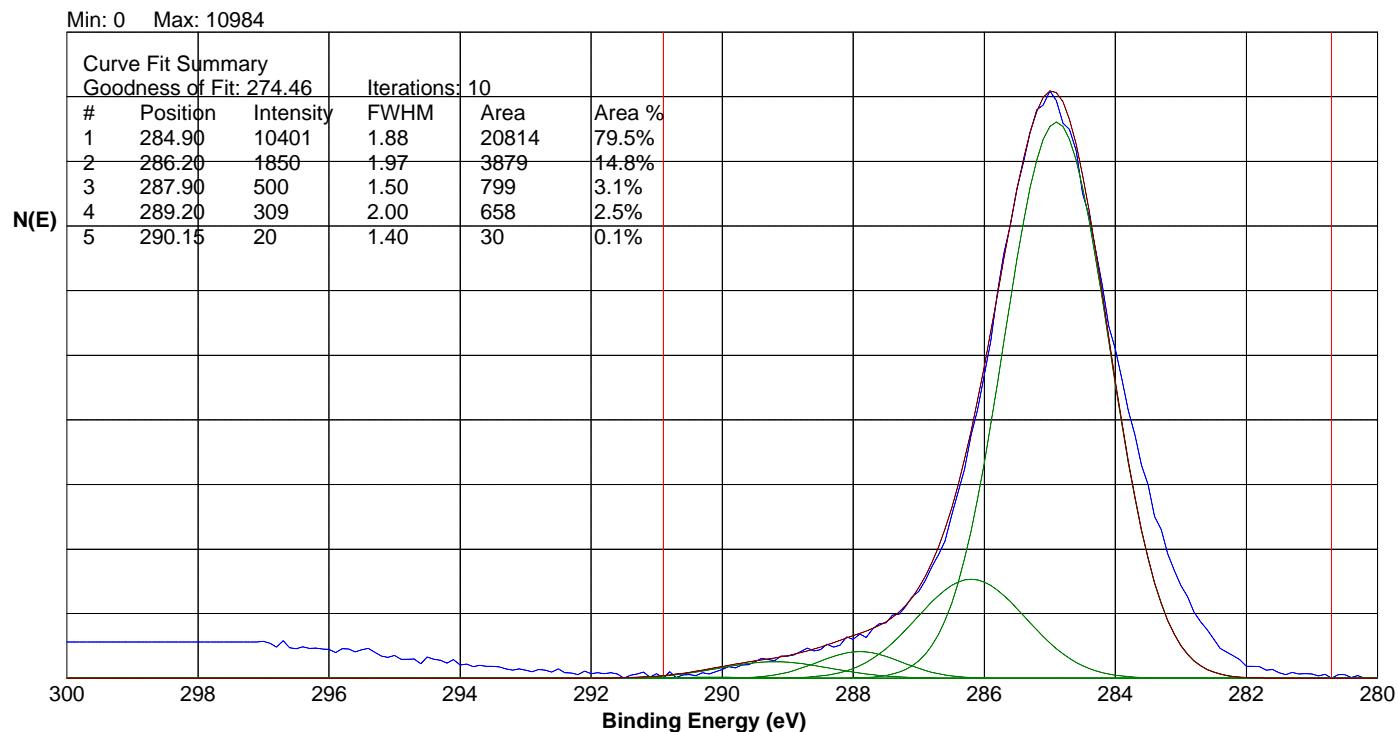
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Fibers 1, Sample #: 5, Angle: 45

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XPS Multiplex

C 1s

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Source: Mg, Pass Energy: 35.75 eV, Work Function: 4 eV



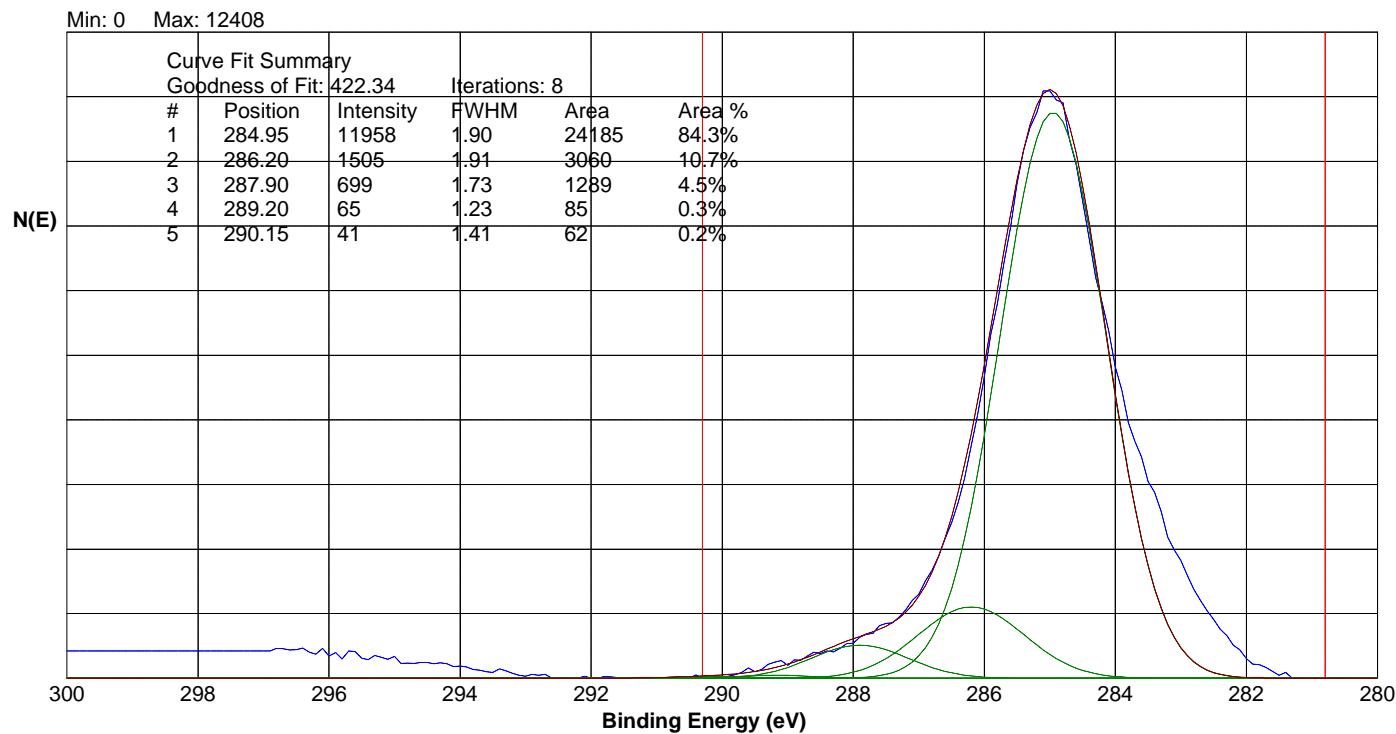
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C 1s

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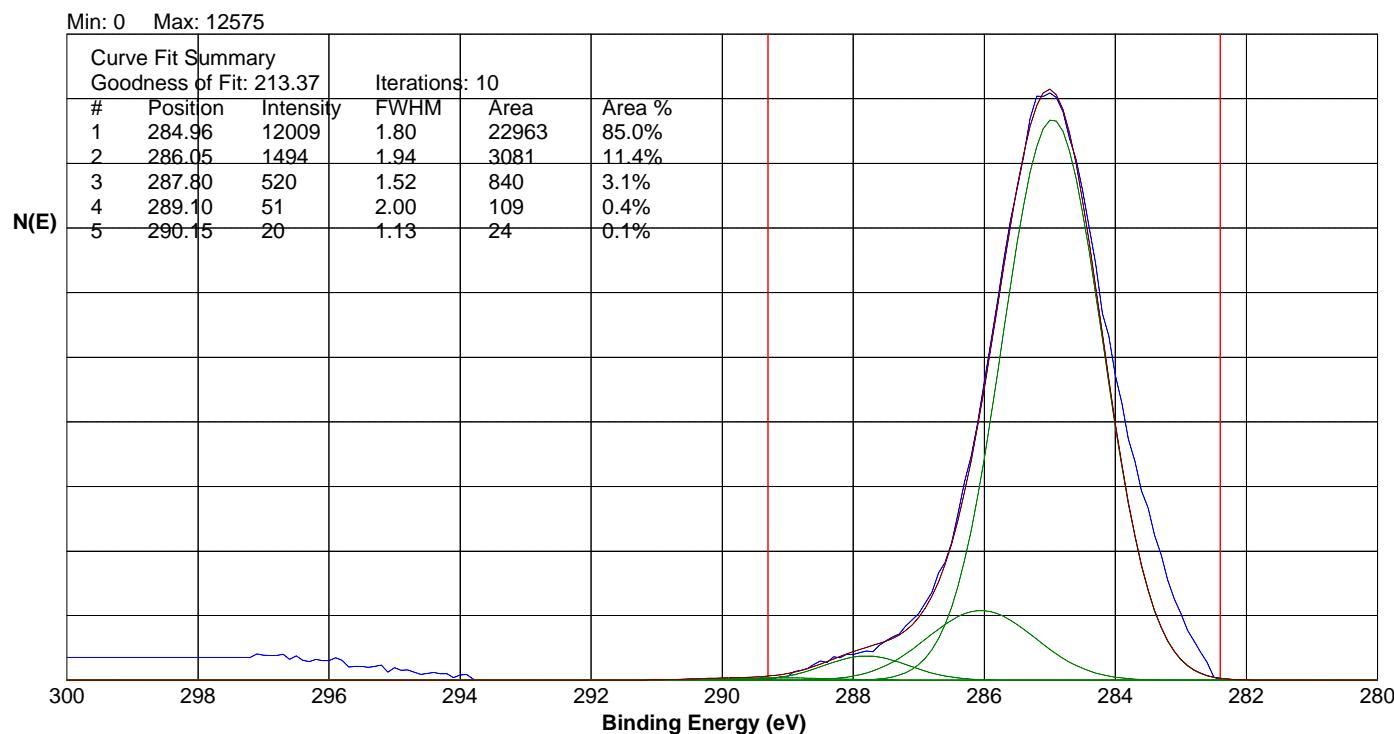


AugerScan 2
Fibers 2, Sample #: 4, Angle: 45

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XPS Multiplex
C 1s

EV/Step: 0.1 eV, Time/Step: 100 mSec, Sweeps: 15
Source: Mg, Pass Energy: 35.75 eV, Work Function: 4 eV



Appendix I. Minitab Statistical Analysis

5/13/02 11:33:36 AM

One-way ANOVA: Time (Days) versus Control

Analysis of Variance for Time (Da

Source	DF	SS	MS	F	P
Control	3	0.80	0.27	0.11	0.952
Error	70	164.85	2.35		
Total	73	165.65			

Individual 95% CIs For Mean
Based on Pooled StDev

Level	N	Mean	StDev				
1	19	3.759	1.299	(-----*	-----)		
2	17	3.676	0.963	(-----*	-----)		
3	17	3.881	1.527	(-----*	-----)		
4	21	3.602	2.029	(-----*	-----)		

Pooled StDev = 1.535 3.00 3.50 4.00 4.50

Tukey's pairwise comparisons

Family error rate = 0.0500
Individual error rate = 0.0105

Critical value = 3.72

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-1.265		
	1.431		
3	-1.470	-1.589	
	1.226	1.180	
4	-1.121	-1.242	-1.037
	1.436	1.392	1.597

Fisher's pairwise comparisons

Family error rate = 0.200
Individual error rate = 0.0500
Critical value = 1.994

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-0.939		
	1.104		
3	-1.144	-1.254	
	0.900	0.845	
4	-0.811	-0.924	-0.719
	1.126	1.073	1.278

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Welcome to Minitab, press F1 for help.

One-way ANOVA: Time (Days) versus O2 30sec

Analysis of Variance for Time (Da
Source DF SS MS F P
O2 30sec 3 21.83 7.28 1.74 0.166
Error 70 292.36 4.18
Total 73 314.20
Individual 95% CIs For Mean
Based on Pooled StDev
Level N Mean StDev -----+-----+-----
1 18 5.283 2.491 (- - * - - -)
2 19 5.519 1.879 (- - * - - -)
3 18 5.485 2.470 (- - * - - -)
4 19 4.204 1.043 (- - * - - -)
-----+-----+-----+-----
Pooled StDev = 2.044 4.0 5.0 6.0

Tukey's pairwise comparisons

Family error rate = 0.0500
Individual error rate = 0.0105

Critical value = 3.72

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-2.005		
	1.531		
3	-1.995	-1.734	
	1.589	1.802	
4	-0.690	-0.429	-0.487
	2.846	3.059	3.049

Fisher's pairwise comparisons

Family error rate = 0.200
Individual error rate = 0.0500

Critical value = 1.994

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-1.577		
	1.104		
3	-1.561	-1.306	
	1.156	1.374	
4	-0.262	-0.007	-0.060
	2.419	2.637	2.621

Welcome to Minitab, press F1 for help.

One-way ANOVA: Time (Days) versus O2 1min

Analysis of Variance for Time (Da
Source DF SS MS F P
O2 1min 3 74.27 24.76 7.08 0.000
Error 66 230.82 3.50
Total 69 305.09
Individual 95% CIs For Mean
Based on Pooled StDev
Level N Mean StDev -----+---+---+---+---
1 16 6.193 2.040 (------*-----)
2 16 6.308 2.322 (------*-----)
3 17 6.059 1.820 (------*-----)
4 21 3.944 1.314 (------*-----)
-----+---+---+---+---
Pooled StDev = 1.870 3.6 4.8 6.0 7.2

Tukey's pairwise comparisons

Family error rate = 0.0500
Individual error rate = 0.0104

Critical value = 3.73

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-1.859		
	1.629		
3	-1.584	-1.469	
	1.852	1.967	
4	0.612	0.727	0.505
	3.886	4.001	3.724

Fisher's pairwise comparisons

Family error rate = 0.200
Individual error rate = 0.0500

Critical value = 1.997

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-1.435		
	1.206		
3	-1.166	-1.052	
	1.435	1.550	
4	1.010	1.124	0.896
	3.488	3.603	3.333

Welcome to Minitab, press F1 for help.

One-way ANOVA: TIme (Days) versus He 2min

Analysis of Variance for TIme (Da

Source	DF	SS	MS	F	P
He 2min	3	225.38	75.13	7.72	0.000
Error	71	690.88	9.73		
Total	74	916.26			

Individual 95% CIs For Mean
Based on Pooled StDev

Level	N	Mean	StDev				
1	19	10.317	4.362	(-----*	-----)		
2	19	10.392	2.757	(-----*	-----)		
3	17	9.235	3.189	(-----*	-----)		
4	20	6.219	1.602	(-----*	-----)		

Pooled StDev = 3.119 6.0 8.0 10.0 12.0

Tukey's pairwise comparisons

Family error rate = 0.0500
Individual error rate = 0.0105

Critical value = 3.72

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-2.737		
	2.587		
3	-1.657	-1.582	
	3.822	3.896	
4	1.469	1.544	0.309
	6.727	6.802	5.723

Fisher's pairwise comparisons

Family error rate = 0.200
Individual error rate = 0.0500

Critical value = 1.994

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-2.093		
	1.943		
3	-0.994	-0.919	
	3.159	3.234	
4	2.105	2.180	0.964
	6.091	6.166	5.068

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One-way ANOVA: Time (Days) versus He 4min

Analysis of Variance for Time (Da
Source DF SS MS F P
He 4min 3 102.6 34.2 3.02 0.036
Error 67 757.4 11.3
Total 70 860.0
Individual 95% CIs For Mean
Based on Pooled StDev
Level N Mean StDev -----+-----+-----+-----
1 19 10.141 2.271 (- - * - - -)
2 19 10.026 5.150 (- - * - - -)
3 15 10.228 2.249 (- - * - - -)
4 18 7.366 2.617 (- - * - - -)
Pooled StDev = 3.362 6.0 8.0 10.0 12.0

Tukey's pairwise comparisons

Family error rate = 0.0500
Individual error rate = 0.0104

Critical value = 3.73

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-2.762		
	2.992		
3	-3.150	-3.265	
	2.976	2.861	
4	-0.142	-0.257	-0.239
	5.692	5.577	5.962

Fisher's pairwise comparisons

Family error rate = 0.200
Individual error rate = 0.0500

Critical value = 1.996

Intervals for (column level mean) - (row level mean)

	1	2	3
2	-2.062		
	2.293		
3	-2.405	-2.520	
	2.231	2.116	
4	0.568	0.452	0.516
	4.983	4.867	5.208

————— 4/26/02 12:26:17 PM —————

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One-way ANOVA: MPa versus Treatment

Analysis of Variance for MPa

Source	DF	SS	MS	F	P
Treatmen	4	656.29	164.07	22.27	0.000
Error	86	633.66	7.37		
Total	90	1289.95			

Individual 95% CIs For Mean
Based on Pooled StDev

Level	N	Mean	StDev	-----+-----+-----+-----
1	19	3.759	1.299	(----*----)
2	18	5.283	2.491	(----*----)
3	16	6.193	2.040	(----*----)
4	19	10.317	4.362	(----*----)
5	19	10.141	2.271	(----*----)
-----+-----+-----+-----				
Pooled StDev =		2.714		5.0 7.5 10.0

Tukey's pairwise comparisons

Family error rate = 0.0500
Individual error rate = 0.00656

Critical value = 3.94

Intervals for (column level mean) - (row level mean)

	1	2	3	4
2	-4.011			
	0.964			
3		-5.000	-3.509	
		0.132	1.688	
4		-9.012	-7.522	-6.690
		-4.104	-2.547	-1.558
5		-8.835	-7.346	-6.514
		-3.928	-2.371	-1.382
				-2.277
				2.630

Fisher's pairwise comparisons

Family error rate = 0.281
Individual error rate = 0.0500

Critical value = 1.988

Intervals for (column level mean) - (row level mean)

	1	2	3	4
2	-3.298			
	0.252			
3		-3.298		
		0.252		
4			-3.298	
			0.252	

3	-4.265 -0.603	-2.765 0.944			
4	-8.309 -4.807	-6.810 -3.260	-5.955 -2.293		
5	-8.132 -4.631	-6.633 -3.083	-5.779 -2.117	-1.574 1.927	

Results for: Worksheet 2

One-way ANOVA: MPa versus Treatment

Analysis of Variance for MPa

Source	DF	SS	MS	F	P
Treatmen	4	622.04	155.51	17.09	0.000
Error	85	773.50	9.10		
Total	89	1395.54			

Individual 95% CIs For Mean
Based on Pooled StDev

Level	N	Mean	StDev	-----+-----+-----+-----+-----
1	17	3.676	0.963	(----*----)
2	19	5.519	1.879	(----*----)
3	16	6.308	2.322	(----*----)
4	19	10.392	2.757	(----*----)
5	19	10.026	5.150	(----*----)
-----+-----+-----+-----+-----				
Pooled StDev =		3.017		3.0 6.0 9.0 12.0

Tukey's pairwise comparisons

Family error rate = 0.0500
Individual error rate = 0.00658

Critical value = 3.94

Intervals for (column level mean) - (row level mean)

	1	2	3	4
2	-4.649 0.963			
3	-5.559 0.296	-3.640 2.063		
4	-9.522 -3.910	-7.600 -2.146	-6.936 -1.233	
5	-9.155 -3.544	-7.233 -1.780	-6.570 -0.866	-2.360 3.093

Fisher's pairwise comparisons

Family error rate = 0.281
Individual error rate = 0.0500

Critical value = 1.988

Intervals for (column level mean) - (row level mean)

	1	2	3	4
2	-3.845 0.159			
3	-4.720 -0.543	-2.823 1.246		
4	-8.718 -4.714	-6.819 -2.927	-6.119 -2.050	
5	-8.351 -4.347	-6.452 -2.561	-5.753 -1.683	-1.579 2.312

Results for: Worksheet 3

One-way ANOVA: MPa versus Treatment

Analysis of Variance for MPa

Source	DF	SS	MS	F	P
Treatmen	4	461.73	115.43	21.33	0.000
Error	79	427.62	5.41		
Total	83	889.36			

Individual 95% CIs For Mean
Based on Pooled StDev

Level	N	Mean	StDev			
1	17	3.881	1.527	(----*---		
2	18	5.485	2.470	(---*---		
3	17	6.059	1.820	(---*---		
4	17	9.235	3.189	(----*---		
5	15	10.228	2.249	(----*---		

Pooled StDev = 2.327

Tukey's pairwise comparisons

Family error rate = 0.0500
Individual error rate = 0.00655
Critical value = 3.95

Intervals for (column level mean) - (row level mean)

	1	2	3	4
2	-3.802 0.594			
3	-4.406 0.051	-2.771 1.624		

4	-7.583	-5.948	-5.405	
	-3.125	-1.552	-0.947	
5	-8.648	-7.014	-6.471	-3.294
	-4.044	-2.470	-1.867	1.309

Fisher's pairwise comparisons

Family error rate = 0.280
 Individual error rate = 0.0500

Critical value = 1.990

Intervals for (column level mean) - (row level mean)

	1	2	3	4
2	-3.170			
3	-3.765	-2.139		
4	-6.942	-5.316	-4.764	
5	-7.986	-6.361	-5.809	-2.633
	-4.706	-3.124	-2.529	0.648

Results for: Worksheet 4

One-way ANOVA: MPa versus Treatment

Analysis of Variance for MPa

Source	DF	SS	MS	F	P
Treatmen	4	206.98	51.74	16.13	0.000
Error	94	301.59	3.21		
Total	98	508.56			

Individual 95% CIs For Mean
 Based on Pooled StDev

Level	N	Mean	StDev				
1	21	3.602	2.029	(-----*	-----)		
2	19	4.204	1.043	(-----*	-----)		
3	21	3.944	1.314	(-----*	-----)		
4	20	6.219	1.602		(-----*	-----)	
5	18	7.366	2.617			(-----*	-----)

Pooled StDev = 1.791

3.0 4.5 6.0 7.5

Tukey's pairwise comparisons

Family error rate = 0.0500
 Individual error rate = 0.00659

Critical value = 3.93

Intervals for (column level mean) - (row level mean)

1	2	3	4
---	---	---	---

2	-2.179 0.973			
3	-1.878 1.194	-1.316 1.836		
4	-4.173 -1.062	-3.609 -0.420	-3.830 -0.720	
5	-5.363 -2.165	-4.799 -1.524	-5.021 -1.823	-2.764 0.471

Fisher's pairwise comparisons

Family error rate = 0.281
 Individual error rate = 0.0500

Critical value = 1.986

Intervals for (column level mean) - (row level mean)

	1	2	3	4
2	-1.729 0.524			
3	-1.440 0.756	-0.866 1.387		
4	-3.729 -1.506	-3.154 -0.875	-3.387 -1.164	
5	-4.907 -2.621	-4.331 -1.991	-4.564 -2.279	-2.302 0.009

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 Retrieving project from file: F:\chris\Tensile.mpj

Results for: Worksheet 1

One-way ANOVA: Peak Load Gm versus Treatment

Analysis of Variance for Peak Loa

Source	DF	SS	MS	F	P
Treatmen	4	6110	1527	2.61	0.040
Error	105	61479	586		
Total	109	67589			

Level	N	Individual 95% CIs For Mean Based on Pooled StDev			
		Mean	StDev	(-----*-----)	(-----*-----)
1	22	274.84	28.66	(-----*-----)	(-----*-----)
2	22	270.50	20.55		(-----*-----)
3	23	282.28	11.66		(-----*-----)
4	20	258.90	36.77	(-----*-----)	
5	23	273.17	17.41		(-----*-----)
Pooled StDev = 24.20		252	264	276	288

Tukey's pairwise comparisons

Family error rate = 0.0500

Individual error rate = 0.00646

Critical value = 3.93

		Intervals for (column level mean) - (row level mean)			
		1	2	3	4
2	-15.94				
	24.61				
3	-27.50	-31.83			
	12.61	8.28			
4	-4.84	-9.17	2.82		
	36.71	32.38	43.94		
5	-18.38	-22.72	-10.72	-34.82	
	21.72	17.39	28.94	6.29	

Fisher's pairwise comparisons

Family error rate = 0.282

Individual error rate = 0.0500

Critical value = 1.983

		Intervals for (column level mean) - (row level mean)			
		1	2	3	4
2	-10.14				
	18.80				
3	-21.75	-26.08			
	6.87	2.53			
4	1.11	-3.22	8.71		
	30.76	26.43	38.05		
5	-12.64	-16.97	-5.04	-28.93	
	15.98	11.65	23.26	0.41	

Appendix J. Transverse Compression Machine Instructions

Instructions for Compression Tester:

1. Plug in all machines (four plugs total). Turn on computer. Switch on Load control machine at back. Press tare.
2. Turn on air at Fume hood.
3. Turn on air valve at Compression machine to read 25psi.
4. Open Quick Log Application. Immediately stop the program from running.
5. Go to File – Open Configuration – Chris50
6. Open API-mate application. Go to File – click on “1.68 micron run” and also, “backup sequence2”
7. For Compliance, run the device until a load of approx. 5000g (top machine read-out). The bottom machine read-out is the displacement, it should appear to get lower as the device is run and compression takes place.
8. To run device, click on >> (Play) on the APImate. Then immediately click RUN on the Quick log application.
9. Once the load reaches 5000g, click the STOP button on the APImate app. Then wait a few seconds before clicking the STOP on the Quick log app. This is because of a slight delay.
10. The file is saved in sequential order in the file CHRIS, in .txt format.
11. The .txt files can then be opened in excel and/or saved on a floppy to take to another computer for analysis.

Appendix K. Compliance Instructions

Instructions for taking compliance values into account for seven fiber bundle compression testing:

1. Find 100g Load. Delete, Shift up.

Time Disp. Load

2. \$B\$1 – B! New Column - Copy down. Copy and paste special values

3. Delete Column B

4. Save

5. Open Compliance .exe app

6. Type Tes3.txt

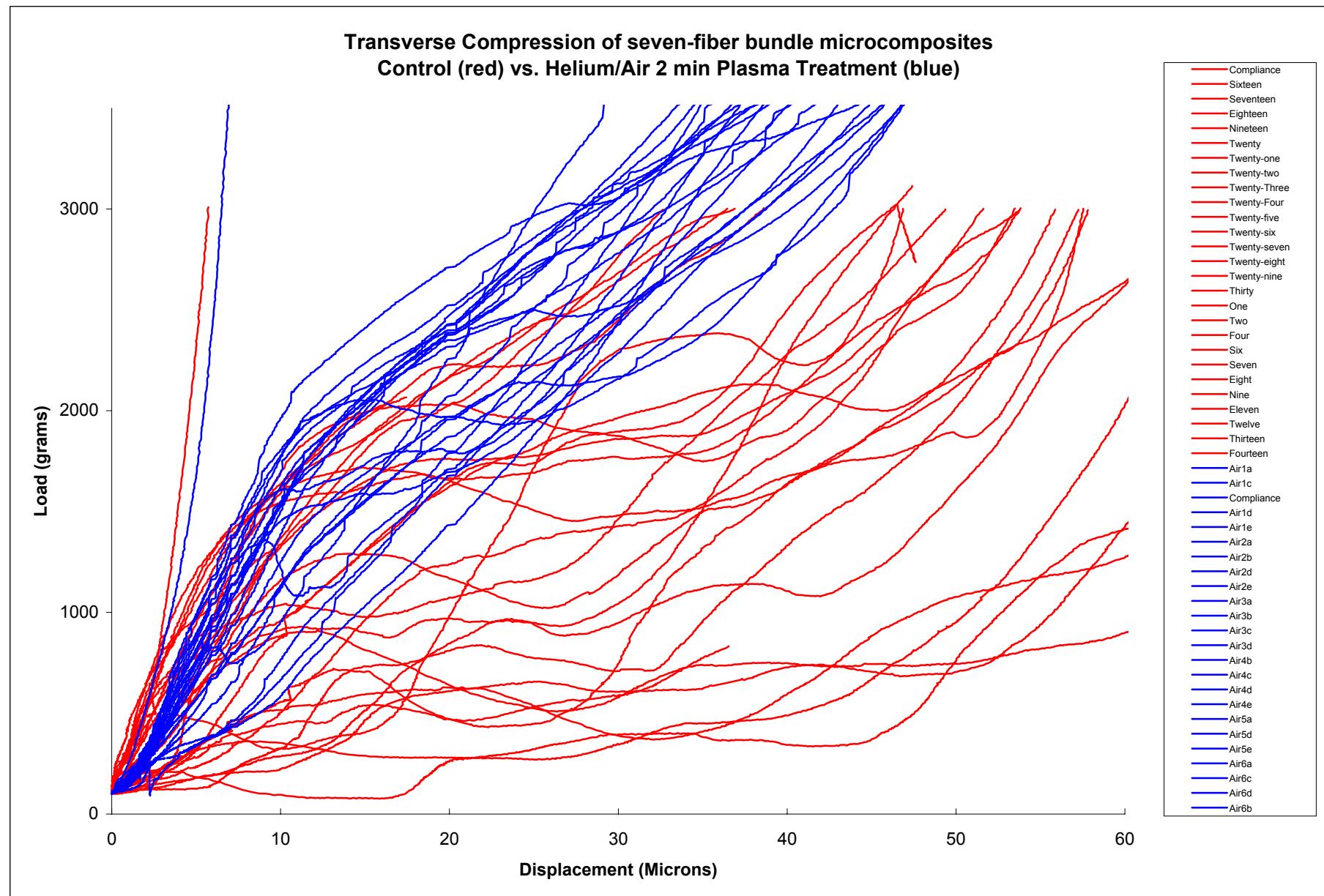
7. Then Tes3.out

8. In Excel, open tes.out. Type labels.

Time Load (g) Disp.(Microns)

9. Copy load and Displacement columns to tes2. And Graph

Reminder: Change displacement to X column and load to Y column in order to graph.



Appendix M. Transverse Compression data for Figure 3.11

2min He/Air Plasma						2min He/Air Plasma						Control						Control		
Displacement (Microns)			Control			Displacement (Microns)			Load (Grams)			Load (Newtons)			2min He/Air Plasma			Control		
Load (Grams)	Load (Newtons)		Load (Grams)	Load (Newtons)		Load (Microns)	Load (Grams)	Load (Newtons)	Load (Grams)	Load (Newtons)		Load (Microns)	Load (Grams)	Load (Newtons)	Load (Microns)	Load (Grams)	Load (Newtons)	Load (Microns)	Load (Grams)	Load (Newtons)
0	101.661	0.996676471	0	101.869	0.998715868	0	106.282	1.041980392	3.102	285.967	2.823990196	3.12769	292.458	2.867235294	2.27774	288.288	2.826352941	2.306	290.329	2.846362745
0.0554581	105.36	1.032941176	0	106.907	1.041980392	0	110.1	1.079411765	3.17906	295.693	2.898950908	3.20563	288.587	2.829284314	2.35466	290.569	2.848715686	2.40063	295.854	2.929941176
0.0779445	107.747	1.056343137	0.10493	109.418	1.07272549	0.0349892	109.622	1.07472549	3.20474	296.329	2.924794118	3.20474	297.97	2.92127451	2.36947	291.529	2.858127451	2.40347	295.611	2.898147059
0.104238	113.238	1.110176471	0.123877	113.477	1.112519608	0.0684538	108.907	1.067715868	3.20474	296.928	2.930666667	3.20474	297.97	2.92127451	2.40063	295.854	2.929941176	2.4168	299.935	2.940539216
0.157034	116.581	1.14295098	0.254179	118.97	1.166372549	0.0836845	115.588	1.13322549	3.25612	301.685	2.957696078	3.25612	301.685	2.957696078	2.44217	299.971	2.940892157	2.44217	299.971	2.940892157
0.311677	121.837	1.194480392	0.303234	126.36	1.238823529	0.169251	120.125	1.177696078	3.312	307.319	3.012931373	3.312	307.319	3.012931373	2.49236	302.487	3.065558824	2.49236	306.801	3.024896078
0.372778	127.075	1.245833333	0.38684	129.697	1.27163216	0.269648	125.021	1.225696078	3.30442	310.917	3.048205882	3.30442	310.917	3.048205882	2.51805	307.76	3.017254902	2.51805	307.76	3.017254902
0.39884	133.363	1.313917647	0.398474	133.663	1.35453016	0.307166	133.211	1.310519608	3.30291	311.035	3.08185139	3.30291	311.035	3.08185139	2.55607	311.956	3.062254257	2.55607	311.956	3.062254257
0.473588	139.052	1.372078431	0.503387	141.264	1.384941176	0.407053	137.429	1.347343137	3.34892	318.155	3.123039232	3.34892	318.155	3.123039232	2.62315	313.514	3.073666667	2.62315	313.514	3.073666667
0.610701	142.696	1.398880392	0.618893	145.321	1.424715686	0.404412	145.299	1.4245	3.36863	319.915	3.136421569	3.36863	319.915	3.136421569	2.65461	313.274	3.071313725	2.65461	313.274	3.071313725
0.6744	147.111	1.442264706	0.675194	150.452	1.475019608	0.4749472	153.172	1.501686275	3.43771	323.755	3.174068627	3.43771	323.755	3.174068627	2.69946	315.313	3.091303922	2.69946	318.551	3.149243092
0.735209	153.914	1.508960784	0.768794	156.899	1.53822549	0.599009	157.349	1.542637255	3.50440	322.278	3.228215686	3.50440	322.278	3.228215686	2.71283	324.789	3.184205882	2.71283	324.789	3.184205882
0.812364	158.212	1.551098039	0.88111	161.915	1.587401961	0.601191	161.646	1.584764706	3.56612	324.682	3.281196078	3.56612	325.509	3.191264706	2.78578	325.509	3.191264706	2.78578	325.509	3.191264706
0.945806	165.14	1.619019608	0.960463	168.008	1.647137255	0.654406	167.259	1.639794118	3.57089	320.208	3.302392157	3.57089	320.208	3.302392157	2.82588	327.429	3.210088235	2.82588	327.429	3.210088235
1.03022	169.92	1.656882353	1.04994	171.473	1.68107843	0.640895	173.351	1.695919608	3.61777	320.929	3.342441176	3.61777	320.929	3.342441176	2.81946	328.389	3.2195	2.81946	328.389	3.2195
1.12752	171.354	1.727334118	1.1778	175.776	1.732308447	0.701819	174.903	1.714745098	3.6315	328.756	3.203490196	3.6315	328.756	3.203490196	2.84674	331.99	3.254803922	2.84674	331.99	3.254803922
1.195	179.445	1.78702549	1.1503	179.722	1.761690392	0.701819	182.622	1.802034118	3.66697	328.756	3.228215686	3.66697	328.756	3.228215686	2.88446	334.03	3.274803922	2.88446	334.03	3.274803922
1.21816	180.559	1.770186275	1.21894	182.712	1.791249118	0.801247	186.977	1.833107843	3.6755	328.551	3.246162875	3.6755	328.551	3.246162875	2.90446	335.153	3.3071313725	2.90446	335.153	3.3071313725
1.23608	185.463	1.818264706	1.2890	187.018	1.833509804	0.881201	189.488	1.85772549	3.78614	325.437	3.294480392	3.78614	325.437	3.294480392	2.94446	343.397	3.366837255	2.94446	343.397	3.366837255
1.29457	191.096	1.873921517	1.35541	190.128	1.884	0.938323	193.914	1.901117647	3.82415	326.394	3.256170582	3.82415	326.394	3.256170582	2.98446	340.847	3.343674671	2.98446	340.847	3.343674671
1.33418	192.283	1.885127451	1.35045	193.24	1.894508084	0.969662	196.187	1.923401961	3.8755	326.979	3.257822549	3.8755	326.979	3.257822549	2.98446	343.153	3.462058824	2.98446	343.153	3.462058824
1.42637	194.437	1.906245098	1.42637	195.275	1.914460784	0.972127	198.82	1.949215686	3.90118	326.781	3.261801960	3.90118	326.781	3.261801960	2.98446	343.015	3.490343137	2.98446	343.015	3.490343137
1.42637	195.275	1.914460784	1.45207	196.352	1.925019608	1.04052	202.889	1.981074743	3.96602	327.028	3.2647813725	3.96602	327.028	3.2647813725	2.98446	343.015	3.503423529	2.98446	343.015	3.503423529
1.47777	197.78	1.939098039	1.50347	199.105	1.950200804	1.07001	204.204	2.002019608	3.92152	326.465	3.269083333	3.92152	326.465	3.269083333	2.98446	343.089	3.596460784	2.98446	343.089	3.596460784
1.52918	200.422	1.964921569	1.58059	203.057	1.990754902	1.141424	209.359	2.05890196	3.95975	326.853	3.275049402	3.95975	326.853	3.275049402	2.98446	343.163	3.62717451	2.98446	343.163	3.62717451
1.65768	203.363	1.995666667	1.68838	204.309	2.004147059	1.1707	210.101	2.068921589	3.96462	326.853	3.280903922	3.96462	326.853	3.280903922	2.98446	343.164	3.62717451	2.98446	343.164	3.62717451
1.69366	208.096	2.040156893	1.71692	208.338	2.045221569	1.18815	212.467	2.083090804	4.005	326.918	3.293312725	4.005	326.918	3.293312725	2.98446	343.397	3.631097843	2.98446	343.397	3.631097843
1.71692	208.338	2.045221569	1.75459	210.399	2.083215686	1.20399	214.144	2.090549598	4.03068	326.803	3.281590839	4.03068	326.803	3.281590839	2.98446	343.397	3.631907843	2.98446	343.397	3.631907843
1.75459	209.529	2.054205882	1.77068	210.666	2.062411764	1.22475	214.983	2.107674741	4.05636	326.803	3.283090839	4.05636	326.803	3.283090839	2.98446	343.397	3.632807843	2.98446	343.397	3.632807843
1.77068	210.666	2.062411764	1.79422	212.278	2.081568683	1.22804	215.521	2.125205882	4.09785	326.803	3.288090392	4.09785	326.803	3.288090392	2.98446	343.397	3.633252949	2.98446	343.397	3.633252949
1.79422	212.278	2.081568683	1.824728	213.877	2.108264706	1.23606	218.108	2.138313725	4.13783	326.803	3.29572549	4.13783	326.803	3.29572549	2.98446	343.397	3.63783414	2.98446	343.397	3.63783414
1.824728	213.877	2.108264706	1.863321	215.624	2.113960784	1.34244	221.091	2.167607483	4.1798	326.803	3.301950878	4.1798	326.803	3.301950878	2.98446	343.397	3.639057888	2.98446	343.397	3.639057888
1.863321	215.624	2.113960784	1.88936	218.374	2.140921569	1.37594	224.084	2.196901961	4.18365	326.803	3.309509878	4.18365	326.803	3.309509878	2.98446	343.397	3.640491961	2.98446	343.397	3.640491961
1.88936	218.374	2.140921569	1.93822	218.972	2.146784314	1.38459	224.988	2.208991061	4.21179	326.803	3.317254902	4.21179	326.803	3.317254902	2.98446	343.397	3.642058824	2.98446	343.397	3.642058824
1.93822	218.972	2.146784314	1.953413	220.572	2.146784314	1.40444	227.073	2.240917647	4.24145	326.803	3.322016725	4.24145	326.803	3.322016725	2.98446	343.397	3.642252949	2.98446	343.397	3.642252949
1.953413	220.572	2.146784314	1.97404	221.837	2.146784314	1.42441	227.467	2.240917647	4.25044	326.803	3.322016725	4.25044	326.803	3.322016725	2.98446	343.397	3.642452949	2.98446	343.397	3.642452949
1.97404</																				

2min He/Air Plasma				Control				2min He/Air Plasma				Control			
4.99471	569.139	5.579794118	5.13453	604.401	5.9255	6.98192	1119.06	10.97117647	9.06466	1080.05	10.58872549	9.04646	1082.75	10.61519608	
5.02435	573.856	5.526039216	5.15692	608.278	5.963509804	6.95402	1127.01	11.04911765	9.14946	1082.75	10.61519608	9.14946	1082.75	10.61519608	
5.0138	578.937	5.675852941	5.20823	609.368	5.974196078	7.02029	1131.56	11.09372549	9.1599	1087.54	10.66215686	9.1599	1087.54	10.66215686	
5.04528	583.899	5.7245	5.19275	615.307	6.032421569	7.02873	1138.94	11.16607843	9.16458	1091.6	10.70196078	9.16458	1091.6	10.70196078	
5.06566	586.805	5.75201956	5.2348	618.702	6.063805082	7.0689	1145.58	11.23117647	9.18692	1094.55	10.73088235	9.18692	1094.55	10.73088235	
5.09132	588.291	5.79008919	5.27629	620.522	6.083305116	7.0941	1152.92	11.28656932	9.24055	1101.68	10.80077931	9.24055	1101.68	10.80077931	
5.0922	595.281	5.80089235	5.27017	627.459	6.128715969	7.05547	1155.31	11.32656932	9.20976	1105.74	10.84058824	9.20976	1105.74	10.84058824	
5.14156	599.642	5.87884137	5.32043	627.917	6.156986927	7.09855	1161.59	11.38813725	9.22429	1108.82	10.877087431	9.22429	1108.82	10.877087431	
5.16329	601.46	5.896666667	5.32245	630.831	6.184617647	7.03834	1168.98	11.46058824	9.34229	1110.82	10.877087431	9.34229	1110.82	10.877087431	
5.15554	604.126	5.922803922	5.3849	634.107	6.216735294	7.15342	1176.26	11.53196078	9.40906	1112.38	10.90568827	9.40906	1112.38	10.90568827	
5.13057	607.762	5.95845098	5.38466	637.511	6.250107843	7.18557	1181.68	11.58509804	9.47068	1116.32	10.94431373	9.47068	1116.32	10.94431373	
5.21834	614.067	6.020264706	5.39706	641.817	6.292843137	7.15185	1189.57	11.66245098	9.53534	1120.76	10.98784314	9.53534	1120.76	10.98784314	
5.20776	618.675	6.065441176	5.38408	647.079	6.343911765	7.19391	1193.77	11.70362745	9.53185	1123.46	11.01431373	9.53185	1123.46	11.01431373	
5.2275	623.663	6.114343137	5.44197	649.865	6.371253294	7.23709	1201.3	11.77745098	9.56263	1127.28	11.05176471	9.56263	1127.28	11.05176471	
5.18866	628.02	6.157058824	5.46762	652.047	6.392617647	7.26279	1208.46	11.84764706	9.63946	1131.59	11.09401961	9.63946	1131.59	11.09401961	
5.24045	630.078	6.177325294	5.53962	657.38	6.444901961	7.28849	1213.01	11.8922549	9.66507	1133.32	11.1098039	9.66507	1133.32	11.1098039	
5.28594	633.59	6.211666667	5.49363	660.411	6.474617647	7.32978	1220.28	11.96352941	9.7163	1134.9	11.12666667	9.7163	1134.9	11.12666667	
5.29396	636.133	6.236598039	5.57059	662.59	6.496090804	7.34029	1225.71	12.01676471	9.76752	1140.78	11.18411765	9.76752	1140.78	11.18411765	
5.33707	640.252	6.276980392	5.58114	665.626	6.525753294	7.38871	1232.12	12.07960784	9.83896	1142.37	11.19970588	9.83896	1142.37	11.19970588	
5.33782	647.522	6.348254902	5.61451	669.871	6.567362475	7.36629	1240.39	12.16068627	9.96641	1147.17	11.24676471	9.96641	1147.17	11.24676471	
5.33782	649.933	6.368460784	5.65458	671.327	6.581637255	7.39199	1246.94	12.22409196	9.8969	1151.5	11.24676471	9.8969	1151.5	11.24676471	
5.35467	652.493	6.396991916	5.7245	676.181	6.62922549	7.4449	1251.76	12.27215686	9.94158	1153.2	11.30588235	9.94158	1153.2	11.30588235	
5.38661	658.435	6.455182549	5.74745	680.79	6.658182549	7.48176	1258.28	12.33980804	10.003	1154.49	11.357151	10.003	1154.49	11.357151	
5.41154	662.446	6.49804902	5.75037	688.969	6.682515683	7.49165	1264.59	12.3990329	9.98903	1159.81	11.41	9.98903	1159.81	11.41	
5.41156	662.169	6.540287249	5.75166	689.901	6.703525294	7.51789	1271.04	12.45117647	10.15262	1165.76	12.49021961	10.15262	1165.76	12.49021961	
5.44139	671.178	6.580147764	5.75191	694.517	6.80890196	7.47701	1276.23	12.51205982	10.15262	1168.1	11.45190678	10.15262	1168.1	11.45190678	
5.45382	674.331	6.611082835	5.80321	696.826	6.831627451	7.53031	1284.77	12.59578431	10.2294	1169.57	11.46637255	10.2294	1169.57	11.46637255	
5.45382	678.945	6.656323259	5.88016	700.35	6.866176471	7.58871	1291.83	12.665	10.2806	1174.5	11.51470588	10.2806	1174.5	11.51470588	
5.52769	683.682	6.702267406	5.86888	704.97	6.914705888	7.59316	1296.79	12.71362745	10.352	1177.34	12.54524902	10.352	1177.34	12.54524902	
5.537	688.299	6.740294912	5.88005	707.888	6.940074831	7.61843	1304.46	12.78828353	10.3947	1180.5	11.57932157	10.3947	1180.5	11.57932157	
5.51965	693.524	6.799254902	5.93198	711.655	6.970392162	7.62847	1310.27	12.84578431	10.4381	1183.5	11.60294118	10.4381	1183.5	11.60294118	
5.54537	699.861	6.812362745	5.93201	715.429	7.014009804	7.60028	1315.71	12.89911765	10.5098	1185.23	11.61990196	10.5098	1185.23	11.61990196	
5.59715	699.968	6.862431373	5.97487	720.816	7.06862329	7.63904	1322.27	12.96343137	10.5448	1189.42	11.66098039	10.5448	1189.42	11.66098039	
5.6016	705.562	6.91727451	6.00944	725.551	7.112345098	7.72	1329.08	13.03019608	10.6121	1192.01	11.68637255	10.6121	1192.01	11.68637255	
5.61932	707.873	6.939931373	6.00761	729.073	7.1477451	7.67737	1337.5	13.1127451	10.6565	1194.35	11.70931373	10.6565	1194.35	11.70931373	
5.62081	711.859	6.998617647	6.03544	733.931	7.195401961	7.75446	1345.68	13.1924118	10.6975	1197.81	11.74320529	10.6975	1197.81	11.74320529	
5.62947	718.473	7.043852941	6.06109	736.601	7.221607843	7.80585	1348.28	13.21843137	10.7796	1199.17	11.75656863	10.7796	1199.17	11.75656863	
5.67595	719.565	7.054558824	6.08155	745.234	7.306215688	7.84714	1351.5	13.28529412	10.82	1201.51	11.7795998	10.82	1201.51	11.7795998	
5.68334	722.723	7.085519608	6.11776	748.151	7.338173259	7.88722	1354.22	13.3482529	10.9008	1205.46	11.81215686	10.9008	1205.46	11.81215686	
5.6972	726.875	7.14362529	6.16776	750.559	7.387345294	7.92834	1361.5	13.4090839	10.952	1210.28	11.88554902	10.952	1210.28	11.88554902	
5.69463	726.875	7.221262745	6.18963	750.904	7.407284314	7.92834	1362.31	13.52059882	11.0048	1210.28	11.88554902	11.0048	1210.28	11.88554902	
5.72824	740.465	7.250460784	6.19511	758.612	7.437372549	7.93482	1368.82	13.62382253	11.1118	1212.5	11.88772549	11.1118	1212.5	11.88772549	
5.75396	748.492	7.338156863	6.23893	758.237	7.48382157	7.93629	1369.33	13.67970588	11.1374	1216.82	11.92960784	11.1374	1216.82	11.92960784	
5.77989	750.56	7.358431373	6.29318	769.566	7.547467406	7.95105	1371.04	13.7446275	11.2067	1219.82	11.95019608	11.2067	1219.82	11.95019608	
5.83113	753.845	7.390637253	6.30181	772.732	7.575803922	7.98807	1372.32	13.78044118	11.2634	1224.45	12.00441176	11.2634	1224.45	12.00441176	
5.80883	755.897	7.41495088	6.30961	777.603	7.623558824	8.01376	1374.55	13.877944118	11.3379	1225.19	12.01166667	11.3379	1225.19	12.01166667	
5.86791	759.847	7.432176471	6.34912	784.426	7.670478431	8.05494	1375.82	14.00823529	11.4154	1228.89	12.04794118	11.4154	1228.89	12.04794118	
5.81161	761.863	7.441666667	6.45279	791.74	7.721625683	8.03865	1374.18	14.06058824	11.5013	1230.13	12.06090084	11.5013	1230.13	12.06090084	
5.80359	762.877	7.481063725	6.46279	794.924	7.78845098	8.07277	1374.49	14.13127451	11.5673	1234.81	12.10598039	11.5673	1234.81	12.10598039	
5.81566	763.357	7.506372549	6.48005	797.379	7.810574831	8.11751	1375.86	14.1922549	11.6136	1237.28	12.3019608	11.6136	1237.28	12.3019608	
6.01566	791.807	9.00711765	6.70367	870.333	9.32732549	8.15276	1385.45	14.2727451	12.0249	1256.66	12.3019608	12.0249	1256.66	12.3019608	
6.32675	907.001	9.11765	6.80085	834.574	9.48098039	8.24714	1386.78	14.37994092	12.094	1257.27	12.33509839	12.094	1257.27	12.33509839	
6.35814	923.7	9.16606	6.88085	842.847	9.54582157	8.24746	1387.08	14.45940921	12.1343	1257.57	12.43341373	12.1343	1257.57	12.43341373	
6.3788	929.73	9.11460784	6.96992</												

2min He/Ar Plasma										Control									
10.3944	1852.57	18.16245098	Control	17.5915	1255.55	12.30931373	17.3485	2383.87	23.37127451	29.8556	1175	11.51960784							
10.4277	1857.85	18.21421569		17.7022	1252.22	12.27666667	17.3938	2386.3	23.39509804	29.9006	1177.58	11.54490196							
10.4927	1862.51	18.25990196		17.8308	1245.06	12.20647059	17.445	2388.85	23.42009804	29.9643	1182.52	11.59333333							
10.5408	1869.31	18.32656863		17.9436	1238.64	12.14352941	17.5475	2395.75	23.4877451	30.0292	1186.34	11.63078431							
10.5921	1874.73	18.37970588		18.1061	1234.32	12.19117629	17.5698	2401.75	23.54656863	30.0617	1191.15	11.67794118							
10.6691	1880.45	18.43758431		18.2019	1227.41	12.0343137	17.6677	2404.56	23.57411765	30.1011	1195.09	11.68568663							
10.7461	1886.17	18.49362569		18.3036	1220.66	11.983937255	17.7224	2407.69	23.62540902	30.1519	1204.4	11.71646735							
10.7718	1890.64	18.53569026		18.4570	1220.16	11.96230469	17.8128	2409.8	23.6546902	30.2122	1205.05	11.74166764							
10.8026	1898.69	18.61469764		18.5726	1214.47	11.90656863	17.8577	2413.76	23.66431373	30.2666	1209.91	11.86168275							
10.8491	1903.22	18.65901961		18.6606	1207.47	11.83401961	17.922	2420.93	23.73460784	30.2666	1215.09	11.91264706							
10.9004	1909.52	18.72078431		18.8389	1202.62	11.79039216	17.9968	2422.08	23.74588223	30.4138	1218.55	11.94656863							
11.003	1914.31	18.76774751		18.9646	1197.07	11.73598039	18.0634	2427.84	23.80235294	30.4576	1220.53	11.965808039							
11.0479	1918.09	18.80408392		19.1053	1192.75	11.69362745	18.1114	2431.16	23.83409196	30.4953	1224.21	12.00205882							
11.098	1924.27	18.86539216		19.1985	1188.49	11.64892157	18.1626	2434.88	23.87137255	30.5895	1226.67	12.02617647							
11.1063	1931.33	18.93460784		19.337	1185.47	11.62225459	18.2393	2438.46	23.90647059	30.6496	1231.98	12.07823529							
11.1833	1937.26	18.9927451		19.4452	1183.62	11.60411765	18.3132	2446.27	23.98303922	30.7146	1237.78	12.13509804							
11.2551	1942.3	19.04215666		19.5551	1180.79	11.57637255	18.3597	2449.09	24.01068627	30.7108	1239.63	12.15233529							
11.2975	1947.23	19.0904902		19.6355	1179.43	11.56303922	18.4554	2453.06	24.04960784	30.8048	1243.94	12.1954902							
11.3634	1953.29	19.14990196		19.7444	1176.97	11.53892157	18.5247	2457.55	24.09362745	30.856	1246.79	12.22343137							
11.4292	1958.85	19.20441176		19.8769	1174.38	11.51352941	18.5714	2460.24	24.12	30.956	1252.96	12.28392157							
11.4215	1962.64	19.24156863		19.9793	1168.47	11.45556824	18.6378	2466.65	24.18284314	30.9978	1256.79	12.32147059							
11.5432	1967.27	19.28696078		20.1073	1165.39	11.42539216	18.7052	2472.03	24.23558824	31.0946	1259.01	12.37493229							
11.5945	1974.33	19.35817647		20.1841	1163.05	11.40245098	18.752	2476.25	24.28969078	31.1379	1262.22	12.37470588							
11.6182	1985.18	19.46254902		20.2033	1159.72	11.38052352	18.8032	2482.4	24.34725395	31.2145	1268.77	12.421557							
11.6763	1988.69	19.50089235		20.4491	1158.28	11.33984343	18.8908	2489.19	24.39394949	31.2507	1271.26	12.48656863							
11.703	1998.09	19.55089235		20.5945	1152.46	11.29862745	18.9272	2493.42	24.44529412	31.3851	1275.7	12.5068275							
11.7747	2004.42	19.57274751		20.6861	1158.28	11.25764706	19.0238	2497.52	24.49540203	31.4181	1279.04	12.539680784							
11.826	1998.06	19.58823533		20.8269	1145.69	11.23225459	19.094	2502.26	24.53196078	31.4214	1283.12	12.57968078							
11.903	2005.14	19.65823529		20.9325	1141.39	11.19009804	19.1605	2509.31	24.60107843	31.4792	1286.58	12.61352941							
11.949	2009.81	19.70401961		21.0279	1138.19	11.15872549	19.1798	2512.52	24.63254902	31.525	1289.06	12.63784314							
12.0313	2017.02	19.77407588		21.1815	1135.41	11.13147059	19.2426	2519.58	24.70176471	31.6659	1293.89	12.68519608							
12.1035	2021.19	19.81558824		21.2583	1132.33	11.10127451	19.3412	2526.51	24.76970588	31.7244	1296.81	12.71162575							
12.1736	2024.99	19.85284314		21.3606	1127.28	11.05176471	19.3924	2529.72	24.80117645	31.756	1301.81	12.76284314							
12.2455	2027.77	19.88009084		21.5615	1121.25	10.99247067	19.4436	2535.63	24.85911765	31.8472	1304.18	12.7860743							
12.2611	2033.44	19.93470588		21.6579	1118.05	10.91627451	19.5268	2539.1	24.93931375	31.8779	1308.63	12.82970588							
12.3327	2036.38	19.9645098		21.7976	1113.49	10.91656863	19.558	2543.34	24.93470588	31.926	1314.69	12.88911765							
12.4247	2041.06	20.01039216		21.848	1112.02	10.90215686	19.6114	2551.61	25.01578431	31.9896	1317.16	12.91333333							
12.4442	2045.88	20.05764706		21.9864	1108.32	10.86586235	19.6704	2556.23	25.06107843	31.9633	1322.11	12.96166275							
12.5297	2052.97	20.12715686		22.1265	1104.42	10.82372549	19.7092	2564.19	25.13911765	32.0555	1324.58	12.98607403							
12.5452	2057.14	20.16803922		22.2578	1102.91	10.81284314	19.7826	2567.14	25.16803922	32.0888	1331.51	13.05401961							
12.5965	2059.03	20.18858863		22.3491	1100.81	10.7985824	19.8739	2574.77	25.19835294	32.1728	1334.48	13.08081725							
12.6091	2060.49	20.19991667		22.4501	1097.47	10.75217451	19.9235	2581.47	25.22896275	32.2595	1339.59	13.088957							
12.6264	2071.3	20.30698275		22.6438	1086.65	10.67039022	19.9851	2583.58	25.29211693	32.3147	1346.16	13.39196161							
12.6274	2074.98	20.34294118		22.7911	1082.51	10.61284314	20.0692	2591.68	25.40862745	32.4073	1343.28	13.6941176							
12.9043	2079.91	20.39127451		22.8529	1078.45	10.57303922	20.1206	2596.18	25.4527451	32.4531	1344.39	13.63029412							
12.9566	2083.33	20.42480392		23.0008	1074.28	10.53215686	20.1829	2600.3	25.493313725	32.489	1347.61	13.211866275							
12.9968	2088.28	20.48313725		23.1722	1069.24	10.4827451	20.2284	2608.27	25.57127451	32.5437	1352.81	13.26284314							
13.053	2095.11	20.54029412		23.2818	1068.14	10.47196078	20.2751	2612.78	25.6154902	32.5817	1357.77	13.31147059							
13.059	2097.65	20.56519608		23.4553	1067.83	10.39980392	20.3521	2619.86	25.658490196	32.6558	1359.51	13.32852541							
13.1821	2103.73	20.62480392		23.5662	1055.26	10.34568627	20.3777	2625.53	25.7404902	32.6666	1364.22	13.37470588							
13.227	2107.28	20.65960784		23.6662	1054.52	10.33841373	20.4804	2629.65	25.78088235	32.7335	1367.82	13.41							
13.2947	2112.48	20.71058824		23.7379	1052.28	10.32203598	20.5036	2635.42	25.83745098	32.8033	1372.66	13.457470598							
13.3343	2118.96	20.77411765		23.8395	1050.43	10.29833333	20.565	2644.55	25.82696078	32.8534	1375.14	13.48176471							
13.4008	2124.67	20.83009804		23.939	1049.57	10.28909196	20.6095	2648.54	25.88067843	32.891	1377.74	13.5072549							
13.4697	2129.11	20.87362745		24.1209	1046.25	10.25735294	20.6736	2654.72	26.02666667	32.9223	1381.59	13.545							
13.5723	2134.11	20.92647067		24.2534	1041.95	10.21519608	20.7165	2658.71	26.05678431	33.0123	1383.95	13.56813725							
14.1914	2180.81	21.3804902		25.5213	1010.88	10.1558824	21.4156	2701.72	26.67558824	33.6229	1422.08	13.91496078							
14.3123	2186.27	21.43401961		25.5838	1012.39	10.13823529	21.4823	2702.5	26.72745098	33.7132	1424.31	13.96323523							
14.4344	2190.08	21.47137255		25.7446	1022.43	10.20382533	21.5208	2726.52	26.78931373	33.7387	1429.03	14.01009804							
14.4475	2198.6	21.55490196		25.8256	1021.57	10.1539216	21.5615	2737.68	26.8444	33.7765	1431.64	14.03586627							
14.4494	2202.8	21.59670784		25.901	1021.82	10.1784314	21.6144	2748.52	26.88803922	33.8532	1435.38	14.10147059							
14.4982	2207.2	21.6372549		25.9709	1023.16	10.30396039	21.6657	2749.94	26.96019608	33.8788	1440.09	14.11582941							
14.6826	2212.22	21.68843137		26.1813	1023.63	10.30767471	21.7899	2761.05	27.06911765	33.9979	1445.44	14.17098039							
14.6927	2218.26	21.75294118		26.2288	1020.99	10.28835253	21.8925	2769.19	27.14892157	34.0672	1448.71	14.1977451							
14.8064	2223.13	21.79543216		26.3084	1020.55	10.26839216	22.0347	2777.8	27.203255	34.1602	1454.02	14.25008804							
14.8782	2224.02	21.84747067		26.4555	1020.45	10.23594949	22.0463	2785.75</td											

2min He/Air Plasma	Control	2min He/Air Plasma	Control								
24.3297	2992.25	29.33578431	36.7888	1522.28	14.92431373	31.2205	3708.66	36.35941176	44.388	1845.8	18.09607843
24.3752	2997.71	29.38931373	36.8877	1523.15	14.93264314	31.2972	3713.31	36.405	44.4603	1847.43	18.11205882
24.4619	3004.86	29.45941176	36.9792	1526.01	14.96068235	31.3483	3718.23	36.45323529	44.529	1852.21	18.15892157
24.4557	3008.11	29.49127451	37.0331	1525.27	14.95362745	31.4468	3723	36.5	44.6309	1856.24	18.19843137
24.5453	3016.19	29.5704902	37.0998	1529.75	14.99754902	31.4851	3729.76	36.68627451	44.6457	1862.28	18.25764706
24.5619	3021.91	29.6268863	37.1764	1531.87	15.01833333	31.553	3735.6	36.82035241	44.6683	1868.95	18.32303922
24.6254	3024.29	29.7001951	37.2533	1533.15	15.03058861	31.6253	3741.16	36.97003216	44.774	1874	18.40339524
24.7004	3036.39	29.7722549	37.3551	1533.42	15.05038861	31.6953	3748.35	36.98492341	44.833	1877.14	18.40339333
24.7897	3036.77	29.85109608	37.4062	1534.49	15.05323253	31.7456	3755.93	36.98234312	45.5362	1912.06	17.5989827
24.8515	3044.72	29.85109608	37.5083	1537.86	15.07705882	31.7816	3760.31	36.86578473	45.7048	1915.76	17.78156863
24.9296	3052.28	29.92431373	37.576	1541.1	15.10828253	31.8092	3767.36	36.93490196	45.7511	1917.23	17.79637255
24.9993	3057.49	29.97539216	37.6249	1541.6	15.1372549	31.937	3774.54	37.00529412	45.8583	1915.59	17.80292412
25.0983	3063.74	30.03666667	37.6946	1543.22	15.12960784	31.9626	3780.8	37.06666667	45.955	1917.35	17.879754902
25.1588	3068.04	30.07882353	37.7586	1545.46	15.15294212	32.0137	3786.39	37.12147059	46.0213	1920.63	17.89705888
25.2077	3075.47	30.15166667	37.8441	1551.2	15.20743414	32.1159	3792.52	37.18156863	46.0755	1919.62	18.19809392
25.2207	3080.81	30.20401961	37.9462	1553.2	15.22745098	32.2182	3795.98	37.2154902	46.141	1923.15	18.4541176
25.288	3085.76	30.25254902	37.9972	1555.58	15.25074831	32.2343	3803.37	37.28794118	46.1903	1926.8	18.89019608
25.324	3093.85	30.33186275	38.0738	1559.19	15.28617647	32.3066	3806.56	37.31921569	46.2924	1930.46	18.92607843
25.3867	3098.68	30.37921569	38.1504	1560.0	15.29470588	32.3463	3814.81	37.40009804	46.3435	1932.85	18.9495098
25.4984	3104.29	30.43421569	38.2015	1560.31	15.29715686	32.4483	3819.86	37.44960784	46.4456	1935.5	18.9754902
25.5356	3109.51	30.48539216	38.2898	1566.79	15.36068627	32.525	3822.79	37.47833333	46.5064	1938.53	0.00519608
25.5748	3114.66	30.55333333	38.3337	1568.41	15.37656863	32.5761	3828.75	37.53075882	46.5358	1942.59	19.01407843
25.637	3122.05	30.60833333	38.4099	1569.5	15.39171676	32.6229	3834.86	37.5945098	46.6139	1941.94	18.39824706
25.7397	3126.01	30.66833333	38.4837	1570.32	15.42940661	32.6569	3837.97	37.6578474	46.7126	1947.61	19.16207569
25.8566	3134.01	30.7255924	38.5549	1571.32	15.47215689	32.7089	3844.97	37.69303553	46.7522	1951.78	19.15529112
25.9221	3140.28	30.78705882	38.6167	1575.61	15.45569039	32.8363	3845.43	37.70029412	46.8288	1953.04	17.47455039
25.9792	3145.8	30.83823529	38.6981	1577.13	15.46205882	32.8319	3851.83	37.76330322	46.8799	1956.32	18.17908784
25.9772	3152.28	30.90470588	38.7652	1580.87	15.49872549	32.9567	3857.16	37.81529412	47.0075	1958.72	18.20313725
26.0144	3159.47	30.97519608	38.8408	1581.24	15.50235294	33.0108	3862.23	37.88517	47.0585	1961.38	18.22921569
26.1011	3165.61	31.03539216	38.9083	1584.86	15.53784314	33.1067	3866.77	37.92991176	47.1099	1964.79	18.26264706
26.1872	3171.49	31.09303922	39.0107	1587.44	15.53666667	33.1546	3871.77	37.95264706	47.2116	1967.07	18.28490196
26.1934	3175.41	31.13417059	39.0545	1587.61	15.56480392	33.1973	3877.45	38.01421569	47.2615	1970.09	18.31460784
26.3235	3183.26	31.20843137	39.1513	1589.73	15.58558824	33.2694	3880.92	38.04823529	47.3253	1975.23	19.365
26.3618	3191.37	31.28794118	39.2169	1593.97	15.62715686	33.3336	3887.46	38.11235294	47.3934	1978.64	18.39843137
26.4267	3197.26	31.34568627	39.2683	1593.84	15.62582353	33.3777	3893.19	38.16852941	47.4383	1981.42	18.42586627
26.4552	3204.07	31.41425098	39.3155	1595.34	15.64058824	33.4732	3899.32	38.22867245	47.536	1983.56	18.44666667
26.5141	3209.44	31.46509804	39.3777	1600.83	15.69441176	33.55601	3903.33	38.26794118	47.5785	1987.85	18.48872549
26.6112	3217.04	31.53696784	39.4436	1604.71	15.73245098	33.5997	3908.8	38.32156863	47.649	1990.88	18.51843137
26.6368	3222.66	31.59470588	39.5121	1607.58	15.76058824	33.7159	3914.27	38.37519608	47.7195	1993.02	18.53941176
26.7137	3228.94	31.65627451	39.5831	1609.5	15.78019608	33.7918	3920.94	38.44058824	47.7494	1997.19	18.58029412
26.7649	3235.49	31.7204902	39.6497	1610.73	15.8145152	33.8587	3925.75	38.47801569	47.8028	2001.98	18.61505656
26.8155	3240.82	31.78530939	39.6967	1612.69	15.83521568	33.9097	3930.56	38.48901491	47.9013	2004.76	18.63725949
26.8757	3247.32	31.84627451	39.7665	1616.07	15.863043137	33.9475	3932.56	38.55450986	47.9707	2007.87	18.693022
26.9446	3252.77	31.88991956	39.8364	1620.82	15.88039216	34.0597	3939.35	38.62800904	48.0288	2011.21	19.71774511
26.998	3258.67	31.94477451	39.9294	1624.7	15.9243137	40.359	3945.26	38.67901961	48.1228	2012.15	17.75617647
26.9927	3267.72	32.03647059	39.9708	1628.82	15.96828253	40.4162	3949.94	38.72490196	48.1649	2017.53	17.79770588
27.0358	3272.96	32.08784314	40.0151	1630.72	15.970582	40.4237	3954.28	38.74980392	48.2294	2020.81	18.81186275
27.1289	3279.13	32.14833333	40.0534	1636.38	16.04735294	40.4391	3959.04	38.81411765	48.2492	2022.33	18.82677471
27.1835	3284.38	32.19980392	40.1188	1641.25	16.09068862	40.4354	3963.32	38.856707843	48.357	2028.15	18.88323533
27.2627	3292.65	32.28088235	40.1563	1644.49	16.12025882	40.4423	3970.75	38.89281257	48.3998	2031.31	19.91480392
27.3483	3300.14	32.35431373	40.2364	1645.12	16.12862745	40.4979	3978.24	39.00235294	48.4908	2035.61	19.95696078
27.3892	3305.7	32.40882353	40.3326	1647.37	16.15086227	40.5487	3983.86	39.05078431	48.5395	2037.64	19.97686275
27.4313	3310.55	32.54537255	40.3718	1654.24	16.21803922	40.6281	3987.32	39.09137255	48.5905	2040.04	20.00039216
27.4825	3317.25	32.55205882	40.4229	1657.24	16.24745098	40.7415	3990.67	39.12421569	48.667	2043.59	20.03519608
27.5593	3323.28	32.58117647	40.5249	1660.24	16.27668275	40.7903	3996.82	39.1845098	48.7946	2048.02	20.07862745
27.6255	3330.5	32.65196078	40.6269	1662.69	16.2927451	40.8449	4003.91	39.25401961	48.8173	2051.57	20.11343173
27.6644	3337.59	32.72147059	40.6683	1665.36	16.32705882	40.8515	4004.9	39.30784314	48.89	2053.34	20.13078431
27.7379	3344.97	32.79627451	40.7336	1671.24	16.37017647	40.8731	4011.56	39.36431373	48.9814	2058.29	20.16039223
27.7945	3352.18	32.86505688	40.7744	1674.24	16.41417165	40.9097	4028.01	39.40109804	49.0286	2061.75	20.20294118
27.8567	3358.72	32.92974549	40.8970	1676.67	16.46931765	40.9378	4030.65	39.45505934	49.1519	2062.66	20.20215693
27.9024	3363.41	33.00095392	40.9597	1683.38	16.50372549	40.995	4039.95	39.60735294	49.1974	2075.11	20.34421692
27.9949	3369.80	33.14441176	41.0099	1684.88	16.51843137	40.9972	4046.12	39.65784317	49.2687	2078.02	20.37274511
28.0563	3385.72	33.19333333	41.0864	1687.26	16.54176471	41.0365	4050.28	39.70862745	49.3426	2079.54	20.38764706
28.0872	3391.49	33.25592157	41.1374	1692.24	16.55495098	41.0548	4056.32	39.76784314	49.4117	2083.72	20.42674549
28.1787	3398.47	33.31833333	41.2779	1699.03	16.57151686	41.0879	4059.7088	39.80997588	49.5627	2091.31	20.50303922
28.2032	3413.6	33.46666667	41.3243	1701.28	16.57921569	41.1582	4077.49	39.87539216	49.6015	2093.97	20.52117647
28.2535	3421.1	33.54019608	41.3908	1706.04	16.72582353	41.2046	4084.6	40.04509804	49.6369	2099.17	20.58009804
28.4513	3428.34	33.61117647	41.4312	1710.13	16.76892157	41.2558	4087.42	40.0727451	49.7134	2103.73	20.62480392
28.5511	3436.11	33.									

2min He/ Air Plasma			Control			
38.4135	4416.29	43.29696078	52.6206	2282.79	22.38029412	
38.4819	4424.81	43.3804902	52.6951	2285.6	22.40784314	
38.5467	4433.21	43.46284314	52.748	2283.94	23.39156863	
38.5897	4439.58	43.52529421	52.8723	2286.24	24.411765	
38.6348	4445	43.57843137	52.8904	2290.32	24.4511765	
38.7033	4452.46	43.65156863	53.013	2292.99	24.48209412	
38.7976	4461.01	43.73539216	53.0837	2296.18	25.21156863	
38.8232	4466.17	43.78598039	53.1775	2300.39	25.52268434	
38.8743	4475.24	43.87490196	53.1735	2302.18	25.57039216	
38.951	4483.33	43.95970784	53.2679	2304.18	26.60666675	
39.072	4494.91	44.04170568	53.3077	2307.25	27.00000000	
39.073	4497.2	44.04932695	53.4267	2317.25	27.18137675	
39.1237	4506.29	44.17931373	53.4986	2321.58	27.76058824	
39.19	4515.25	44.26715686	53.5623	2327.77	28.02058824	
39.232	4523.13	44.34441176	53.6396	2330.25	28.44585824	
39.3246	4531.28	44.42431373	53.7138	2335.86	28.90058824	
39.3662	4539.44	44.50431373	53.741	2342.62	29.26686275	
39.4676	4548	44.58823293	53.7931	2349.51	29.3441176	
39.4782	4556.91	44.67558824	53.8862	2354.11	29.73095098	
39.5592	4560.57	44.71147059	53.9059	2357.17	29.13509058	
39.6854	4572.12	44.82470588	53.9456	2363.18	23.16843527	
39.7027	4579.46	44.89666667	54.0041	2371.74	23.25253294	
39.7698	4589.39	44.99401961	54.0686	2377.23	23.30617647	
39.8465	4598.86	45.08470588	54.1293	2384.01	23.37624706	
39.8465	4604.9	45.14607843	54.1707	2388.1	23.41274751	
39.9235	4613.33	45.22892157	54.2542	2395.72	23.48745098	
39.951	4618.93	45.28362745	54.2619	2402.75	23.5563275	
39.976	4622.15	45.3381176	54.2947	2409.71	23.6327474	
39.9762	4630.77	45.38970588	54.4288	2423.71	27.76058824	
40.0519	4646.46	45.53352041	54.4706	2428.82	28.31160876	
40.103	4653.54	45.62204118	54.5392	2433.17	23.85460784	
40.1541	4661.17	45.69774373	55.1432	2519.68	24.70254902	
40.282	4670.31	45.78735294	55.1854	2520.02	24.79493173	
40.3179	4679.17	45.87421569	55.2753	2534.93	24.8525249	
40.3553	4685.31	45.93441176	55.2717	2536.73	24.86990196	
40.4118	4694.31	46.02264706	55.3447	2542.76	49.2901961	
40.4273	4702.1	46.09901961	55.4234	2547.9	24.97941176	
40.4926	4711.52	46.19137255	55.4743	2555.1	25.05	
40.5586	4720.88	46.28117647	55.525	2563.12	25.12862745	
40.6253	4729.5	46.36746706	55.5956	2570.69	25.20284314	
40.6678	4737.14	46.42454902	55.627	2579.04	25.28704588	
40.6989	4744.7	46.51754902	55.7288	2584.3	25.33627451	
40.7853	4753.4	46.60196078	55.7196	2590.85	25.4004902	
40.8696	4761.59	46.68225249	55.808	2598.95	25.47990196	
40.9074	4770.47	46.74549058	55.8464	2608.8	25.55686275	
40.9369	4778.85	46.83591373	55.8973	2610.01	25.64005876	
41.0149	4782.67	46.94328235	55.9328	2617.59	25.71576431	
41.0589	4794.7	47.00647397	56.0236	2623.29	25.80677647	
41.1532	4801.83	47.07676471	56.0236	2638.76	25.88	
41.194	4811.3	47.16960784	56.0601	2650.16	25.98160708	
41.1855	4819.1	47.24607843	56.1365	2655.44	26.03372549	
41.3056	4827.99	47.33323259	56.1757	2665.74	26.13470588	
41.3823	4836.35	47.41519608	56.2129	2672.82	26.20411765	
41.4078	4844.15	47.49166667	56.2892	2680.03	26.27480392	
41.51	4850.46	47.55332041	56.328	2688.93	26.3620582	
41.5356	4857.86	47.62607843	56.3842	2699.73	26.45813725	
41.6378	4867.6	47.72156863	56.3657	2708.28	26.55176471	
41.6484	4867.51	47.80892157	56.4442	2714.48	26.61524902	
41.7391	4881.1	47.854670588	56.5184	2723.39	26.69990196	
41.7916	4893.51	47.97558624	56.5947	2731.94	26.78372549	
41.8171	4901.19	48.05088235	56.6047	2740.45	26.86715676	
41.8593	4910.24	48.1396784	56.6369	2749.74	26.95823529	
41.8989	4919.61	48.22303622	56.6728	2757.5	27.03627451	
41.9706	4927.53	48.31167588	56.773	2767.14	27.11601961	
42.0196	4936.6	48.39803022	56.8545	2775.97	27.24539216	
42.0849	4944.15	48.47205882	56.8243	2783.59	27.29000804	
42.1217	4953.09	48.55970588	56.9003	2792.26	27.35790804	
42.1785	4962.3	48.65	56.9174	2800.54	27.45627451	
42.2135	4969.73	48.72284314	56.948	2811.38	27.5254902	
42.2533	4974.6	48.78823529	57.0131	2817.71	27.62460749	
42.3659	4985.46	48.87705882	57.0532	2827.15	27.71715686	
42.4003	4993.16	48.95254902	57.123	2839.83	27.84147059	
42.4749	5004.03	49.05911765	57.1073	2849.42	27.9354902	
42.5679	5012.42	49.14137255	57.155	2857.83	28.01794118	
42.586	5019.31	49.20892157	57.2567	2866.65	28.10441176	
42.6098	5028.53	49.29931373	57.2999	2877.15	28.20735294	
42.6609	5035.97	49.37225491	57.306	2886.5	28.29901961	
			57.3654	2897.02	28.40215686	
			57.4093	2907.78	28.50764706	
			57.4604	2917.38	28.59160000	
			57.5124	2925.64	28.69305207	
			57.5356	2935.67	28.76107843	
			57.5968	2945.54	28.87734434	
			57.6385	2956.69	28.98617647	
			57.7148	2967.77	29.05784311	
			57.7772	2975.71	29.17632745	
			57.7658	2986.44	29.28782353	
			57.8155	2995.03	29.36294118	

CONTROL						2MIN HE/AIR PLASMA TREATED					
EXTENSION (In)	TIME (Sec)	LOAD (Lb)	Stress (PSI)	Adj Strain (mm/mm)		EXTENSION (In)	TIME (Sec)	LOAD (Lb)	Stress (PSI)	Adj Strain (mm/mm)	
0	0	0.018266	0.146128	0	0	0.001	0.018266	0.146128	0	0.028625	0
0.028875	0.174	0.109596	0.87677	0.028875	0.028625	0.172	0.273991	2.191925	0.028625	0.028625	0
0.079	0.422	0.694109	5.552876	0.079	0.078875	0.428	1.022998	8.183166	0.078875	0.078875	0
0.112625	0.599	1.406485	11.25188	0.112625	0.12925	0.683	2.301521	18.412168	0.12925	0.12925	0
0.17725	0.928	2.575512	20.604093	0.17725	0.1795	0.938	3.507079	28.056637	0.1795	0.1795	0
0.227625	1.183	2.81297	22.50376	0.227625	0.228875	1.194	4.164657	33.317257	0.229875	0.229875	0
0.261	1.353	2.831236	22.64989	0.261	0.29625	1.531	5.425014	43.400111	0.29625	0.29625	0
0.311375	1.602	3.214823	25.718584	0.311375	0.3465	1.786	5.662472	45.299779	0.3465	0.3465	0
0.3615	1.863	3.792197	27.033739	0.3615	0.39675	2.041	5.370215	42.961724	0.39675	0.39675	0
0.395125	2.033	3.452281	27.618253	0.395125	0.43025	2.211	5.699004	45.592034	0.43025	0.43025	0
0.445125	2.287	3.488813	27.910508	0.445125	0.480375	2.466	6.210453	49.683627	0.480375	0.480375	0
0.4955	2.543	3.726272	29.810176	0.4955	0.530625	2.721	6.922829	55.382633	0.530625	0.530625	0
0.528875	2.713	3.799336	30.39469	0.528875	0.581	2.976	6.575774	52.606196	0.581	0.581	0
0.579375	2.966	4.109859	32.878871	0.579375	0.63125	3.232	6.539242	52.313937	0.63125	0.63125	0
0.629375	3.223	4.000263	32.002102	0.629375	0.681625	3.487	6.393113	51.144909	0.681625	0.681625	0
0.663125	3.394	3.835868	30.686947	0.663125	0.73175	3.742	6.447912	51.583295	0.73175	0.73175	0
0.713375	3.649	3.653208	29.225663	0.713375	0.782125	3.997	6.466178	51.729423	0.782125	0.782125	0
0.76375	3.904	3.762804	30.102433	0.76375	0.8155	4.167	6.630572	53.044578	0.8155	0.8155	0
0.79725	4.075	3.981996	31.855973	0.79725	0.86575	4.421	6.667104	53.336837	0.86575	0.86575	0
0.8475	4.33	4.109859	32.878871	0.8475	0.916	4.677	6.137389	49.099113	0.916	0.916	0
0.89775	4.585	3.927198	31.417588	0.89775	0.966375	4.932	6.466178	51.729423	0.966375	0.966375	0
0.93125	4.755	4.018528	32.148229	0.93125	1.016625	5.187	7.032425	56.259401	1.016625	1.016625	0
0.9815	5.01	4.109859	32.878871	0.9815	1.07	5.459	6.959361	55.674888	1.07	1.07	0
1.015	5.18	4.000263	32.002102	1.015	1.12025	5.714	7.087223	56.697788	1.12025	1.12025	0
1.068375	5.451	4.164657	33.17257	1.068375	1.170375	5.968	7.215085	57.720684	1.170375	1.170375	0
1.11875	5.707	4.292519	34.340155	1.11875	1.203875	6.139	7.47081	59.76648	1.203875	1.203875	0
1.15225	5.877	4.000263	32.002102	1.15225	1.254125	6.393	7.233352	57.866815	1.254125	1.254125	0
1.202375	6.131	4.182923	33.463384	1.202375	1.30475	6.65	7.269883	58.15907	1.30475	1.30475	0
1.25275	6.386	3.981996	31.855973	1.25275	1.355	6.906	7.087223	56.697788	1.355	1.355	0
1.286125	6.556	3.817602	30.540818	1.286125	1.40525	7.161	7.744801	61.958408	1.40525	1.40525	0
1.336375	6.811	3.671474	29.371792	1.336375	1.4555	7.416	7.397745	59.181966	1.4555	1.4555	0
1.3865	7.066	3.744538	29.956306	1.3865	1.505625	7.671	7.087223	56.697788	1.505625	1.505625	0
1.420125	7.237	3.817602	30.540818	1.420125	1.55575	7.926	7.105489	56.843915	1.55575	1.55575	0
1.470375	7.491	3.835868	30.686947	1.470375	1.598975	8.096	7.269883	58.15907	1.598975	1.598975	0
1.520625	7.747	3.141749	27.325996	1.520625	1.6395	8.352	7.269883	58.15907	1.6395	1.6395	0
1.554	7.917	4.561877	28.495021	1.554	1.68975	8.606	6.922822	55.382633	1.68975	1.68975	0
1.60425	8.172	3.634942	29.079535	1.60425	1.74	8.862	6.941095	55.52876	1.74	1.74	0
1.6545	8.427	3.452281	27.618253	1.6545	1.79025	9.117	7.123755	56.990043	1.79025	1.79025	0
1.688125	8.597	3.488813	27.910508	1.688125	1.840375	9.372	6.886296	55.090374	1.840375	1.840375	0
1.738375	8.852	3.634942	29.079535	1.738375	1.8905	9.627	6.97627	55.821015	1.8905	1.8905	0
1.788625	9.107	3.360951	28.887609	1.788625	1.94075	9.881	6.703636	53.629092	1.94075	1.94075	0
1.822	9.277	3.397483	27.179866	1.822	1.974375	10.052	6.667104	53.336837	1.974375	1.974375	0
1.87225	9.531	3.287887	26.303098	1.87225	2.0245	10.307	6.86803	54.944246	2.0245	2.0245	0
1.905625	9.701	3.360951	26.887609	1.905625	2.074625	10.562	6.429646	51.437168	2.074625	2.074625	0
1.956	9.956	3.634942	29.079535	1.956	2.127125	10.828	6.758434	54.067478	2.127125	2.127125	0
2.00625	10.211	3.671474	29.371792	2.00625	2.177375	11.083	6.918197	53.190709	2.177375	2.177375	0
2.039625	10.381	3.561877	28.495021	2.039625	2.227625	11.338	6.137389	49.099113	2.227625	2.227625	0
2.08975	10.635	3.616676	28.933408	2.08975	2.27775	11.593	6.064325	48.514603	2.27775	2.27775	0
2.14225	10.902	3.525345	28.202764	2.14225	2.327875	11.848	5.680738	45.445907	2.327875	2.327875	0
2.175875	11.072	3.525345	28.202764	2.175875	2.361375	12.018	5.735536	45.884293	2.361375	2.361375	0
2.226125	11.327	3.835868	30.686947	2.226125	2.411625	12.273	5.918197	47.345576	2.411625	2.411625	0
2.2765	11.582	4.201189	33.609514	2.2765	2.46175	12.527	5.845132	46.761062	2.46175	2.46175	0
2.309875	11.752	4.402115	35.216924	2.309875	2.511875	12.782	6.046059	48.368472	2.511875	2.511875	0
2.360125	12.007	4.365583	34.924667	2.360125	2.562125	13.038	6.137389	49.099113	2.562125	2.562125	0
2.4105	12.263	4.201189	33.609514	2.4105	2.612375	13.293	6.283517	50.26814	2.612375	2.612375	0
2.444	12.433	4.201189	33.609514	2.444	2.6625	13.547	6.173921	49.391372	2.6625	2.6625	0
2.494375	12.688	4.182923	33.463384	2.494375	2.71275	13.803	6.027793	48.222344	2.71275	2.71275	0
2.544375	12.943	4.420381	35.363053	2.544375	2.7646375	13.973	6.301783	50.414268	2.746375	2.746375	0
2.59925	13.221	4.511712	36.093694	2.59925	2.7965	14.228	6.520796	52.167809	2.7965	2.7965	0
2.649375	13.476	4.310785	34.486283	2.649375	2.84675	14.483	6.721902	53.775219	2.84675	2.84675	0
2.683	13.646	4.56651	36.532081	2.683	2.897	14.738	6.758434	54.067478	2.897	2.897	0
2.73325	13.901	4.56651	36.532081	2.73325	2.947375	14.994	6.557508	52.460064	2.947375	2.947375	0
2.783	14.156	4.639574	37.11659	2.783	2.997625	15.249	6.612306	52.89845	2.997625	2.997625	0
2.816875	14.326	4.438647	35.509181	2.816875	3.04775	15.504	6.922829	55.382633	3.04775	3.04775	0
2.86725	14.581	4.163931	33.171228	2.86725	3.098125	15.76	6.886296	55.090374	3.098125	3.098125	0
2.917625	14.837	4.475183	35.801444	2.917625	3.131625	15.93	6.886296	55.090374	3.131625	3.131625	0
2.951125	15.006	4.456913	35.655308	2.951125	3.181625	16.184	6.703636	53.629092	3.181625	3.181625	0
3.001375	15.261	4.420381	35.363053	3.001375	3.231875	16.438999	6.703636	53.629092	3.231875	3.231875	0
3.0515	15.517	4.8405	38.724004	3.0515	3.282125	16.695	7.160287	57.282302	3.282125	3.282125	0
3.085125	15.687	4.968362	39.7469	3.085125	3.33225	16.949999	6.849765	54.798119	3.33225	3.33225	0
3.1355	15.942	4.438647	35.509181	3.1355	3.3825	17.205	7.160287	57.282302	3.3825	3.3825	0
3.185625	16.197001	4.785702	38.285618	3.185625	3.43275	17.459	7.251617	58.012943	3.43275	3.43275	0
3.21925	16.368	4.548243	36.385949	3.21925	3.482875	17.715	7.142021	57.13617	3.482875	3.482875	0
3.269375	16.622	4.65784	37.262722	3.269375	3.516375	17.885	7.379479	59.035839	3.516375	3.516375	0
3.319625	16.877001	4.822234	38.577877	3.319625	3.5392						

4.379	22.254	3.908932	31.271461	4.379	4.726125	24.028999	8.402378	67.219028	4.726125
4.42925	22.509001	3.744538	29.956306	4.42925	4.77625	24.283001	8.237984	65.903873	4.77625
4.462625	22.677999	4.109859	32.878871	4.462625	4.8265	24.539	8.146653	65.173227	4.8265
4.513	22.934	3.981996	31.855973	4.513	4.876875	24.794001	8.183185	65.465486	4.876875
4.563125	23.188994	3.981996	31.855973	4.563125	4.927	25.049	8.53024	68.241924	4.927
4.596625	23.358994	4.292519	34.340155	4.596625	4.9775	25.304001	9.187817	73.50254	4.9775
4.646875	23.614	4.274253	34.194026	4.646875	5.02775	25.559999	8.639836	69.118692	5.02775
4.69725	23.868999	4.493446	35.947567	4.69725	5.06125	25.73	8.292781	66.342255	5.06125
4.73075	24.039	4.56651	36.532081	4.73075	5.11125	25.983999	8.055323	64.442586	5.11125
4.781	24.294001	4.310785	34.486283	4.781	5.161625	26.239	8.037057	64.296459	5.161625
4.83125	24.544	4.255987	34.047898	4.83125	5.211875	26.493999	7.452544	59.620353	5.211875
4.864625	24.719	3.835868	30.686947	4.864625	5.262125	26.749001	7.489076	59.912612	5.262125
4.914875	24.974001	4.073327	32.586614	4.914875	5.312375	27.004	8.402378	67.219028	5.312375
4.965125	25.229	4.091593	32.732743	4.965125	5.362625	27.26	8.80423	70.433847	5.362625
4.99875	25.4	3.78107	30.248561	4.99875	5.412875	27.514999	8.365846	66.926769	5.412875
5.048875	25.653994	4.018528	32.148229	5.048875	5.446375	27.684999	8.292781	66.342255	5.446375
5.099375	25.91	4.274253	34.194026	5.099375	5.496625	27.940001	7.817865	62.542922	5.496625
5.13275	26.08	4.055061	32.440486	5.13275	5.546875	28.195	7.708268	61.666149	5.546875
5.183125	26.336	3.945464	31.563716	5.183125	5.595725	28.451	7.708268	61.666149	5.59725
5.23325	26.59	4.000263	32.002102	5.23325	5.64725	28.705	7.817865	62.542922	5.64725
5.266875	26.761	3.744538	29.956306	5.266875	5.697625	28.959999	8.146653	65.173227	5.697625
5.317	27.016001	3.908932	31.271461	5.317	5.747875	29.216	7.763067	62.104535	5.747875
5.350375	27.184999	3.835868	30.686947	5.350375	5.798125	29.471001	7.726534	61.812276	5.798125
5.400875	27.441	4.307079	28.056637	5.400875	5.831625	29.641001	7.580406	60.643253	5.831625
5.451125	27.695999	3.671479	29.371792	5.451125	5.882125	29.896999	7.744801	61.958408	5.882125
5.484625	27.865999	3.525345	28.202764	5.484625	5.954625	30.264999	7.507342	60.058739	5.954625
5.534875	28.121	3.561877	28.495021	5.534875	6.005	30.521	7.434272	59.474225	6.005
5.585	28.375999	3.634942	29.079535	5.585	6.055	30.775	7.863131	62.689049	6.055
5.6185	28.546	4.274538	29.956306	5.6185	6.10525	31.030001	7.909195	63.273563	6.10525
5.66875	28.801001	3.616676	28.933408	5.66875	6.155625	31.285999	7.671736	61.373894	6.155625
5.719	29.055	4.018528	32.148229	5.719	6.205875	31.541	7.671736	61.373894	6.205875
5.7525	29.225	3.945464	31.563716	5.7525	6.23925	31.709999	7.690001	61.520021	6.23925
5.80275	29.481001	3.793936	30.39469	5.80275	6.2895	31.966	7.744801	61.958408	6.2895
5.853	29.738	4.383849	35.070796	5.853	6.33975	32.21001	8.055323	64.442586	6.33975
5.886625	29.907	4.219455	33.755641	5.886625	6.39	32.476002	8.201451	65.611614	6.39
5.936875	30.160999	4.219455	33.755641	5.936875	6.44025	32.73	8.585038	68.68031	6.44025
5.98725	30.417	4.146391	33.171128	5.98725	6.4905	33.098	8.676368	69.410951	6.4905
6.020625	30.587	3.945464	31.563716	6.020625	6.5405	33.240002	8.877295	71.018361	6.5405
6.070875	30.841999	4.292519	34.340155	6.070875	6.59075	33.495998	8.785965	70.28772	6.59075
6.121125	31.096001	4.347317	34.778539	6.121125	6.62425	33.665001	8.493708	67.949669	6.62425
6.154625	31.267	4.47518	35.80144	6.154625	6.6745	33.919998	8.365846	66.926769	6.6745
6.204875	31.521	4.676106	37.408849	6.204875	6.72475	34.174999	8.402378	67.219028	6.72475
6.2385	31.691999	4.47518	35.80144	6.2385	6.775	34.431	8.329314	66.634514	6.775
6.288625	31.947001	4.438647	35.509181	6.288625	6.825125	34.686001	8.365846	66.926769	6.825125
6.339	32.202	4.65784	37.262722	6.339	6.8755	34.941002	8.274516	66.196128	6.8755
6.372375	32.372002	4.511712	36.093694	6.372375	6.925625	35.195999	7.963993	63.711945	6.925625
6.422625	32.626999	4.858766	38.870132	6.422625	6.976	35.452	8.055323	64.442586	6.976
6.472875	32.882	5.114491	40.915928	6.472875	7.009375	35.622002	7.872663	62.981304	7.009375
6.5065	33.051998	5.041427	40.331414	6.5065	7.059625	35.876999	8.037057	64.296459	7.059625
6.55675	33.307999	4.8405	38.724004	6.55675	7.109875	36.132	7.799598	62.396979	7.109875
6.607	33.562	5.004895	40.039159	6.607	7.160125	36.387001	8.219717	65.577741	7.160125
6.6405	33.733002	4.8405	38.724004	6.6405	7.21025	36.641998	7.744801	61.958408	7.21025
6.690875	33.987999	4.529977	36.239822	6.690875	7.2605	36.896999	7.288149	58.305198	7.2605
6.741125	34.243	4.511712	36.093694	6.741125	7.31075	37.152	8.237984	65.903873	7.31075
6.774625	34.412998	4.730904	37.847236	6.774625	7.361	37.407001	7.854397	62.835176	7.361
6.824875	34.667999	4.8405	38.724004	6.824875	7.3945	37.577	7.763067	62.104535	7.3945
6.875	34.923	5.096225	40.678001	6.875	7.444875	37.833	8.877295	71.018361	7.444875
6.908625	35.092999	5.571142	44.569138	6.908625	7.495	38.080001	8.603304	68.826438	7.495
6.958875	35.348999	5.260619	42.084956	6.958875	7.545125	38.341998	9.32093	71.156748	7.545125
7.009125	35.603001	5.53461	44.276879	7.009125	7.595375	38.597	9.827129	78.617036	7.595375
7.0425	35.772998	5.662472	45.299779	7.0425	7.645625	38.852001	10.009789	80.078319	7.645625
7.09275	36.028	4.694372	37.554977	7.09275	7.695625	39.106998	9.388744	75.109954	7.695625
7.126375	36.198002	4.456913	35.655308	7.126375	7.746	39.362	9.352212	74.817699	7.746
7.176625	36.453999	4.109859	32.878871	7.176625	7.797375	39.532001	9.808863	78.4740905	7.797375
7.22675	36.704	4.584776	36.678208	7.22675	7.829625	39.768999	9.58967	76.717368	7.829625
7.260375	36.879002	4.292519	34.340155	7.260375	7.87975	40.041	8.932093	71.156748	7.87975
7.310625	37.133999	4.237721	33.901769	7.310625	7.95525	40.424999	9.443542	75.54834	7.95525
7.360875	37.389	4.639574	37.11659	7.360875	8.0055	40.68	9.754064	78.032518	8.0055
7.394625	37.560001	4.565651	36.532081	7.394625	8.0555	40.933998	10.502972	84.02378	8.0555
7.444875	37.814999	4.347317	34.778539	7.444875	8.10575	41.188999	10.028055	80.224442	8.10575
7.49525	38.07	5.132757	41.602055	7.49525	8.15575	41.443001	10.28055	80.224442	8.15575
7.528625	38.240002	4.730904	37.847236	7.528625	8.18925	41.613998	9.754064	78.032518	8.18925
7.579	38.494999	4.694372	37.554977	7.579	8.239375	41.868	9.534872	76.278981	8.239375
7.629125	38.75	4.968362	39.7469	7.629125	8.289625	42.123001	9.754064	78.032518	8.289625
7.662625	38.919998	4.65784	37.262722	7.662625	8.339625	42.375999	9.790597	78.324782	8.339625
7.712875	39.174999	5.059693	40.477546	7.712875	8.389875	42.632	10.064588	80.516705	8.389875
7.763125	39.43	4.877032	39.091259	7.763125	8.44	42.887001	9.754064	78.032518	8.440001
7.796625	39.599998	4.548243	36.385949	7.796625	8.498625	43.18998	9.206083	73.646871	8.498625
7.847	39.855	5.370215	42.961724	7.847	8.54875	43.438998	10.228981	81.831856	8.54875
7.89725	40.110001	4.694372	37.554977	7.89725	8.58225	43.609001	10.393376	83.147015	8.58225
7.93075	40.279999	4.730904	37.847236	7.93075	8.63225	43.862998	10.74183	81.39347	8.63225
7.98125	40.535999	5.297151	42.37721	7.98125	8.682375	44.118	10.55777	84.462166	8.682375
8.014625	40.706001	4.93183	43.454646	8.014625	8.732625	44.372002	10.302046	82.416374	8.732625
8.065	40.960999	4.968362	39.7469	8.065	8.782625	44.625998	11.160		